



S.C. CONFIND - S.R.L.

CAMPINA

**BREVIAR DE CALCUL**

**SEPARATOR ORIZONTAL TRIFAZIC 80mc**

**P3902-BC**

REVIZIE					
1	30.10.2012		A.Maftei	A.Stoica	G.Colea
Rev.	Data	Descriere	Intocmit	Verificat	Aprobat

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DESIGN CALCULATION

In Accordance with European Code EN-13445

Analysis Performed by : CONFIND S.R.L.

Job File : D:\PROIECTE CONFIND\P3902-SOT 80 MC\LUCRU\PVELIT

Date of Analysis : Nov 7,2012

PV Elite 2011, March 2011, Service Pack 1

PV Elite 2011 SP1 Licensee: CONFIND S.R.L.

FileName : P3902-BC -----

Warnings and Errors : Step: 0 7:08a Nov 7,2012

Class From To : Basic Element Checks.

=====

Class From To: Check of Additional Element Data

=====

There were no geometry errors or warnings.

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**PV Elite Vessel Analysis Program: Input Data**

Design Internal Pressure (for Hydrotest)	0.6000	MPa
Design Internal Temperature	60	C
Type of Hydrotest	1	
Hydrotest Position	Horizontal	
Projection of Nozzle from Vessel Top	0.0000	mm
Projection of Nozzle from Vessel Bottom	0.0000	mm
Minimum Design Metal Temperature	-20	C
Type of Construction	Welded	
Special Service	None	
Degree of Radiography	N/A	
Miscellaneous Weight Percent	0.	
Use Higher Longitudinal Stresses (Flag)	Y	
Select t for Internal Pressure (Flag)	N	
Select t for External Pressure (Flag)	N	
Select t for Axial Stress (Flag)	N	
Select Location for Stiff. Rings (Flag)	N	
Consider Vortex Shedding	N	
Perform a Corroded Hydrotest	N	
Is this a Heat Exchanger	No	
User Defined Hydro. Press. (Used if > 0)	0.0000	MPa
User defined MAWP	0.0000	MPa
User defined MAPnc	0.0000	MPa
Load Case 1	NP+EW+WI+FW+BW	
Load Case 2	NP+EW+EE+FS+BS	
Load Case 3	NP+OW+WI+FW+BW	
Load Case 4	NP+OW+EQ+FS+BS	
Load Case 5	NP+HW+HI	
Load Case 6	NP+HW+HE	
Load Case 7	IP+OW+WI+FW+BW	
Load Case 8	IP+OW+EQ+FS+BS	

FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Load Case 9	EP+OW+WI+FW+BW
Load Case 10	EP+OW+EQ+FS+BS
Load Case 11	HP+HW+HI
Load Case 12	HP+HW+HE
Load Case 13	IP+WE+EW
Load Case 14	IP+WF+CW
Load Case 15	IP+VO+OW
Load Case 16	IP+VE+OW
Load Case 17	IP+VF+CW
Load Case 18	FS+BS+IP+OW
Load Case 19	FS+BS+EP+OW

Wind Design Code	ASCE-7	93
Design Wind Speed	112.65	km/hr
Exposure Constant	C	
Importance Factor	1.	
Roughness Factor	1	
Base Elevation	0.0000	mm
Percent Wind for Hydrotest	33.	
Using User defined Wind Press. Vs Elev.	N	
Damping Factor (Beta) for Wind (Ope)	0.0100	
Damping Factor (Beta) for Wind (Empty)	0.0000	
Damping Factor (Beta) for Wind (Filled)	0.0000	

Seismic Design Code	UBC	1997
UBC Seismic Zone (1=1,2=2a,3=2b,4=3,5=4)	0	
UBC Importance Factor	1.250	
UBC Seismic Coefficient Ca	0.360	
UBC Seismic Coefficient Cv	0.840	
UBC Seismic Coefficient Nv	1.000	
UBC Horizontal Force Factor	2.000	
Apply Allowables per paragraph 1612.3.2	No	

FileName : P3902-BC -----

Input Echo : Step: 1 7:08a Nov 7,2012

Design Nozzle for Des. Press. + St. Head	Y
Consider MAP New and Cold in Noz. Design	N
Consider External Loads for Nozzle Des.	Y
Use ASME VIII-1 Appendix 1-9	N

**Complete Listing of Vessel Elements and Details:**

Element From Node	10
Element To Node	20
Element Type	Elliptical
Description	FUND ELIPSOIDAL 1
Distance "FROM" to "TO"	50.000 mm
Inside Diameter	3000.0 mm
Element Thickness	14.000 mm
Internal Corrosion Allowance	3.0000 mm
Nominal Thickness	14.000 mm
External Corrosion Allowance	0.0000 mm
Design Internal Pressure	0.6000 MPa
Design Temperature Internal Pressure	60 C
Design External Pressure	0.0000 MPa
Design Temperature External Pressure	-20 C
Effective Diameter Multiplier	1.2
Material Name	P355NH
Allowable Stress, Ambient	204.17 MPa
Allowable Stress, Operating	204.17 MPa
Material Density	0.007810 kg/cm <sup>3</sup>
Elliptical Head Factor	2.
Element From Node	10
Detail Type	Nozzle
Detail ID	GV1a
Dist. from "FROM" Node / Offset dist	0.0000 mm
Nozzle Diameter	508. mm

FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Nozzle Schedule	DIN10.0	
Nozzle Class	10	
Layout Angle	0.	
Blind Flange (Y/N)	N	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	10	
Detail Type	Nozzle	
Detail ID	R1	
Dist. from "FROM" Node / Offset dist	1100.0	mm
Nozzle Diameter	219.10001	mm
Nozzle Schedule	DIN10.0	
Nozzle Class	16	
Layout Angle	0.	
Blind Flange (Y/N)	N	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	10	
Detail Type	Nozzle	
Detail ID	R3	
Dist. from "FROM" Node / Offset dist	1200.0	mm
Nozzle Diameter	114.3	mm
Nozzle Schedule	DIN10.0	
Nozzle Class	16	
Layout Angle	180.	
Blind Flange (Y/N)	N	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	



FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Element From Node	20
Element To Node	30
Element Type	Cylinder
Description	VIROLA
Distance "FROM" to "TO"	10350. mm
Inside Diameter	3000.0 mm
Element Thickness	14.000 mm
Internal Corrosion Allowance	3.0000 mm
Nominal Thickness	14.000 mm
External Corrosion Allowance	0.0000 mm
Design Internal Pressure	0.6000 MPa
Design Temperature Internal Pressure	60 C
Design External Pressure	0.0000 MPa
Design Temperature External Pressure	-20 C
Effective Diameter Multiplier	1.2
Material Name	P355NH

Element From Node	20
Detail Type	Saddle
Detail ID	left Sdl
Dist. from "FROM" Node / Offset dist	850.00 mm
Width of Saddle	400.00 mm
Height of Saddle at Bottom	2000.0 mm
Saddle Contact Angle	120.
Height of Composite Ring Stiffener	0.0000 mm
Width of Wear Plate	500.00 mm
Thickness of Wear Plate	14.000 mm
Contact Angle, Wear Plate (degrees)	127.

Element From Node	20
Detail Type	Saddle
Detail ID	right sdl
Dist. from "FROM" Node / Offset dist	9600.0 mm

FileName : P3902-BC -----

Input Echo : Step: 1 7:08a Nov 7,2012

Width of Saddle	400.00	mm
Height of Saddle at Bottom	2000.0	mm
Saddle Contact Angle	120.	
Height of Composite Ring Stiffener	0.0000	mm
Width of Wear Plate	500.00	mm
Thickness of Wear Plate	14.000	mm
Contact Angle, Wear Plate (degrees)	127.	

Element From Node	20	
Detail Type	Nozzle	
Detail ID	GV2a	
Dist. from "FROM" Node / Offset dist	500.00	mm
Nozzle Diameter	508.	mm
Nozzle Schedule	DIN10.0	
Nozzle Class	10	
Layout Angle	0.	
Blind Flange (Y/N)	Y	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	

Element From Node	20	
Detail Type	Nozzle	
Detail ID	GV2b	
Dist. from "FROM" Node / Offset dist	7950.0	mm
Nozzle Diameter	508.	mm
Nozzle Schedule	DIN10.0	
Nozzle Class	10	
Layout Angle	0.	
Blind Flange (Y/N)	Y	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	

FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Element From Node	20	
Detail Type	Nozzle	
Detail ID	M1	
Dist. from "FROM" Node / Offset dist	3860.0	mm
Nozzle Diameter	51.	mm
Nozzle Schedule	DIN12.5	
Nozzle Class	None	
Layout Angle	0.	
Blind Flange (Y/N)	N	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	None	
Nozzle Matl	P285QH	
Element From Node	20	
Detail Type	Nozzle	
Detail ID	R8	
Dist. from "FROM" Node / Offset dist	6940.0	mm
Nozzle Diameter	168.3	mm
Nozzle Schedule	DIN10.0	
Nozzle Class	16	
Layout Angle	0.	
Blind Flange (Y/N)	Y	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	20	
Detail Type	Nozzle	
Detail ID	R2	
Dist. from "FROM" Node / Offset dist	9220.0	mm
Nozzle Diameter	168.3	mm
Nozzle Schedule	DIN10.0	
Nozzle Class	16	

FileName : P3902-BC

Input Echo :

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Layout Angle	0.	
Blind Flange (Y/N)	N	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	20	
Detail Type	Nozzle	
Detail ID	R6	
Dist. from "FROM" Node / Offset dist	9950.0	mm
Nozzle Diameter	219.10001	mm
Nozzle Schedule	DIN10.0	
Nozzle Class	16	
Layout Angle	0.	
Blind Flange (Y/N)	Y	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	20	
Detail Type	Nozzle	
Detail ID	R7c	
Dist. from "FROM" Node / Offset dist	250.00	mm
Nozzle Diameter	168.3	mm
Nozzle Schedule	DIN10.0	
Nozzle Class	40	
Layout Angle	180.	
Blind Flange (Y/N)	N	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	20	

FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Detail Type	Nozzle
Detail ID	R7b
Dist. from "FROM" Node / Offset dist	2700.0 mm
Nozzle Diameter	168.3 mm
Nozzle Schedule	DIN10.0
Nozzle Class	40
Layout Angle	180.
Blind Flange (Y/N)	N
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH
Element From Node	20
Detail Type	Nozzle
Detail ID	R7a
Dist. from "FROM" Node / Offset dist	7650.0 mm
Nozzle Diameter	168.3 mm
Nozzle Schedule	DIN10.0
Nozzle Class	40
Layout Angle	180.
Blind Flange (Y/N)	N
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH
Element From Node	20
Detail Type	Nozzle
Detail ID	R5
Dist. from "FROM" Node / Offset dist	8910.0 mm
Nozzle Diameter	219.10001 mm
Nozzle Schedule	DIN10.0
Nozzle Class	16
Layout Angle	180.

FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Blind Flange (Y/N)	N
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH
Element From Node	20
Detail Type	Nozzle
Detail ID	R4
Dist. from "FROM" Node / Offset dist	10100. mm
Nozzle Diameter	168.3 mm
Nozzle Schedule	DIN10.0
Nozzle Class	16
Layout Angle	180.
Blind Flange (Y/N)	N
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH
Element From Node	20
Detail Type	Nozzle
Detail ID	R17b
Dist. from "FROM" Node / Offset dist	8650.0 mm
Nozzle Diameter	51. mm
Nozzle Schedule	DIN12.5
Nozzle Class	40
Layout Angle	180.
Blind Flange (Y/N)	N
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P285QH
Element From Node	20
Detail Type	Nozzle

FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Detail ID	R17d
Dist. from "FROM" Node / Offset dist	10240. mm
Nozzle Diameter	51. mm
Nozzle Schedule	DIN12.5
Nozzle Class	40
Layout Angle	-161.
Blind Flange (Y/N)	N
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P285QH
Element From Node	20
Detail Type	Nozzle
Detail ID	R11
Dist. from "FROM" Node / Offset dist	9100.0 mm
Nozzle Diameter	88.900002 mm
Nozzle Schedule	DIN8.0
Nozzle Class	16
Layout Angle	90.
Blind Flange (Y/N)	Y
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH
Element From Node	20
Detail Type	Nozzle
Detail ID	R9
Dist. from "FROM" Node / Offset dist	8800.0 mm
Nozzle Diameter	88.900002 mm
Nozzle Schedule	DIN8.0
Nozzle Class	16
Layout Angle	95.685898
Blind Flange (Y/N)	Y

FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	20	
Detail Type	Nozzle	
Detail ID	R18a	
Dist. from "FROM" Node / Offset dist	8950.0	mm
Nozzle Diameter	88.900002	mm
Nozzle Schedule	DIN8.0	
Nozzle Class	16	
Layout Angle	113.347	
Blind Flange (Y/N)	Y	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	20	
Detail Type	Nozzle	
Detail ID	R10	
Dist. from "FROM" Node / Offset dist	9100.0	mm
Nozzle Diameter	88.900002	mm
Nozzle Schedule	DIN8.0	
Nozzle Class	16	
Layout Angle	142.429	
Blind Flange (Y/N)	Y	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	20	
Detail Type	Nozzle	
Detail ID	R12	



FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Dist. from "FROM" Node / Offset dist	8800.0	mm
Nozzle Diameter	88.900002	mm
Nozzle Schedule	DIN8.0	
Nozzle Class	16	
Layout Angle	131.338	
Blind Flange (Y/N)	Y	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P355NH	
Element From Node	20	
Detail Type	Nozzle	
Detail ID	R17a	
Dist. from "FROM" Node / Offset dist	8650.0	mm
Nozzle Diameter	51.	mm
Nozzle Schedule	DIN12.5	
Nozzle Class	40	
Layout Angle	283.	
Blind Flange (Y/N)	N	
Weight of Nozzle ( Used if > 0 )	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	P285QH	
Element From Node	20	
Detail Type	Nozzle	
Detail ID	R20	
Dist. from "FROM" Node / Offset dist	8920.0	mm
Nozzle Diameter	88.900002	mm
Nozzle Schedule	DIN8.0	
Nozzle Class	16	
Layout Angle	0.	
Blind Flange (Y/N)	Y	
Weight of Nozzle ( Used if > 0 )	0.0000	N

FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Grade of Attached Flange	GR 1.1
Nozzle Matl	P285QH
Element From Node	30
Element To Node	40
Element Type	Elliptical
Description	FUND ELIPSOIDAL 2
Distance "FROM" to "TO"	50.000 mm
Inside Diameter	3000.0 mm
Element Thickness	14.000 mm
Internal Corrosion Allowance	3.0000 mm
Nominal Thickness	14.000 mm
External Corrosion Allowance	0.0000 mm
Design Internal Pressure	0.6000 MPa
Design Temperature Internal Pressure	60 C
Design External Pressure	0.0000 MPa
Design Temperature External Pressure	-20 C
Effective Diameter Multiplier	1.2
Material Name	P355NH
Elliptical Head Factor	2.
Element From Node	30
Detail Type	Nozzle
Detail ID	GV1b
Dist. from "FROM" Node / Offset dist	0.0000 mm
Nozzle Diameter	508. mm
Nozzle Schedule	DIN10.0
Nozzle Class	10
Layout Angle	0.
Blind Flange (Y/N)	Y
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH

FileName : P3902-BC

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Input Echo :

Step: 1 7:08a Nov 7,2012

Element From Node	30
Detail Type	Nozzle
Detail ID	R19
Dist. from "FROM" Node / Offset dist	850.00 mm
Nozzle Diameter	219.10001 mm
Nozzle Schedule	DIN10.0
Nozzle Class	16
Layout Angle	180.
Blind Flange (Y/N)	Y
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH

Element From Node	30
Detail Type	Nozzle
Detail ID	R15
Dist. from "FROM" Node / Offset dist	700.00 mm
Nozzle Diameter	88.900002 mm
Nozzle Schedule	DIN8.0
Nozzle Class	16
Layout Angle	90.
Blind Flange (Y/N)	Y
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH

Element From Node	30
Detail Type	Nozzle
Detail ID	R18b
Dist. from "FROM" Node / Offset dist	860.00 mm
Nozzle Diameter	88.900002 mm
Nozzle Schedule	DIN8.0

FileName : P3902-BC

Input Echo :

Step: 1 7:08a Nov 7,2012

Nozzle Class	16
Layout Angle	126.
Blind Flange (Y/N)	Y
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH
Element From Node	30
Detail Type	Nozzle
Detail ID	R16
Dist. from "FROM" Node / Offset dist	1100.0 mm
Nozzle Diameter	88.900002 mm
Nozzle Schedule	DIN8.0
Nozzle Class	16
Layout Angle	141.
Blind Flange (Y/N)	Y
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH
Element From Node	30
Detail Type	Nozzle
Detail ID	R14
Dist. from "FROM" Node / Offset dist	1086.0 mm
Nozzle Diameter	88.900002 mm
Nozzle Schedule	DIN8.0
Nozzle Class	16
Layout Angle	158.
Blind Flange (Y/N)	Y
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P355NH

FileName : P3902-BC

Input Echo :

-----  
Step: 1 7:08a Nov 7,2012

Element From Node	30
Detail Type	Nozzle
Detail ID	R13
Dist. from "FROM" Node / Offset dist	1011.0 mm
Nozzle Diameter	88.900002 mm
Nozzle Schedule	DIN8.0
Nozzle Class	16
Layout Angle	81.
Blind Flange (Y/N)	Y
Weight of Nozzle ( Used if > 0 )	0.0000 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	P285QH

**XY Coordinate Calculations**

From	To	X (Horiz.)	Y (Vert.)	DX (Horiz.)	DY (Vert.)
		mm	mm	mm	mm
FUND ELIPS		50.0000	...	50.0000	...
VIROLA		10400.0	...	10350.0	...
FUND ELIPS		10450.0	...	50.0000	...

FileName : P3902-BC -----

Internal Pressure Calculations : Step: 3 7:08a Nov 7,2012

**Element Thickness, Pressure, Diameter and Allowable Stress :**

From	To	Int. Press + Liq. Hd MPa	Nominal Thickness mm	Total Corr Allowance mm	Element Diameter mm	Allowable Stress(SE) MPa
FUND ELIPS		0.6000	14.000	3.0000	3000.0	204.17
VIROLA		0.6000	14.000	3.0000	3000.0	204.17
FUND ELIPS		0.6000	14.000	3.0000	3000.0	204.17

**Element Required Thickness and MAWP :**

From	To	Design Pressure MPa	M.A.W.P. Corroded MPa	M.A.P. New & Cold MPa	Minimum Thickness mm	Required Thickness mm
FUND ELIPS		0.60000	1.09314	1.41195	14.0000	9.47391
VIROLA		0.60000	1.48884	1.89677	14.0000	7.42332
FUND ELIPS		0.60000	1.29354	1.68320	14.0000	8.87677
Minimum			1.093	1.412		

MAWP: 1.093 MPa, limited by: FUND ELIPSOIDAL 1.

**Internal Pressure Calculation Results :**

European Std: EN 13445-3: 2009(E) Issue 1 (2009-07)

**Elliptical Head From 10 To 20 P355NH at 60 C**

FUND ELIPSOIDAL 1

Design Stress at Ambient Temperature = 204.174 MPa

Required thickness = 9.474 mm

Required thickness in the crown = 6.970 mm

This is an Elliptical Head

The Material is Carbon Steel which affects value of fb:

Buckling Strs at ope. fb = Yield/1.5 = 329.552 /1.5 = 219.701 MPa

Buckling Strs at amb. fb = Yield/1.5 = 355.012 /1.5 = 236.675 MPa

Ellipsoidal head find geometry for equiv. tori-head EN13445 Equation 7.5.4:

Pressure including hydro head P : 0.6000 MPa

Inside Head Diameter (new) Di : 3000.0000 mm

Head Thickness (new) e : 14.0000 mm

Head Aspect Ratio (new) Di/2h : 2.0000

Head Corrosion Allowance Internal ci : 3.0000 mm

Head Corrosion Allowance External co : 0.0000 mm

Joint Efficiency Z : 1.0000

hi = 0.5 \* Di/AR + c = 0.5 \* 3000.00/2.000+3.00 = 753.0000 mm

K = Di / (2 \* hi) = 3006.000 / (2 \* 753.000) = 1.9960

r = Di((0.5 / K) - 0.08) = 3006.000((0.5/1.996)-0.08) = 512.5200 mm

R = Di(0.44\*K+ 0.02) = 3006.000 (0.44\*1.996+0.02) = 2700.1306 mm

The nozzle is outside the 80% limit:

Nozzle corroded inside Diameter di : 225.1000 mm

Y = Min(e/R, 0.04) = Min(6.47391/2700.1306, 0.04) = 0.0023976

Z = Log10(1 / Y) = Log10(1 / 0.002) = 2.6202

X = r / Di = 512.5200 / 3006.0000 = 0.17050

N = 1.006 - 1 / ( 6.2+( 90 \* Y )^4 )

= 1.006-1 / ( 6.2+( 90\*0.0024)^4 ) = 0.84477

Beta01 = N( -0.1833\*Z^3 + 1.0383\*Z^2 - 1.2943\*Z + 0.837 )

= 0.845 (-.1833\*2.620^3 + 2.2124\*2.620^2 - 3.2937\*2.620 + 1.887

= 1.0785



FileName : P3902-BC -----

Internal Pressure Calculations : Step: 3 7:08a Nov 7,2012

$$\begin{aligned} \text{Beta02} &= \max ( 0.95 * ( 0.56 - 1.94 * Y - 82.5 * Y^2), 0.5 ) \\ &= \max ( 0.95 * ( 0.56 - 1.94 * 0.002 - 82.5 * 0.002^2), 0.5 ) \\ &= 0.5271 \end{aligned}$$

$$\begin{aligned} \text{Beta} &= 10 ( ( 0.2 - X ) * \text{Bets01} + ( X - 0.1 ) * \text{Beta02} ) \\ &= 10 ( ( 0.2 - 0.1705 ) * 1.0785 + ( 0.1705 - 0.1 ) * 0.5271 ) \\ &= 0.6898 \end{aligned}$$

Thickness Due to Design Internal Pressure: e = Max(es, ey, eb) - para 7.5.3.2

Compute from Section 7.7.3 [Beta\_k]:

$$\begin{aligned} V &= \text{Log10}(1000 * P / S ) \\ &= \text{Log10}( 1000 * 0.600 / 204.174 ) = 0.4682 \end{aligned}$$

Note: Compute Beta\_K using the Korboggen equations.

$$\begin{aligned} A &= 0.54 + 0.41 * V - 0.044 * V^3 \\ &= 0.54 + 0.41 * 0.468 - 0.044 * 0.468^3 = 0.7274 \\ B &= 7.77 - 4.53 * V + 0.744 * V^2 \\ &= 7.77 - 4.53 * 0.468 + 0.744 * 0.468^2 = 5.8123 \\ \text{Beta}_k &= \max ( A + B * di/De, 1.0 + 0.5 * B * di/De ) \\ &= \max ( 0.727 + 5.812 * 225.100 / 3028.000 , 1.0 + \\ &\quad 0.3 * 5.812 * 225.100 / 3028.000 ) = 1.2160 \end{aligned}$$

Thickness Due to Design Internal Pressure: e = Max(es, ey, eb) - para 7.5.3.2

Required Crown Thickness due to Internal Pressure, see Figure 7.5-3 [es]:

$$\begin{aligned} &= P * R / ( 2 * f * z - 0.5 * P ) \\ &= 0.600 * 2700.1 / ( 2 * 204.174 * 1.00 - 0.5 * 0.600 ) \\ &= 3.9703 \text{ mm} \end{aligned}$$

$$\begin{aligned} e_y &= \text{Beta} * \text{Beta}_k * P ( 0.75 * R + 0.2 * Di ) / f \\ &= 1.216 * 0.690 * 0.600 ( 0.75 * 2700.13 + 0.2 * 3006.00 / \\ &\quad 204.1738 ) = 6.4739 \text{ mm} \end{aligned}$$

FileName : P3902-BC -----

Internal Pressure Calculations : Step: 3 7:08a Nov 7,2012

$$\begin{aligned}
 eb &= (0.75R+0.2Di)((P/111*fb)*(Di/r)^{0.825})^{1/1.5} \\
 &= (0.75*2700.13+0.2*3006.00) * \\
 &\quad (0.60/111 *219.70)(3006.00/512.52)^{0.825})^{1/1.5} \\
 &= 5.8780 \text{ mm}
 \end{aligned}$$

Computed Head Thickness per EN13445 - 7.5.4:

$$\begin{aligned}
 &= \text{Max}(es, eb, ey) + c + c_{ext} = \text{Max}(3.9703, 5.8780, 6.4739) + 3.0000 + 0.0000 \\
 &= 6.4739 + 3.0000 + 0.0000 = 9.4739 \text{ mm}
 \end{aligned}$$

The head is suitable for the design pressure.

Computed Stress at Design Pressure [Stres]:

$$= 112.048 \text{ MPa}$$

Computed Maximum Allowable Working Pressure - Design [MAWP]:

$$\text{MAWP} - \text{Phydro} = 1.0931 - 0.0000 = 1.0931 \text{ MPa}$$

Computed Maximum Pressure New and Cold [MAPNC]:

$$= 1.412 \text{ MPa}$$

Uncorrected (for liquid) Hydrotest Pressure:

$$= 1.534 \text{ MPa}$$

Elongation of the extreme fiber:

For a spun head per EN 13445-4 Para 9.2.1 [F]:

$$\begin{aligned}
 &\text{assuming the blank is 20\% larger than the head diameter} \\
 &= 100 * \ln( 3633.600 / ( 3000.000 - 2 * 14.000 ) ) = 19.161 \%
 \end{aligned}$$

For a segmented head per EN 13445-4 Para 9.2.5 [F]:

$$= 100 * e / R = 100 * 0.551 / 59.331 = 0.929 \%$$

If F exceeds 5% PWHT may be required. Please refer

to EN 13445 Part 4 Table 9.4.1

Required Thickness of Straight Flange = 7.423 mm

**Min. Des. Temp. per EN13445 Part 2: Thickness 14.000 mm at -20 C:**

User entered Minum Design Temperature: -20.000 C

Note: The Component is NOT Post Weld Heat Treated.

Yield Value of the Component 355.0125 MPa

Selected Graph: B4.1-4 per EN13445-2 B4

Note: Computed Impact Test Temp. Tkt: -3 C.

Impact Testing is Required!

**Cylindrical Shell From 20 To 30 P355NH at 60 C**

**VIROLA**

Design Stress at Ambient Temperature = 204.174 MPa

Thickness due to internal pressure [e]:

$$= P * Di / ( 2 * f * z - P ) \text{ EN13445 Equation: 7.4.2:}$$

$$= 0.60 * 3006.00 / ( 2 * 204.174 * 1.000 - 0.600 ) + c + cext$$

$$= 4.4233 + 3.0000 + 0.0000 = 7.4233 \text{ mm}$$

The shell is suitable for the design pressure.

Maximum Working Pressure Hot and Corroded [MAWP]:

$$= ( 2 * f * ecor * z ) / (Di + ecor) - Phead$$

$$= ( 2 * 204.17 * 11.0000 * 1.000 ) / (3006.00 + 11.0000 ) - 0.00$$

$$= 1.489 \text{ MPa}$$

Maximum Pressure New and Cold [MAPNC]:

$$= ( 2 * fa * e * z ) / ( D + e )$$

$$= ( 2 * 204.17 * 14.00 * 1.000 ) / ( 3000.00 + 14.00 )$$

$$= 1.897 \text{ MPa}$$

Stress at Design Pressure [Stres]:

$$= P * ( Di + ecor ) / (2 * ecor * z)$$

$$= 0.600 * (3006.000 + 11.0000 ) / ( 2 * 11.0000 * 1.000 )$$

$$= 82.282 \text{ MPa}$$

Uncorrected (for liquid) Hydrotest Pressure per 6.2.2 stress limitation:

$$= ( 2 * Ftest * e * z ) / (Di + e)$$

$$= ( 2 * 338.11 * 14.00 * 1.000 ) / ( 3000.00 + 14.00 )$$

$$= 3.141 \text{ MPa}$$

Elongation of the extreme fiber per EN 13445-4 Para 9.2.2 [F]

$$= 50 * \text{Max}(e, enom) / Rm = 50 * \text{Max}(14.000 , 14.000 ) / 1508.500 = 0.464 \%$$

**Min. Des. Temp. per EN13445 Part 2: Thickness 14.000 mm at -20 C:**

User entered Minum Design Temperature: -20.000 C

Note: The Component is NOT Post Weld Heat Treated.

Yield Value of the Component 355.0125 MPa

Selected Graph: B4.1-4 per EN13445-2 B4

Note: Computed Impact Test Temp. Tkt: -3 C.

Impact Testing is Required!

**Elliptical Head From 30 To 40 P355NH at 60 C**

**FUND ELIPSOIDAL 2**

Design Stress at Ambient Temperature = 204.174 MPa

Required thickness = 8.877 mm

Required thickness in the crown = 6.970 mm

**This is an Elliptical Head**

**The Material is Carbon Steel which affects value of fb:**

Buckling Strs at ope. fb = Yield/1.5 = 329.652 /1.5 = 219.768 MPa

Buckling Strs at amb. fb = Yield/1.5 = 355.012 /1.5 = 236.675 MPa

Ellipsoidal head find geometry for equiv. tori-head EN13445 Equation 7.5.4:

Pressure including hydro head P : 0.6000 MPa

Inside Head Diameter (new) Di : 3000.0000 mm

Head Thickness (new) e : 14.0000 mm

Head Aspect Ratio (new) Di/2h : 2.0000

Head Corrosion Allowance Internal ci : 3.0000 mm

Head Corrosion Allowance External co : 0.0000 mm

Joint Efficiency Z : 1.0000

$hi = 0.5 * Di/AR + c = 0.5 * 3000.00/2.000+3.00 = 753.0000 \text{ mm}$

$K = Di / (2 * hi) = 3006.000 / (2 * 753.000) = 1.9960$

$r = Di((0.5 / K) - 0.08) = 3006.000((0.5/1.996)-0.08) = 512.5200 \text{ mm}$

$R = Di(0.44*K+ 0.02) = 3006.000 (0.44*1.996+0.02) = 2700.1306 \text{ mm}$

$Y = \text{Min}(e/R, 0.04) = \text{Min}(5.87677/2700.1306, 0.04) = 0.0021765$

$Z = \text{Log10}(1 / Y) = \text{Log10}(1 / 0.002) = 2.6622$

$X = r / Di = 512.5200 / 3006.0000 = 0.17050$

$N = 1.006 - 1 / ( 6.2+( 90 * Y )^4 )$

$= 1.006-1 / ( 6.2+( 90*0.0022)^4 ) = 0.84475$

$\text{Beta01} = N( -0.1833*Z^3 + 1.0383*Z^2 - 1.2943*Z + 0.837 )$

$= 0.845 ( -.1833*2.662^3 + 2.2124*2.662^2 - 3.2937*2.662 + 1.887$

$= 1.0911$

$\text{Beta02} = \text{max} ( 0.95 * ( 0.56 - 1.94 * Y - 82.5 * Y^2), 0.5 )$

$= \text{max} ( 0.95 * ( 0.56 - 1.94 * 0.002 - 82.5 * 0.002^2), 0.5 )$

$= 0.5276$

$\text{Beta} = 10 ( ( 0.2 - X ) * \text{Beta01} + ( X - 0.1 ) * \text{Beta02}$

FileName : P3902-BC -----

Internal Pressure Calculations : Step: 3 7:08a Nov 7,2012

$$= 10 ( ( 0.2 - 0.1705 ) * 1.0911 + ( 0.1705 - 0.1 ) * 0.5276 )$$

$$= 0.6938$$

Thickness Due to Design Internal Pressure:  $e = \text{Max}(e_s, e_y, e_b)$  - para 7.5.3.2

Required Crown Thickness due to Internal Pressure, see Figure 7.5-3 [es]:

$$= P * R / ( 2 * f * z - 0.5 * P )$$

$$= 0.600 * 2700.1 / ( 2 * 204.174 * 1.00 - 0.5 * 0.600 )$$

$$= 3.9703 \text{ mm}$$

$$e_y = \text{Beta} * P * (0.75 * R + 0.2 * D_i) / f = 0.69 * 0.6 * (0.75 * 2700.1 + 0.2 * 3006.0) /$$

$$204.1738 = 5.3550 \text{ mm}$$

$$e_b = (0.75R + 0.2D_i) * ((P / 111 * f_b) * (D_i / r)^{0.825})^{1/1.5}$$

$$= (0.75 * 2700.13 + 0.2 * 3006.00) * (0.60 / 111 * 219.77) * (3006.00 / 512.52)^{0.825}^{1/1.5}$$

$$= 5.8768 \text{ mm}$$

Computed Head Thickness per EN13445 - 7.5.4:

$$= \text{Max}(e_s, e_b, e_y) + c + c_{ext} = \text{Max}(3.9703, 5.8768, 5.3550) + 3.0000 + 0.0000$$

$$= 5.8768 + 3.0000 + 0.0000 = 8.8768 \text{ mm}$$

The head is suitable for the design pressure.

Actual stress at design pressure cannot be computed

because the thickness for buckling pressure controls.

Computed Maximum Allowable Working Pressure - Design [MAWP]:

$$\text{MAWP} - \text{Phydro} = 1.2935 - 0.0000 = 1.2935 \text{ MPa}$$

Computed Maximum Pressure New and Cold [MAPNC]:

$$= 1.683 \text{ MPa}$$

Uncorrected (for liquid) Hydrotest Pressure:

$$= 1.534 \text{ MPa}$$

Elongation of the extreme fiber:

For a spun head per EN 13445-4 Para 9.2.1 [F]:

assuming the blank is 20% larger than the head diameter

$$= 100 * \ln( 3633.600 / ( 3000.000 - 2 * 14.000 ) ) = 19.161 \%$$

For a segmented head per EN 13445-4 Para 9.2.5 [F]:

$$= 100 * e / R = 100 * 0.551 / 59.331 = 0.929 \%$$

If F exceeds 5% PWHT may be required. Please refer to EN 13445 Part 4 Table 9.4.1

Required Thickness of Straight Flange = 7.423 mm

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FileName : P3902-BC -----

Internal Pressure Calculations : Step: 3 7:08a Nov 7,2012

**Min. Des. Temp. per EN13445 Part 2: Thickness 14.000 mm at -20 C:**

User entered Minum Design Temperature: -20.000 C

Note: The Component is NOT Post Weld Heat Treated.

Yield Value of the Component 355.0125 MPa

Selected Graph: B4.1-4 per EN13445-2 B4

Note: Computed Impact Test Temp. Tkt: -3 C.

Impact Testing is Required!

**Hydrostatic Test Pressure Results:**

Note: The Hydrotest Pressure Derivation is an Iterative Process

Limited by: Ellipse Head FUND ELIPSOIDAL

Hydrotest pressure is based upon stress (ftest) in the weakest element:

Note:  $1.5 / 1.05 = 1.429$ , The PED requirement is 1.43

$f_{test} = \text{Yield Stress} / 1.05$

$= 355.012 / 1.05 = 338.107 \text{ MPa}$

Test Pressure = Calc Test Press - Liquid Head

$= 1.534 - 0.029 = 1.504 \text{ MPa}$

Elements Suitable for Internal Pressure.

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FileName : P3902-BC

Element and Detail Weights : Step: 5 7:08a Nov 7,2012

**Element and Detail Weights**

From	To	Element Metal Wgt. kg	Element ID Volume cm3	Corroded Metal Wgt. kg	Corroded ID Volume cm3	Extra due Misc % kg
10	20	1240.58	3.888E+06	974.741	3.911E+06	...
20	30	10756.8	73.17E+06	8460.26	73.47E+06	...
30	40	1245.35	3.888E+06	978.486	3.911E+06	...
Total		13242	80949800.00	10413	81288112.00	0

**Weight of Details**

From	Type	Weight of Detail kg	X Offset, Dtl. Cent. mm	Y Offset, Dtl. Cent. mm	Description
10	Noz1	88.1762	-750.000	...	GV1a
10	Noz1	22.4052	-509.902	...	R1
10	Noz1	18.7288	-450.000	...	R3
20	Sad1	851.391	850.000	1736.00	left sd1
20	Sad1	851.391	9600.00	1736.00	right sd1
20	Noz1	88.1762	500.000	1754.00	GV2a
20	Noz1	88.1762	7950.00	1754.00	GV2b
20	Noz1	1.07926	3860.00	1525.50	M1
20	Noz1	17.3716	6940.00	1584.15	R8
20	Noz1	17.3716	9220.00	1584.15	R2
20	Noz1	24.8760	9950.00	1609.55	R6
20	Noz1	18.5834	250.000	1584.15	R7c
20	Noz1	18.5834	2700.00	1584.15	R7b
20	Noz1	18.5834	7650.00	1584.15	R7a
20	Noz1	20.7100	8910.00	1609.55	R5
20	Noz1	17.3716	10100.0	1584.15	R4
20	Noz1	3.10788	8650.00	1525.50	R17b
20	Noz1	0.44871	10240.0	1525.50	R17d

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FileName : P3902-BC -----

Element and Detail Weights : Step: 5 7:08a Nov 7,2012

20 Noz1	8.76849	9100.00	1544.45	R11
20 Noz1	5.35114	8800.00	1544.45	R9
20 Noz1	6.91690	8950.00	1544.45	R18a
20 Noz1	6.45956	9100.00	1544.45	R10
20 Noz1	6.10067	8800.00	1544.45	R12
20 Noz1	2.45844	8650.00	1525.50	R17a
20 Noz1	8.76849	8920.00	1544.45	R20
30 Noz1	88.6208	800.000	...	GV1b
30 Noz1	28.8188	667.960	...	R19
30 Noz1	5.74814	713.325	700.000	R15
30 Noz1	7.02170	664.492	695.755	R18b
30 Noz1	7.02170	559.902	692.252	R16
30 Noz1	7.02170	567.350	406.823	R14
30 Noz1	4.26533	604.048	998.553	R13

**Total Weight of Each Detail Type**

Total Weight of Saddles	1702.8
Total Weight of Nozzles	657.1

-----  
Sum of the Detail Weights 2359.9 kg

**Weight Summary**

Fabricated Wt. - Bare Weight W/O Removable Internals	15602.6 kg
Shop Test Wt. - Fabricated Weight + Water ( Full )	96503.0 kg
Shipping Wt. - Fab. Wt + Rem. Intls.+ Shipping App.	15602.6 kg
Erected Wt. - Fab. Wt + Rem. Intls.+ Insul. (etc)	15602.6 kg
Ope. Wt. no Liq - Fab. Wt + Intls. + Details + Wghts.	15602.6 kg
Operating Wt. - Empty Wt + Operating Liq. Uncorroded	15602.6 kg
Oper. Wt. + CA - Corr Wt. + Operating Liquid	12773.4 kg
Field Test Wt. - Empty Weight + Water (Full)	96503.0 kg

Note: The Corroded Weight and thickness are used in the Horizontal

Vessel Analysis (Ope Case) and Earthquake Load Calculations.



**Outside Surface Areas of Elements**

		Surface	
From	To	Area	
		mm <sup>2</sup>	
-----			
10	20	10.41E+06	
20	30	98.46E+06	
30	40	10.41E+06	
-----			
Total		119285824.000	mm <sup>2</sup>

FileName : P3902-BC

Nozzle Flange MAWP :

Step: 6 7:08a Nov 7,2012

Nozzle Flange MAWP Results :

Nozzle	----- Flange Rating		Temperature	Class	Grade Group
Description	Operating	Ambient			
	MPa	MPa	C		
GV1a	6895.0	6895.0	60	10	GR 1.1
R1	6895.0	6895.0	60	16	GR 1.1
R3	6895.0	6895.0	60	16	GR 1.1
GV2a	6895.0	6895.0	60	10	GR 1.1
GV2b	6895.0	6895.0	60	10	GR 1.1
R8	6895.0	6895.0	60	16	GR 1.1
R2	6895.0	6895.0	60	16	GR 1.1
R6	6895.0	6895.0	60	16	GR 1.1
R7c	6895.0	6895.0	60	40	GR 1.1
R7b	6895.0	6895.0	60	40	GR 1.1
R7a	6895.0	6895.0	60	40	GR 1.1
R5	6895.0	6895.0	60	16	GR 1.1
R4	6895.0	6895.0	60	16	GR 1.1
R17b	6895.0	6895.0	60	40	GR 1.1
R17d	6895.0	6895.0	60	40	GR 1.1
R11	6895.0	6895.0	60	16	GR 1.1
R9	6895.0	6895.0	60	16	GR 1.1
R18a	6895.0	6895.0	60	16	GR 1.1
R10	6895.0	6895.0	60	16	GR 1.1
R12	6895.0	6895.0	60	16	GR 1.1
R17a	6895.0	6895.0	60	40	GR 1.1
R20	6895.0	6895.0	60	16	GR 1.1
GV1b	6895.0	6895.0	60	10	GR 1.1
R19	6895.0	6895.0	60	16	GR 1.1
R15	6895.0	6895.0	60	16	GR 1.1
R18b	6895.0	6895.0	60	16	GR 1.1
R16	6895.0	6895.0	60	16	GR 1.1
R14	6895.0	6895.0	60	16	GR 1.1

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FileName : P3902-BC -----

Nozzle Flange MAWP : Step: 6 7:08a Nov 7,2012

R13 6895.0 6895.0 60 16 GR 1.1

-----  
Minimum Rating 6895.000 6895.000 MPa

Note: ANSI Ratings are per ANSI/ASME B16.5 2003 Edition

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FileName : P3902-BC -----

Wind Load Calculation : Step: 7 7:08a Nov 7,2012

**Wind Analysis Results**

User Entered Importance Factor is	1.000	
Gust Factor (Gh, Gbar) Static	1.239	
Shape Factor (Cf) for the Vessel is	0.546	
User Entered Basic Wind Speed	112.7	km/hr
Exposure Category	C	
Table Lookup Value Alpha from Table C6	7.0000	
Table Lookup Value Zg from Table C6	900.0000	
Table Lookup Value Do from Table C6	0.0050	

Wind Load Results per ASCE-7 93:

Sample Calculation for the First Element:

Roughness Factor = 1.000

Values [cf1] and [cf2]

Because RoughFact = 1 and DQZ > 2.5 and H/D < 7.0

Interpolating to find the final cf:

Because H / D < 7.0

$$CF = CF1 + (CF2-CF1)*( H/D - 1) / (7 - 1)$$

$$= 0.500 + (0.600 - 0.500 )*( 3.758 - 1) / (7 - 1)$$

$$= 0.546$$

Value of Alpha, Zg is taken from Table C6-2 [Alpha, Zg]

For Exposure Category C:

Alpha = 7.000 , Zg = 274320.000 mm

Height of Interest for First Element [z]

= Centroid Hgt + Base Height

= 2000.000 + 0.000 = 2000.000 mm

but: z = Max(4572.000 , 2000.000 ) = 4572.000 mm

Note: Because z < 15 feet, use 15 feet to compute kz.

Velocity Pressure Coefficient [kZ]:

$$= 2.58( z/zg )^{2/Alpha} : z \text{ is Elevation of First Element}$$

$$= 2.58( 4572.000 /900 )^{2/7.0}$$

$$= 0.801$$

Determine if Static or Dynamic Gust Factor Applies

FileName : P3902-BC -----

Wind Load Calculation : Step: 7 7:08a Nov 7,2012

Height to Diameter ratio :

$$= \text{Maximum Height}(\text{length})^2 / \text{Sum of Area of the Elements}$$

$$= 11214.001 (^2) / 33459536$$

$$= 3.758$$

Vibration Frequency = 33.000 Hz

Because H/D < 5 And Frequency > 1.0: Static Analysis Implemented

The following two calculations allow for any user units

Compute [tz]

$$= 2.35 * \text{Sqrt}(\text{DO} / \text{VesselHtg}/30(\text{feet})^{1/\text{Alpha}}$$

$$= 2.35 * \text{Sqrt}(0.005 / 11214.001 )^{1/9144.000}$$

$$= 0.161$$

Compute [Gh]

$$= 0.65 + 3.65 * tz$$

$$= 0.65 + 3.65 * 0.161 = 1.239$$

Wind Pressure - (performed in Imperial Units) [qz]

Importance Factor: I = 1.000

Wind Speed = 112.651 km/hr Converts to 70.000 mph

$$qz = 0.00256 * kZ * (I * Vr)^2$$

$$= 0.00256 * 0.801 * (1.000 * 70.000 )^2 = 10.046 \text{ psf}$$

Converts to: 49.052 kg/m<sup>2</sup>

Force on the First Element [Fz]

$$= qz * Gh * CF * \text{Wind Area}$$

$$= 49.052 * 1.239 * 0.546 * 1998613$$

$$= 650.357 \text{ N}$$

Element	z	GH	Area	qz	Force
	mm		mm <sup>2</sup>	kg/m <sup>2</sup>	N
FUND ELIPSOIDAL	2000.0	1.239	1998613.9	49.1	650.4
VIROLA	2000.0	1.239	37607764.0	49.1	12237.7
FUND ELIPSOIDAL	2000.0	1.239	1998613.9	49.1	650.4

FileName : P3902-BC

Wind Load Calculation : Step: 7 7:08a Nov 7,2012

**Wind Load Calculation**

From	To	Wind Height mm	Wind Diameter mm	Wind Area mm <sup>2</sup>	Height Factor kg/m <sup>2</sup>	Element Wind Load N
10	20	2000.00	3633.60	1.999E+06	49.0518	650.357
20	30	2000.00	3633.60	37.61E+06	49.0518	12237.7
30	40	2000.00	3633.60	1.999E+06	49.0518	650.357

Note: Loads multiplied by the Scalar multiplier value of 0.7143

**Earthquake Analysis Results per UBC 1997**

The UBC Zone Factor for the Vessel is ..... 0.0000  
 The Importance Factor as Specified by the User is . 1.250  
 The UBC Force Factor as Specified by the User is .. 2.000  
 The UBC Total Weight (W) for the Vessel is ..... 125255.5 N  
 The UBC Total Shear (V) for the Vessel is ..... 28183.1 N  
 The UBC Seismic Coefficient Value Ca is ..... 0.360  
 The UBC Seismic Coefficient Value Cv is ..... 0.840

Note: The base shear printed above has been modified  
by the user defined Earthquake scalar.

**Calculation Steps for Computing the design Base Shear (V) per UBC 1997**

Computation of V per equation (34-1):

$$V = 0.7 * Ca * I * W$$

$$V = 0.7 * 0.360 * 1.250 * 125255$$

$$V = 39455.5 \text{ N}$$

Computation of V per equation (30-5):

$$V = 2.5 * Ca * I * W / R$$

$$V = 2.5 * 0.360 * 1.25 * 125255 / 2.000$$

$$V = 70456.2 \text{ N}$$

The computed base shear is the minimum of V from 34-1 and 30-5.

Computation of V per equation (34-2), minimum V. V cannot be less than this value !

$$V = 0.56 * Ca * I * W$$

$$V = 0.56 * 0.360 * 1.250 * 125255$$

$$V = 31564.4 \text{ N}$$

Total Adjusted Base Shear V:

$$= V * \text{Scalar Multiplier} = 39455.5 * 0.7143 = 28183.1 \text{ N}$$

Next Sum the earthquake weights times their heights (wi\*hi):

$$\text{Current Sum} = \text{Prev. Sum} + \text{Wght } 25051. * \text{Hght } 1500.000 = 37591892.$$

$$\text{Current Sum} = \text{Prev. Sum} + \text{Wght } 25051. * \text{Hght } 1500.000 = 75183784.$$

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Earthquake Load Calculation : Step: 8 7:08a Nov 7,2012

Current Sum = Prev. Sum + Wght 25051. \* Hght 1500.000 = 112775680.

Current Sum = Prev. Sum + Wght 25051. \* Hght 1500.000 = 150367568.

Current Sum = Prev. Sum + Wght 25051. \* Hght 1500.000 = 187959472.

Compute the load at each level based on equation 30-15 and multiply

by the load case scalar. The sum will be the total adjusted shear.

$$F_x = ( V * w_x * h_x / ( \text{sum of } ( w_i * h_i ) ) ) * EqFact$$

$$F_x = [ (39455.) * 25051. * 1500.000 / 187959472. ] * 0.7143 = 5637.$$

$$F_x = [ (39455.) * 25051. * 1500.000 / 187959472. ] * 0.7143 = 5637.$$

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$$F_x = [ (39455.) * 25051. * 1500.000 / 187959472. ] * 0.7143 = 5637.$$

**Earthquake Load Calculation**

From	To	Earthquake Height	Earthquake Weight	Element Ope Load
		mm	N	N
10	20	1500.00	25051.1	5636.61
20	Sad1	1500.00	25051.1	5636.61
Sad1	30	1500.00	25051.1	5636.61
20	30	1500.00	25051.1	5636.61
30	40	1500.00	25051.1	5636.61



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Center of Gravity Calculation : Step: 9 7:08a Nov 7,2012

**Shop/Field Installation Options :**

Note : The CG is computed from the first Element From Node

Center of Gravity of Saddles	5275.000 mm
Center of Gravity of Nozzles	5864.705 mm
Center of Gravity of Bare Shell New and Cold	5227.023 mm
Center of Gravity of Bare Shell Corroded	5227.021 mm
Vessel CG in the Operating Condition	5266.221 mm
Vessel CG in the Fabricated (Shop/Empty) Condition	5259.114 mm

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FileName : P3902-BC -----

Horizontal Vessel Analysis (Ope.) : Step: 10 7:08a Nov 7,2012

**ASME Horizontal Vessel Analysis: Stresses for the Left Saddle**

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Horizontal Vessel Stress Calculations : Operating Case

**Input and Calculated Values:**

Vessel Mean Radius	Rm	1508.50	mm
Stiffened Vessel Length per 4.15.6	L	10450.00	mm
Distance from Saddle to Vessel tangent	a	685.00	mm
Saddle Width	b	400.00	mm
Saddle Bearing Angle	theta	120.00	
Wear Plate Width	b1	500.00	mm
Wear Plate Bearing Angle	thetal	127.00	
Wear Plate Thickness	tr	14.0	mm
Wear Plate Allowable Stress	Sr	204.17	MPa
Inside Depth of Head	h2	753.00	mm
Shell Allowable Stress used in Calculation		204.17	MPa
Head Allowable Stress used in Calculation		204.17	MPa
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Operating Case		86763.61	N

**Horizontal Vessel Analysis Results:      Actual      Allowable**

-----			
Long. Stress at Top of Midspan	39.18	204.17	MPa
Long. Stress at Bottom of Midspan	43.10	204.17	MPa
Long. Stress at Top of Saddles	41.17	204.17	MPa
Long. Stress at Bottom of Saddles	41.11	204.17	MPa
Tangential Shear in Shell	4.60	163.34	MPa
Tangential Shear in Head	4.60	163.34	MPa
Circ. Stress at Horn of Saddle	4.62	255.22	MPa
Addl. Stress in Head as Stiffener	83.92	255.22	MPa
Circ. Compressive Stress in Shell	1.00	204.17	MPa

**Intermediate Results: Saddle Reaction Q due to Wind or Seismic**

Saddle Reaction Force due to Wind Ft [Fwt]:

$$= F_{tr} * ( Ft/Num\ of\ Saddles + Z\ Force\ Load ) * B / E$$

$$= 3.00 * ( 13538.4 / 2 + 0 ) * 2000.0000 / 2612.7986$$

$$= 15544.8\ N$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$= Max( Fl, Friction\ Load, Sum\ of\ X\ Forces ) * B / Ls$$

$$= Max( 4156.52 , 0.00 , 0 ) * 2000.0000 / 8750.0010$$

$$= 950.1\ N$$

Saddle Reaction Force due to Earthquake Fl or Friction [Fsl]:

$$= Max( Fl, Friction\ Force, Sum\ of\ X\ Forces ) * B / Ls$$

$$= Max( 28183.05 , 0.00 , 0 ) * 2000.0000 / 8750.0010$$

$$= 6441.8\ N$$

Saddle Reaction Force due to Earthquake Ft [Fst]:

$$= F_{tr} * ( Ft/Num\ of\ Saddles + Z\ Force\ Load ) * B / E$$

$$= 3.00 * ( 28183 / 2 + 0 ) * 2000.0000 / 2612.7986$$

$$= 32359.6\ N$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$= Saddle\ Load + Max( Fwl, Fwt, Fsl, Fst )$$

$$= 54403 + Max( 950 , 15544 , 6441 , 32359 )$$

$$= 86763.6\ N$$

**Summary of Loads at the base of this Saddle:**

Vertical Load (including saddle weight)	95112.34	N
Transverse Shear Load Saddle	14091.53	N
Longitudinal Shear Load Saddle	28183.05	N

**Formulas and Substitutions for Horizontal Vessel Analysis:**

Note: Wear Plate is Welded to the Shell, k = 0.1

**The Computed K values from Table 4.15.1:**

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0132	K8 = 0.3405
K9 = 0.2711	K10 = 0.0581	K1* = 0.1923	K6p = 0.0472
K7P = 0.0118			

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is less than Rm/2.

Moment per Equation 4.15.3 [M1]:

$$\begin{aligned}
 &= -Q*a [1 - (1 - a/L + (R^2 - h^2)/(2a*L))/(1 + (4h^2)/3L)] \\
 &= -86763*685.00[1 - (1 - 685.00/10450.00 + (1508.500^2 - 753.000^2)/ \\
 &\quad (2*685.00*10450.00))/(1 + (4*753.00)/(3*10450.00))] \\
 &= -2293745.0 \text{ N-mm}
 \end{aligned}$$

Moment per Equation 4.15.4 [M2]:

$$\begin{aligned}
 &= Q*L/4(1 + 2(R^2 - h^2)/(L^2))/(1 + (4h^2)/(3L)) - 4a/L \\
 &= 86763*10450/4(1 + 2(1508^2 - 753^2)/(10450^2))/(1 + (4*753)/ \\
 &\quad (3*10450)) - 4*684/10450 \\
 &= 153901664.0 \text{ N-mm}
 \end{aligned}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$\begin{aligned}
 &= P * Rm/(2t) - M2/(pi*Rm^2*t) \\
 &= 0.60 * 1508.500 / (2*11.00) - .15390E+09/(pi*1508.5^2*11.00) \\
 &= 39.18 \text{ MPa}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned}
 &= P * Rm/(2t) + M2/(pi * Rm^2 * t) \\
 &= 0.60 * 1508.500 / (2 * 11.00) + .15390E+09/(pi * 1508.5^2 * 11.00) \\
 &= 43.10 \text{ MPa}
 \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.8) [Sigma3]:

$$\begin{aligned}
 &= P * Rm/(2t) - M1/(pi * Rm^2 * t) \\
 &= 0.60 * 1508.500 / (2 * 11.00) - -2293745.0/(pi * 1508.5^2 * 11.00) \\
 &= 41.17 \text{ MPa}
 \end{aligned}$$

Longitudinal Stress at bottom of Shell at Support (4.15.9) [Sigma4]:

$$\begin{aligned}
 &= P * Rm/(2t) + M1/(pi*Rm^2*t) \\
 &= 0.60 * 1508.500 / (2*11.00) + -2293745.0/(pi*1508.5^2*11.00) \\
 &= 41.11 \text{ MPa}
 \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
 &= Q(L - 2a)/(L + (4*h^2/3)) \\
 &= 86763 ( 10450.00 - 2 * 685.00 ) / (10450.00 + ( 4 * 753.00 / 3)) \\
 &= 68780.7 \text{ N}
 \end{aligned}$$

Shear Stress in the shell no rings, stiffened (4.15.15) [tau3]:

$$= K3 * Q / ( Rm * t )$$

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Horizontal Vessel Analysis (Ope.) : Step: 10 7:08a Nov 7,2012

$$= 0.8799 * 86763 / ( 1508.5000 * 11.0000 )$$

$$= 4.60 \text{ MPa}$$

Shear Stress in the head, shell stiffened (4.15.16) [ $\tau_3$ ]:

$$= K3 * Q / ( Rm * th )$$

$$= 0.8799 * 86763 / ( 1508.5000 * 11.0000 )$$

$$= 4.60 \text{ MPa}$$

Membrane stress in the head as a stiffener (4.15.18) [ $\sigma_5$ ]:

$$= K4 * Q / (Rm*th) + ((P*Ri)/(2*th)) * (Ri/h2)$$

$$= 0.4011 * 86763 / (1508.50 * 11.000 ) +$$

$$((0.60 * 1503.00 ) / (2 * 11.000 )) * (1503.00 / 753.00 )$$

$$= 83.92 \text{ MPa}$$

Decay Length (4.15.22) [ $x_1, x_2$ ]:

$$= 0.78 * \text{sqrt}( Rm * t )$$

$$= 0.78 * \text{sqrt}( 1508.500 * 11.000 )$$

$$= 100.476 \text{ mm}$$

Circumferential Stress in shell, no rings (4.15.23) [ $\sigma_6$ ]:

$$= -K5 * Q * k / ( t * ( b + X1 + X2 ) )$$

$$= -0.7603 * 86763 * 0.1 / ( 11.000 * ( 400.00 + 100.48 + 100.48 ) )$$

$$= -1.00 \text{ MPa}$$

Effective reinforcing plate width (4.15.1) [ $B_1$ ]:

$$= \text{min}( b + 1.56 * \text{sqrt}( Rm * t ), 2a )$$

$$= \text{min}( 400.00 + 1.56 * \text{sqrt}( 1508.500 * 11.000 ), 2 * 685.000 )$$

$$= 600.95 \text{ mm}$$

Wear Plate/Shell Stress ratio (4.15.29) [ $\eta$ ]:

$$= \text{min}( Sr/S, 1 )$$

$$= \text{min}( 204.174 / 204.174 , 1 )$$

$$= 1.0000$$

Circumferential Stress at wear plate (4.15.26) [ $\sigma_{6,r}$ ]:

$$= -K5 * Q * k / ( B1( t + \eta * tr ) )$$

$$= -0.7603 * 86763 * 0.1 / ( 600.953 ( 11.000 + 1.000 * 14.000 ) )$$

$$= -0.44 \text{ MPa}$$

Circ. Comp. Stress at Horn of Saddle,  $L < 8Rm$  (4.15.28) [ $\sigma_{7,r}$ ]:

$$= -Q / (4(t+\eta*tr)b1) - 12*K7*Q*Rm / (L(t+\eta*tr)^2)$$

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Horizontal Vessel Analysis (Ope.) : Step: 10 7:08a Nov 7,2012

$$= -86763 / (4(11.000 + 1.000 * 14.000 )600.953 ) -$$

$$12*0.013*86763*1508.500 / (10450.00(11.000+1.000*14.000)^2)$$

$$= -4.62 \text{ MPa}$$

**Results for Vessel Ribs, Web and Base:**

Baseplate Length	Bplen	2400.0000	mm
Baseplate Thickness	Bpthk	20.0000	mm
Baseplate Width	Bpwid	400.0000	mm
Number of Ribs ( inc. outside ribs )	Nribs	4	
Rib Thickness	Ribtk	20.0000	mm
Web Thickness	Webtk	20.0000	mm
Web Location	Webloc	Center	

Moment of Inertia of Saddle - Lateral Direction

	Y	A	AY	Io
Shell	6.	7706.	42385.	310826.
Wearplate	18.	7000.	126000.	2382331.
Web	227.	8060.	1825590.	522580448.
BasePlate	438.	8000.	3504000.	1535017472.
Totals	688.	30766.	5497976.	2060291072.

Value C1 = Sumof(Ay)/Sumof(A) = 179. mm

Value I = Sumof(Io) - C1\*Sumof(Ay) = 1077801472. mm\*\*4

Value As = Sumof(A) - Ashell = 23060. mm<sup>2</sup>

K1 = (1+Cos(beta)-.5\*Sin(beta)<sup>2</sup> )/(pi-beta+Sin(beta)\*Cos(beta)) = 0.2035

Fh = K1 \* Q = 0.2035 \* 86763.609 = 17658.2793 N

Tension Stress, St = ( Fh/As ) = 0.7658 MPa

Allowed Stress, Sa = 0.6 \* Yield Str = 143.9676 MPa

d = B - R\*Sin(theta) / theta = 708.0291 mm

Bending Moment, M = Fh \* d = 12507644.0000 N-mm

Bending Stress, Sb = ( M \* C1 / I ) = 2.0731 MPa

Allowed Stress, Sa = 2/3 \* Yield Str = 159.9640 MPa

**Minimum Thickness of Baseplate per Moss :**

$$= ( 3 * ( Q + Saddle\_Wt ) * BasePlateWidth / ( 4 * BasePlateLength *$$

$$AllStress ))^{1/2}$$

$$= ( 3 * (86763 + 8348 ) * 400.00 / ( 4 * 2400.000 * 159.964 ))^{1/2}$$

FileName : P3902-BC -----

Horizontal Vessel Analysis (Ope.) : Step: 10 7:08a Nov 7,2012

$$= 8.621 \text{ mm}$$

Calculation of Axial Load, Intermediate Values and Compressive Stress

Effective Baseplate Length [e]:

$$= ( Bplen - Clearance ) / ( Nr ribs - 1 )$$

$$= ( 2400.0000 - 25.4 ) / ( 4 - 1 ) = 791.5333 \text{ mm}$$

Baseplate Pressure Area [Ap]:

$$= e * Bpwid / 2$$

$$= 791.5333 * 400.0000 / 2 = 0.2E+06 \text{ mm}^2$$

Axial Load [P]:

$$= Ap * Bp$$

$$= 158306.7 * 0.09 = 14307.6 \text{ N}$$

Area of the Rib and Web [Ar]:

$$= ( Bpwid - Clearance - Webtk ) * Ribtk + e/2 * Webtk$$

$$= ( 400.000 - 25.4 - 20.000 ) * 20.000 + 791.5333 / 2 * 20.000$$

$$= 15007.332 \text{ mm}^2$$

Compressive Stress [Sc]:

$$= P/Ar$$

$$= 14307.6 / 15007.3320 = 0.9535 \text{ MPa}$$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	Y	A	AY	Ay <sup>2</sup>	Io
Rib	200.0	7346.0	1469200.0	0.0	96825736.0
Web	200.0	7915.3	1583066.5	0.0	527688.2
Values	200.0	15261.3	3052266.5	0.0	97353424.0

Bending Moment [Rm]:

$$= Fl / ( 2 * Bplen ) * e * rl / 2$$

$$= 28183.1 / ( 2 * 2400.00 ) * 791.533 * 1194.00 / 2$$

$$= 2775661 \text{ N-mm}$$

KL/R < Cc ( 14.8245 < 128.2550 ) per AISC E2-1

$$Sca = ( 1 - (Klr)^2 / (2 * Cc^2) ) * Fy / ( 5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3) )$$

$$Sca = ( 1 - ( 14.82 )^2 / ( 2 * 128.25^2 ) ) * 239 /$$

$$( 5/3 + 3 * ( 14.82 ) / ( 8 * 128.25 ) - ( 14.82^3 ) / ( 8 * 128.25^3 ) )$$

$$Sca = 139.40 \text{ MPa}$$

**AISC Unity Check on Outside Ribs ( must be <= 1.0 )**

$$\text{Check} = Sc/Sca + (Rm/Z)/Sba$$

$$\text{Check} = 0.95 / 139.40 + (2775661 / 486767.094) / 159.96$$

$$\text{Check} = 0.04$$

Check of Inside Ribs

Inertia of Saddle, Inner Ribs - Axial Direction

	Y	A	AY	Ay <sup>2</sup>	Io
Rib	187.3	7092.0	1328331.4	0.0	87609584.0
Web	187.3	15830.7	2965083.8	0.0	527688.2
Values	187.3	22922.7	4293415.0	0.0	88137272.0

$$KL/R < Cc \quad ( 7.6626 < 128.2550 ) \text{ per AISC E2-1}$$

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = ( 1 - ( 7.66 )^2 / ( 2 * 128.25^2 ) ) * 239 / ( 5/3 + 3 * ( 7.66 ) / ( 8 * 128.25 ) - ( 7.66^3 ) / ( 8 * 128.25^3 ) )$$

$$Sca = 141.81 \text{ MPa}$$

**AISC Unity Check on Inside Ribs ( must be <= 1.0 )**

$$\text{Check} = Sc/Sca + (Rm/Z)/Sba$$

$$\text{Check} = 1.25 / 141.81 + ( 2209106 / 470567.375 ) / 159.96$$

$$\text{Check} = 0.04$$

**Input Data for Base Plate Bolting Calculations:**

Total Number of Bolts per BasePlate	Nbolts	4
Total Number of Bolts in Tension/Baseplate	Nbt	4
Bolt Material Specification	C45E	N
Bolt Allowable Stress	Stba	113.34 MPa
Bolt Corrosion Allowance	Bca	0.0000 mm
Distance from Bolts to Edge	Edgedis	99.9998 mm
Nominal Bolt Diameter	Bnd	30.0000 mm
Thread Series	Series	TEMA Metric
BasePlate Allowable Stress	S	204.17 MPa
Area Available in a Single Bolt	BltArea	502.9650 mm <sup>2</sup>
Saddle Load QO (Weight)	QO	62752.7 N
Saddle Load QL (Wind/Seismic contribution)	QL	6441.8 N



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Horizontal Vessel Analysis (Ope.) : Step: 10 7:08a Nov 7,2012

Maximum Transverse Force	Ft	14091.5	N
Maximum Longitudinal Force	F1	28183.1	N
Saddle Bolted to Steel Foundation			No

**Bolt Area Calculation per Dennis R. Moss**

Bolt Area Requirement Due to Longitudinal Load [Bltarearl]:

$$= 0.0 \text{ (} QO > QL \text{ --> No Uplift in Longitudinal direction)}$$

Bolt Area due to Shear Load [Bltarears]:

$$= F1 / (Stba * Nbolts)$$

$$= 28183.05 / (113.34 * 4.00 )$$

$$= 62.1718 \text{ mm}^2$$

Bolt Area due to Transverse Load

Moment on Baseplate Due to Transverse Load [Rmom]:

$$= B * Ft + \text{Sum of X Moments}$$

$$= 2000.00 * 14091.53 + 0.00$$

$$= 28194480.00 \text{ N-mm}$$

Eccentricity (e):

$$= Rmom / QO$$

$$= 28194480 / 62752.73$$

$$= 449.11 \text{ mm} > Bplen/6 \text{ --> Uplift in Transverse direction}$$

$$f = Bplen / 2 - Edgedis$$

$$= 2400.00 / 2 - 100.00$$

$$= 1100.00 \text{ mm}$$

Modular Ratio Of Steel/Concrete (n1):

$$= ES / EC$$

$$= 203402.50 / 21526.32$$

$$= 9.45$$

$$K1 = 3 (e - 0.5 * Bplen)$$

$$= 3 (449.11 - 0.5*2400.00 )$$

$$= -2252.66 \text{ mm}$$

$$K2 = 6 * n1 * At / Bpwid * (f + e)$$

$$= 6 * 9.45 * 2011.86 / 400.00 * (1100.00 + 449.11 )$$

$$= 441731.77 \text{ mm}^2$$

$$\begin{aligned}
 K3 &= -K2 * (0.5 * Bplen + f) \\
 &= -441731.78 * (0.5 * 2400.00 + 1100.00 ) \\
 &= -1015983110.23 \text{ mm}^3
 \end{aligned}$$

Iteratively Solving for the Effective Bearing Length:

$$\begin{aligned}
 Y^3 + K1 * Y^2 + K2 * Y + K3 &= 0 \\
 Y^3 + -2252.66 * Y^2 + 441731.78 * Y + -0.1E+10 &= 0 \\
 Y &= 2256.44 \text{ mm}
 \end{aligned}$$

$$\begin{aligned}
 \text{Num} &= (Bplen / 2 - Y / 3 - e) \\
 &= (2400.00 / 2 - 2256.44 / 3 - 449.11 ) \\
 &= -1.26
 \end{aligned}$$

$$\begin{aligned}
 \text{Denom} &= (Bplen / 2 - Y / 3 + f) \\
 &= (2400.00 / 2 - 2256.44 / 3 + 1100.00 ) \\
 &= 1547.85
 \end{aligned}$$

Total Bolt Tension Force [Tforce]:

$$\begin{aligned}
 &= - QO * \text{Num} / \text{Denom} \\
 &= - 62752.73 * -1.26 / 1547.85 \\
 &= 51.07 \text{ N}
 \end{aligned}$$

Bolt Area Required due to Transverse Load [Bltareart]

$$\begin{aligned}
 &= Tforce / (Stba * Nbt) \\
 &= 51.07 / ( 113.34 * 4.00 ) \\
 &= 0.1127 \text{ mm}^2
 \end{aligned}$$

Required of a Single Bolt [Bltarear]

$$\begin{aligned}
 &= \max[\text{Bltarearl}, \text{Bltarears}, \text{Bltareart}] \\
 &= \max[0.0000 , 62.1718 , 0.1127 ] \\
 &= 62.1718 \text{ mm}^2
 \end{aligned}$$

**Baseplate Thickness Calculation per D. Moss:**

Bearing Pressure (fc)

$$\begin{aligned}
 &= 2 * (QO + Tforce) / (Y * Bplen) \\
 &= 2 * (62752.73 + 51.07 ) / (2256.44 * 2400.00 ) \\
 &= 0.02 \text{ MPa}
 \end{aligned}$$

Distance from Baseplate Edge to the Web [ADIST]:

$$\begin{aligned}
 &= (Bplen - Weblength) / 2 \\
 &= (2400.00 - 2349.20 ) / 2
 \end{aligned}$$

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Horizontal Vessel Analysis (Ope.) : Step: 10 7:08a Nov 7,2012

$$= 25.4000 \text{ mm}$$

Overturning Moment due To Bolt Tension [Mt]:

$$= T_{force} * A_{dist}$$

$$= 51.07 * 25.40$$

$$= 1297.73 \text{ N-mm}$$

Equivalent Bearing Pressure (f1):

$$= f_c * (Y - A_{dist}) / Y$$

$$= 0.02 * (2256.44 - 25.40) / 2256.44$$

$$= 0.02 \text{ MPa}$$

Overturning Moment due to Bearing Pressure [Mc]:

$$= (A_{dist}^2 * B_{pwid} / 6) * (f_1 + 2 * f_c)$$

$$= (25.40^2 * 400.00 / 6) * (0.02 + 2 * 0.02)$$

$$= 2982.78 \text{ N-mm}$$

Baseplate Required Thickness [Treq]:

$$= (6 * \max(M_t, M_c) / (B_{pwid} * S_{ba}))^{1/2}$$

$$= (6 * \max(1297.73, 2982.78) / (400.00 * 306.26))^{1/2}$$

$$= 0.3822 \text{ mm}$$

**ASME Horizontal Vessel Analysis: Stresses for the Right Saddle**

(per ASME Sec. VIII Div. 2 based on the Zick method.)

**Input and Calculated Values:**

Vessel Mean Radius	Rm	1508.50	mm
Stiffened Vessel Length per 4.15.6	L	10450.00	mm
Distance from Saddle to Vessel tangent	a	685.00	mm
Saddle Width	b	400.00	mm
Saddle Bearing Angle	theta	120.00	
Wear Plate Width	b1	500.00	mm
Wear Plate Bearing Angle	thetal	127.00	
Wear Plate Thickness	tr	14.0	mm
Wear Plate Allowable Stress	Sr	204.17	MPa
Inside Depth of Head	h2	753.00	mm
Shell Allowable Stress used in Calculation		204.17	MPa
Head Allowable Stress used in Calculation		204.17	MPa
Circumferential Efficiency in Plane of Saddle		1.00	

FileName : P3902-BC -----

Horizontal Vessel Analysis (Ope.) : Step: 10 7:08a Nov 7,2012

Circumferential Efficiency at Mid-Span 1.00  
 Saddle Force Q, Operating Case 86513.67 N

**Horizontal Vessel Analysis Results: Actual Allowable**

	Actual	Allowable	
Long. Stress at Top of Midspan	39.19	204.17	MPa
Long. Stress at Bottom of Midspan	43.09	204.17	MPa
Long. Stress at Top of Saddles	41.17	204.17	MPa
Long. Stress at Bottom of Saddles	41.11	204.17	MPa
Tangential Shear in Shell	4.59	163.34	MPa
Tangential Shear in Head	4.59	163.34	MPa
Circ. Stress at Horn of Saddle	4.61	255.22	MPa
Addl. Stress in Head as Stiffener	83.91	255.22	MPa
Circ. Compressive Stress in Shell	1.00	204.17	MPa

**Intermediate Results: Saddle Reaction Q due to Wind or Seismic**

Saddle Reaction Force due to Wind Ft [Fwt]:

$$= F_{tr} * ( Ft/Num\ of\ Saddles + Z\ Force\ Load ) * B / E$$

$$= 3.00 * ( 13538.4 / 2 + 0 ) * 2000.0000 / 2612.7986$$

$$= 15544.8\ N$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$= Max( Fl, Friction\ Load, Sum\ of\ X\ Forces ) * B / Ls$$

$$= Max( 4156.52 , 0.00 , 0 ) * 2000.0000 / 8750.0010$$

$$= 950.1\ N$$

Saddle Reaction Force due to Earthquake Fl or Friction [Fsl]:

$$= Max( Fl, Friction\ Force, Sum\ of\ X\ Forces ) * B / Ls$$

$$= Max( 28183.05 , 0.00 , 0 ) * 2000.0000 / 8750.0010$$

$$= 6441.8\ N$$

Saddle Reaction Force due to Earthquake Ft [Fst]:

$$= F_{tr} * ( Ft/Num\ of\ Saddles + Z\ Force\ Load ) * B / E$$

$$= 3.00 * ( 28183 / 2 + 0 ) * 2000.0000 / 2612.7986$$

$$= 32359.6\ N$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$= Saddle\ Load + Max( Fwl, Fwt, Fsl, Fst )$$

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$$= 54154 + \text{Max}( 950 , 15544 , 6441 , 32359 )$$

$$= 86513.7 \text{ N}$$

**Summary of Loads at the base of this Saddle:**

Vertical Load (including saddle weight)	94862.41	N
Transverse Shear Load Saddle	14091.53	N
Longitudinal Shear Load Saddle	28183.05	N

**Formulas and Substitutions for Horizontal Vessel Analysis:**

Note: Wear Plate is Welded to the Shell,  $k = 0.1$

**The Computed K values from Table 4.15.1:**

$$\begin{aligned} K1 &= 0.1066 & K2 &= 1.1707 & K3 &= 0.8799 & K4 &= 0.4011 \\ K5 &= 0.7603 & K6 &= 0.0529 & K7 &= 0.0132 & K8 &= 0.3405 \\ K9 &= 0.2711 & K10 &= 0.0581 & K1^* &= 0.1923 & K6p &= 0.0472 \\ K7p &= 0.0118 \end{aligned}$$

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is less than  $Rm/2$ .

**Moment per Equation 4.15.3 [M1]:**

$$\begin{aligned} &= -Q \cdot a \left[ 1 - \left( 1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left( 1 + \frac{(4h^2)}{3L} \right) \right] \\ &= -86513 \cdot 685.00 \left[ 1 - \left( 1 - \frac{685.00}{10450.00} + \frac{(1508.500^2 - 753.000^2)}{(2 \cdot 685.00 \cdot 10450.00)} \right) / \left( 1 + \frac{(4 \cdot 753.00)}{(3 \cdot 10450.00)} \right) \right] \\ &= -2287137.5 \text{ N-mm} \end{aligned}$$

**Moment per Equation 4.15.4 [M2]:**

$$\begin{aligned} &= Q \cdot L / 4 \left( 1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left( 1 + \frac{(4h^2)}{(3L)} \right) - 4a/L \\ &= 86513 \cdot 10450 / 4 \left( 1 + 2 \frac{(1508^2 - 753^2)}{(10450^2)} \right) / \left( 1 + \frac{(4 \cdot 753)}{(3 \cdot 10450)} \right) - 4 \cdot 684 / 10450 \\ &= 153458320.0 \text{ N-mm} \end{aligned}$$

**Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:**

$$\begin{aligned} &= P \cdot Rm / (2t) - M2 / (\pi \cdot Rm^2 \cdot t) \\ &= 0.60 \cdot 1508.500 / (2 \cdot 11.00) - .15346E+09 / (\pi \cdot 1508.5^2 \cdot 11.00) \\ &= 39.19 \text{ MPa} \end{aligned}$$

**Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:**

$$\begin{aligned} &= P \cdot Rm / (2t) + M2 / (\pi \cdot Rm^2 \cdot t) \\ &= 0.60 \cdot 1508.500 / (2 \cdot 11.00) + .15346E+09 / (\pi \cdot 1508.5^2 \cdot 11.00) \\ &= 43.09 \text{ MPa} \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.8) [Sigma3]:

$$\begin{aligned}
 &= P * Rm / (2t) - M1 / (\pi * Rm^2 * t) \\
 &= 0.60 * 1508.500 / (2 * 11.00) - -2287137.5 / (\pi * 1508.5^2 * 11.00) \\
 &= 41.17 \text{ MPa}
 \end{aligned}$$

Longitudinal Stress at bottom of Shell at Support (4.15.9) [Sigma4]:

$$\begin{aligned}
 &= P * Rm / (2t) + M1 / (\pi * Rm^2 * t) \\
 &= 0.60 * 1508.500 / (2 * 11.00) + -2287137.5 / (\pi * 1508.5^2 * 11.00) \\
 &= 41.11 \text{ MPa}
 \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
 &= Q(L-2a) / (L + (4 * h^2 / 3)) \\
 &= 86513 ( 10450.00 - 2 * 685.00 ) / (10450.00 + ( 4 * 753.00 / 3)) \\
 &= 68582.5 \text{ N}
 \end{aligned}$$

Shear Stress in the shell no rings, stiffened (4.15.15) [tau3]:

$$\begin{aligned}
 &= K3 * Q / ( Rm * t ) \\
 &= 0.8799 * 86513 / ( 1508.5000 * 11.0000 ) \\
 &= 4.59 \text{ MPa}
 \end{aligned}$$

Shear Stress in the head, shell stiffened (4.15.16) [tau3\*]:

$$\begin{aligned}
 &= K3 * Q / ( Rm * th ) \\
 &= 0.8799 * 86513 / ( 1508.5000 * 11.0000 ) \\
 &= 4.59 \text{ MPa}
 \end{aligned}$$

Membrane stress in the head as a stiffener (4.15.18) [sigma5]:

$$\begin{aligned}
 &= K4 * Q / (Rm * th) + ((P * Ri) / (2 * th)) * (Ri / h^2) \\
 &= 0.4011 * 86513 / (1508.50 * 11.000) + \\
 &\quad ((0.60 * 1503.00) / (2 * 11.000)) * (1503.00 / 753.00) \\
 &= 83.91 \text{ MPa}
 \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned}
 &= 0.78 * \text{sqrt}( Rm * t ) \\
 &= 0.78 * \text{sqrt}( 1508.500 * 11.000 ) \\
 &= 100.476 \text{ mm}
 \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned}
 &= -K5 * Q * k / ( t * ( b + X1 + X2 ) ) \\
 &= -0.7603 * 86513 * 0.1 / ( 11.000 * ( 400.00 + 100.48 + 100.48 ) )
 \end{aligned}$$

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= -1.00 MPa

Effective reinforcing plate width (4.15.1) [B1]:

$$= \min( b + 1.56 * \text{sqrt}( Rm * t ), 2a )$$

$$= \min( 400.00 + 1.56 * \text{sqrt}( 1508.500 * 11.000 ), 2 * 685.000 )$$

= 600.95 mm

Wear Plate/Shell Stress ratio (4.15.29) [eta]:

$$= \min( Sr/S, 1 )$$

$$= \min( 204.174 / 204.174 , 1 )$$

= 1.0000

Circumferential Stress at wear plate (4.15.26) [sigma6,r]:

$$= -K5 * Q * k / ( B1( t + eta * tr ) )$$

$$= -0.7603 * 86513 * 0.1 / ( 600.953 ( 11.000 + 1.000 * 14.000 ) )$$

= -0.44 MPa

Circ. Comp. Stress at Horn of Saddle, L<8Rm (4.15.28) [sigma7,r\*]:

$$= -Q/(4(t+eta*tr)b1) - 12*K7*Q*Rm/(L(t+eta*tr)^2)$$

$$= -86513 / (4(11.000 + 1.000 * 14.000 )600.953 ) -$$

$$12*0.013*86513*1508.500/(10450.00(11.000+1.000*14.000)^2)$$

= -4.61 MPa

**Results for Vessel Ribs, Web and Base**

Baseplate Length	Bplen	2400.0000	mm
Baseplate Thickness	Bpthk	20.0000	mm
Baseplate Width	Bpwid	400.0000	mm
Number of Ribs ( inc. outside ribs )	Nribs	4	
Rib Thickness	Ribtk	20.0000	mm
Web Thickness	Webtk	20.0000	mm
Web Location	Webloc	Center	

Moment of Inertia of Saddle - Lateral Direction

	Y	A	AY	Io
Shell	6.	7706.	42385.	310826.
Wearplate	18.	7000.	126000.	2382331.
Web	227.	8060.	1825590.	522580448.
BasePlate	438.	8000.	3504000.	1535017472.
Totals	688.	30766.	5497976.	2060291072.

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Value C1 = Sumof(Ay)/Sumof(A) = 179. mm  
 Value I = Sumof(Io) - C1\*Sumof(Ay) = 1077801472. mm\*\*4  
 Value As = Sumof(A) - Ashell = 23060. mm<sup>2</sup>  
 $K1 = (1 + \cos(\beta) - .5 * \sin(\beta)^2) / (\pi - \beta + \sin(\beta) * \cos(\beta)) = 0.2035$   
 $Fh = K1 * Q = 0.2035 * 86513.672 = 17607.4102 \text{ N}$   
 Tension Stress, St = ( Fh/As ) = 0.7636 MPa  
 Allowed Stress, Sa = 0.6 \* Yield Str = 143.9676 MPa  
 $d = B - R * \sin(\theta) / \theta = 708.0291 \text{ mm}$   
 Bending Moment, M = Fh \* d = 12471613.0000 N-mm  
 Bending Stress, Sb = ( M \* C1 / I ) = 2.0671 MPa  
 Allowed Stress, Sa = 2/3 \* Yield Str = 159.9640 MPa

**Minimum Thickness of Baseplate per Moss :**

$= ( 3 * ( Q + \text{Saddle\_Wt} ) * \text{BasePlateWidth} / ( 4 * \text{BasePlateLength} * \text{AllStress} ) )^{1/2}$   
 $= ( 3 * (86513 + 8348) * 400.00 / ( 4 * 2400.000 * 159.964 ) )^{1/2}$   
 = 8.610 mm

Calculation of Axial Load, Intermediate Values and Compressive Stress

**Effective Baseplate Length [e]:**

$= ( \text{Bplen} - \text{Clearance} ) / ( \text{Nr ribs} - 1 )$   
 $= ( 2400.0000 - 25.4 ) / ( 4 - 1 ) = 791.5333 \text{ mm}$

**Baseplate Pressure Area [Ap]:**

$= e * \text{Bpwid} / 2$   
 $= 791.5333 * 400.0000 / 2 = 0.2E+06 \text{ mm}^2$

**Axial Load [P]:**

$= Ap * Bp$   
 $= 158306.7 * 0.09 = 14266.3 \text{ N}$

**Area of the Rib and Web [Ar]:**

$= ( \text{Bpwid} - \text{Clearance} - \text{Webtk} ) * \text{Ribtk} + e/2 * \text{Webtk}$   
 $= ( 400.000 - 25.4 - 20.000 ) * 20.000 + 791.5333 / 2 * 20.000$   
 = 15007.332 mm<sup>2</sup>

**Compressive Stress [Sc]:**

$= P/Ar$   
 $= 14266.3 / 15007.3320 = 0.9507 \text{ MPa}$



Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	Y	A	AY	Ay <sup>2</sup>	Io
Rib	200.0	7346.0	1469200.0	0.0	96825736.0
Web	200.0	7915.3	1583066.5	0.0	527688.2
Values	200.0	15261.3	3052266.5	0.0	97353424.0

**Bending Moment [Rm]:**

$$= Fl / ( 2 * Bplen ) * e * rl / 2$$

$$= 28183.1 / ( 2 * 2400.00 ) * 791.533 * 1194.00 / 2$$

$$= 2775661 \text{ N-mm}$$

KL/R < Cc ( 14.8245 < 128.2550 ) per AISC E2-1

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = ( 1 - ( 14.82 )^2 / ( 2 * 128.25^2 ) ) * 239 /$$

$$( 5/3 + 3 * ( 14.82 ) / ( 8 * 128.25 ) - ( 14.82^3 ) / ( 8 * 128.25^3 ) )$$

$$Sca = 139.40 \text{ MPa}$$

**AISC Unity Check on Outside Ribs ( must be <= 1.0 )**

$$\text{Check} = Sc / Sca + (Rm / Z) / Sba$$

$$\text{Check} = 0.95 / 139.40 + (2775661 / 486767.094) / 159.96$$

$$\text{Check} = 0.04$$

Check of Inside Ribs

Inertia of Saddle, Inner Ribs - Axial Direction

	Y	A	AY	Ay <sup>2</sup>	Io
Rib	187.3	7092.0	1328331.4	0.0	87609584.0
Web	187.3	15830.7	2965083.8	0.0	527688.2
Values	187.3	22922.7	4293415.0	0.0	88137272.0

KL/R < Cc ( 7.6626 < 128.2550 ) per AISC E2-1

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = ( 1 - ( 7.66 )^2 / ( 2 * 128.25^2 ) ) * 239 /$$

$$( 5/3 + 3 * ( 7.66 ) / ( 8 * 128.25 ) - ( 7.66^3 ) / ( 8 * 128.25^3 ) )$$

$$Sca = 141.81 \text{ MPa}$$

**AISC Unity Check on Inside Ribs ( must be <= 1.0 )**

$$\text{Check} = Sc/Sca + (Rm/Z)/Sba$$

$$\text{Check} = 1.24 / 141.81 + ( 2209106 / 470567.375 ) / 159.96$$

$$\text{Check} = 0.04$$

**Input Data for Base Plate Bolting Calculations:**

Total Number of Bolts per BasePlate	Nbolts	4	
Total Number of Bolts in Tension/Baseplate	Nbt	4	
Bolt Material Specification		C45E	N
Bolt Allowable Stress	Stba	113.34	MPa
Bolt Corrosion Allowance	Bca	0.0000	mm
Distance from Bolts to Edge	Edgedis	99.9998	mm
Nominal Bolt Diameter	Bnd	30.0000	mm
Thread Series	Series	TEMA	Metric
BasePlate Allowable Stress	S	204.17	MPa
Area Available in a Single Bolt	BltArea	502.9650	mm <sup>2</sup>
Saddle Load QO (Weight)	QO	62502.8	N
Saddle Load QL (Wind/Seismic contribution)	QL	6441.8	N
Maximum Transverse Force	Ft	14091.5	N
Maximum Longitudinal Force	F1	28183.1	N
Saddle Bolted to Steel Foundation		No	

**Bolt Area Calculation per Dennis R. Moss**

Bolt Area Requirement Due to Longitudinal Load [Bltarear1]:

$$= 0.0 \text{ (} QO > QL \text{ --> No Uplift in Longitudinal direction)}$$

Bolt Area due to Shear Load [Bltarears]:

$$= F1 / (Stba * Nbolts)$$

$$= 28183.05 / (113.34 * 4.00)$$

$$= 62.1718 \text{ mm}^2$$

Bolt Area due to Transverse Load

Moment on Baseplate Due to Transverse Load [Rmom]:

$$\begin{aligned}
 &= B * Ft + \text{Sum of X Moments} \\
 &= 2000.00 * 14091.53 + 0.00 \\
 &= 28194480.00 \text{ N-mm}
 \end{aligned}$$

Eccentricity (e):

$$\begin{aligned}
 &= Rmom / QO \\
 &= 28194480 / 62502.80 \\
 &= 450.91 \text{ mm} > Bplen/6 \text{ --> Uplift in Transverse direction} \\
 f &= Bplen / 2 - Edgedis \\
 &= 2400.00 / 2 - 100.00 \\
 &= 1100.00 \text{ mm}
 \end{aligned}$$

Modular Ratio Of Steel/Concrete (n1):

$$\begin{aligned}
 &= ES / EC \\
 &= 203402.50 / 21526.32 \\
 &= 9.45
 \end{aligned}$$

$$\begin{aligned}
 K1 &= 3 (e - 0.5 * Bplen) \\
 &= 3 (450.91 - 0.5 * 2400.00) \\
 &= -2247.27 \text{ mm}
 \end{aligned}$$

$$\begin{aligned}
 K2 &= 6 * n1 * At / Bpwid * (f + e) \\
 &= 6 * 9.45 * 2011.86 / 400.00 * (1100.00 + 450.91) \\
 &= 442243.87 \text{ mm}^2
 \end{aligned}$$

$$\begin{aligned}
 K3 &= -K2 * (0.5 * Bplen + f) \\
 &= -442243.84 * (0.5 * 2400.00 + 1100.00) \\
 &= -1017160927.88 \text{ mm}^3
 \end{aligned}$$

Iteratively Solving for the Effective Bearing Length:

$$\begin{aligned}
 Y^3 + K1 * Y^2 + K2 * Y + K3 &= 0 \\
 Y^3 + -2247.27 * Y^2 + 442243.84 * Y + -0.1E+10 &= 0
 \end{aligned}$$

$$Y = 2251.50 \text{ mm}$$

$$\begin{aligned}
 \text{Num} &= (Bplen / 2 - Y / 3 - e) \\
 &= (2400.00 / 2 - 2251.50 / 3 - 450.91) \\
 &= -1.41
 \end{aligned}$$

$$\begin{aligned}
 \text{Denom} &= (Bplen / 2 - Y / 3 + f) \\
 &= (2400.00 / 2 - 2251.50 / 3 + 1100.00) = 1549.50
 \end{aligned}$$

Total Bolt Tension Force [Tforce]:

$$= - QO * Num / Denom$$

$$= - 62502.80 * -1.41 / 1549.50$$

$$= 56.89 \text{ N}$$

Bolt Area Required due to Transverse Load [Bltareart]

$$= Tforce / (Stba * Nbt)$$

$$= 56.89 / ( 113.34 * 4.00 )$$

$$= 0.1255 \text{ mm}^2$$

Required of a Single Bolt [Bltarear]

$$= \max[\text{Bltarearl}, \text{Bltarears}, \text{Bltareart}]$$

$$= \max[0.0000 , 62.1718 , 0.1255 ]$$

$$= 62.1718 \text{ mm}^2$$

**Baseplate Thickness Calculation per D. Moss:**

Bearing Pressure (fc)

$$= 2 * (QO + Tforce) / (Y * Bplen)$$

$$= 2 * (62502.80 + 56.89 ) / (2251.50 * 2400.00 )$$

$$= 0.02 \text{ MPa}$$

Distance from Baseplate Edge to the Web [ADIST]:

$$= (Bplen - Weblength) / 2$$

$$= (2400.00 - 2349.20 ) / 2$$

$$= 25.4000 \text{ mm}$$

Overturning Moment due To Bolt Tension [Mt]:

$$= Tforce * Adist$$

$$= 56.89 * 25.40$$

$$= 1445.50 \text{ N-mm}$$

Equivalent Bearing Pressure (f1):

$$= fc * (Y - Adist) / Y$$

$$= 0.02 * (2251.50 - 25.40 ) / 2251.50$$

$$= 0.02 \text{ MPa}$$

Overturning Moment due to Bearing Pressure [Mc]:

$$= (Adist^2 * Bpwid / 6) * (f1 + 2 * fc)$$

$$= (25.40^2 * 400.00 / 6) * (0.02 + 2 * 0.02 )$$

$$= 2977.68 \text{ N-mm}$$

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Baseplate Required Thickness [Treq]:

$$= (6 * \max(Mt, Mc) / (Bpwid * Sba))^{1/2}$$

$$= (6 * \max(1445.50, 2977.68 / (400.00 * 306.26))^{1/2}$$

$$= 0.3818 \text{ mm}$$

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**ASME Horizontal Vessel Analysis: Stresses for the Left Saddle**

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Horizontal Vessel Stress Calculations : Test Case

**Input and Calculated Values:**

Vessel Mean Radius	Rm	1507.00	mm
Stiffened Vessel Length per 4.15.6	L	10450.00	mm
Distance from Saddle to Vessel tangent	a	685.00	mm
Saddle Width	b	400.00	mm
Saddle Bearing Angle	theta	120.00	
Wear Plate Width	b1	500.00	mm
Wear Plate Bearing Angle	thetal	127.00	
Wear Plate Thickness	tr	14.0	mm
Wear Plate Allowable Stress	Sr	204.17	MPa
Inside Depth of Head	h2	750.00	mm
Shell Allowable Stress used in Calculation		204.17	MPa
Head Allowable Stress used in Calculation		204.17	MPa
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Test Case, no Ext. Forces		474750.47	N

**Horizontal Vessel Analysis Results:      Actual      Allowable**

-----			
Long. Stress at Top of Midspan	73.33	204.17	MPa
Long. Stress at Bottom of Midspan	90.19	204.17	MPa
Long. Stress at Top of Saddles	81.89	204.17	MPa
Long. Stress at Bottom of Saddles	81.64	204.17	MPa
Tangential Shear in Shell	19.80	163.34	MPa
Tangential Shear in Head	19.80	163.34	MPa
Circ. Stress at Horn of Saddle	20.61	255.22	MPa
Addl. Stress in Head as Stiffener	171.79	255.22	MPa
Circ. Compressive Stress in Shell	4.11	204.17	MPa

**Intermediate Results: Saddle Reaction Q due to Wind or Seismic**

Saddle Reaction Force due to Wind Ft [Fwt]:

$$= F_{tr} * ( Ft/Num\ of\ Saddles + Z\ Force\ Load ) * B / E$$

$$= 3.00 * ( 4467.7 / 2 + 0 ) * 2000.0000 / 2610.2007$$

$$= 5134.9\ N$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$= Max( F_l, Friction\ Load, Sum\ of\ X\ Forces ) * B / L_s$$

$$= Max( 4156.52 , 0.00 , 0 ) * 2000.0000 / 8750.0010$$

$$= 313.5\ N$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$= Saddle\ Load + Max( F_{wl}, F_{wt}, F_{sl}, F_{st} )$$

$$= 469615 + Max( 313 , 5134 , 0 , 0 )$$

$$= 474750.5\ N$$

**Summary of Loads at the base of this Saddle:**

Vertical Load (including saddle weight)	483099.19	N
Transverse Shear Load Saddle	2233.84	N
Longitudinal Shear Load Saddle	1371.65	N

Hydrostatic Test Pressure at center of Vessel: 1.519 MPa

**Formulas and Substitutions for Horizontal Vessel Analysis:**

Note: Wear Plate is Welded to the Shell, k = 0.1

The Computed K values from Table 4.15.1:

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0132	K8 = 0.3405
K9 = 0.2711	K10 = 0.0581	K1* = 0.1923	K6p = 0.0472
K7P = 0.0118			

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is less than Rm/2.

Moment per Equation 4.15.3 [M1]:

$$= -Q*a [ 1 - (1 - a/L + (R^2 - h^2)/(2a*L)) / (1 + (4h^2)/3L) ]$$

$$= -474750*685.00 [ 1 - (1 - 685.00/10450.00 + (1507.000^2 - 750.000^2) / (2*685.00*10450.00)) / (1 + (4*750.00)/(3*10450.00)) ]$$

$$= -12441862.0\ N-mm$$

Moment per Equation 4.15.4 [M2]:

$$= Q*L/4 (1 + 2(R^2 - h^2)/(L^2)) / (1 + (4h^2)/(3L)) - 4a/L$$

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$$= 474750 * 10450 / 4(1 + 2(1507^2 - 750^2) / (10450^2)) / (1 + (4 * 750) / (3 * 10450)) - 4 * 684 / 10450$$

$$= 842521984.0 \text{ N-mm}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$= P * Rm / (2t) - M2 / (\pi * Rm^2 * t)$$

$$= 1.52 * 1507.000 / (2 * 14.00) - .84252E+09 / (\pi * 1507.0^2 * 14.00)$$

$$= 73.33 \text{ MPa}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$= P * Rm / (2t) + M2 / (\pi * Rm^2 * t)$$

$$= 1.52 * 1507.000 / (2 * 14.00) + .84252E+09 / (\pi * 1507.0^2 * 14.00)$$

$$= 90.19 \text{ MPa}$$

Longitudinal Stress at Top of Shell at Support (4.15.8) [Sigma3]:

$$= P * Rm / (2t) - M1 / (\pi * Rm^2 * t)$$

$$= 1.52 * 1507.000 / (2 * 14.00) - -0.1E+08 / (\pi * 1507.0^2 * 14.00)$$

$$= 81.89 \text{ MPa}$$

Longitudinal Stress at bottom of Shell at Support (4.15.9) [Sigma4]:

$$= P * Rm / (2t) + M1 / (\pi * Rm^2 * t)$$

$$= 1.52 * 1507.000 / (2 * 14.00) + -0.1E+08 / (\pi * 1507.0^2 * 14.00)$$

$$= 81.64 \text{ MPa}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$= Q(L - 2a) / (L + (4 * h^2 / 3))$$

$$= 474750 (10450.00 - 2 * 685.00) / (10450.00 + (4 * 750.00 / 3))$$

$$= 376483.3 \text{ N}$$

Shear Stress in the shell no rings, stiffened (4.15.15) [tau3]:

$$= K3 * Q / (Rm * t)$$

$$= 0.8799 * 474750 / (1507.0000 * 14.0000)$$

$$= 19.80 \text{ MPa}$$

Shear Stress in the head, shell stiffened (4.15.16) [tau3\*]:

$$= K3 * Q / (Rm * th)$$

$$= 0.8799 * 474750 / (1507.0000 * 14.0000)$$

$$= 19.80 \text{ MPa}$$

Membrane stress in the head as a stiffener (4.15.18) [sigma5]:

$$= K4 * Q / (Rm * th) + ((P * Ri) / (2 * th)) * (Ri / h^2)$$



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$$= 0.4011 * 474750 / (1507.00 * 14.000 ) +$$

$$((1.52 * 1500.00 ) / (2 * 14.000 )) * (1500.00 / 750.00 )$$

$$= 171.79 \text{ MPa}$$

Decay Length (4.15.22) [x1,x2]:

$$= 0.78 * \text{sqrt}( Rm * t )$$

$$= 0.78 * \text{sqrt}( 1507.000 * 14.000 )$$

$$= 113.296 \text{ mm}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$= -K5 * Q * k / ( t * ( b + X1 + X2 ) )$$

$$= -0.7603 * 474750 * 0.1 / ( 14.000 * ( 400.00 + 113.30 + 113.30 ) )$$

$$= -4.11 \text{ MPa}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$= \text{min}( b + 1.56 * \text{sqrt}( Rm * t ), 2a )$$

$$= \text{min}( 400.00 + 1.56 * \text{sqrt}( 1507.000 * 14.000 ), 2 * 685.000 )$$

$$= 626.59 \text{ mm}$$

Wear Plate/Shell Stress ratio (4.15.29) [eta]:

$$= \text{min}( Sr/S, 1 )$$

$$= \text{min}( 204.174 / 204.174 , 1 )$$

$$= 1.0000$$

Circumferential Stress at wear plate (4.15.26) [sigma6,r]:

$$= -K5 * Q * k / ( B1( t + eta * tr ) )$$

$$= -0.7603 * 474750 * 0.1 / ( 626.592 ( 14.000 + 1.000 * 14.000 ) )$$

$$= -2.06 \text{ MPa}$$

Circ. Comp. Stress at Horn of Saddle, L<8Rm (4.15.28) [sigma7,r\*]:

$$= -Q / (4(t+eta*tr)b1) - 12*K7*Q*Rm / (L(t+eta*tr)^2)$$

$$= -474750 / (4(14.000 + 1.000 * 14.000 )626.592 ) -$$

$$12*0.013*474750*1507.000 / (10450.00(14.000+1.000*14.000)^2)$$

$$= -20.61 \text{ MPa}$$

**Results for Vessel Ribs, Web and Base:**

Baseplate Length	Bplen	2400.0000	mm
Baseplate Thickness	Bpthk	20.0000	mm
Baseplate Width	Bpwid	400.0000	mm

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Number of Ribs ( inc. outside ribs )	Nribs	4
Rib Thickness	Ribtk	20.0000 mm
Web Thickness	Webtk	20.0000 mm
Web Location	Webloc	Center

Moment of Inertia of Saddle - Lateral Direction

	Y	A	AY	Io
Shell	7.	10165.	71154.	664107.
Wearplate	21.	7000.	147000.	3201331.
Web	228.	8000.	1824000.	522538176.
BasePlate	438.	8000.	3504000.	1535017088.
Totals	694.	33165.	5546154.	2061420672.

Value C1 = Sumof(Ay)/Sumof(A) = 167. mm

Value I = Sumof(Io) - C1\*Sumof(Ay) = 1133940736. mm\*\*4

Value As = Sumof(A) - Ashell = 23000. mm<sup>2</sup>

K1 = (1+Cos(beta)-.5\*Sin(beta)<sup>2</sup> )/(pi-beta+Sin(beta)\*Cos(beta)) = 0.2035

Fh = K1 \* Q = 0.2035 \* 474750.469 = 96622.0234 N

Tension Stress, St = ( Fh/As ) = 4.2013 MPa

Allowed Stress, Sa = 0.6 \* Yield Str = 143.9676 MPa

d = B - R\*Sin(theta) / theta = 707.5100 mm

Bending Moment, M = Fh \* d = 68388768.0000 N-mm

Bending Stress, Sb = ( M \* C1 / I ) = 10.0825 MPa

Allowed Stress, Sa = 2/3 \* Yield Str = 159.9640 MPa

**Minimum Thickness of Baseplate per Moss :**

$$= ( 3 * ( Q + Saddle\_Wt ) * BasePlateWidth / ( 4 * BasePlateLength * AllStress ) )^{1/2}$$

$$= ( 3 * ( 474750 + 8348 ) * 400.00 / ( 4 * 2400.000 * 159.964 ) )^{1/2}$$

$$= 19.430 \text{ mm}$$

Calculation of Axial Load, Intermediate Values and Compressive Stress

Effective Baseplate Length [e]:

$$= ( Bplen - Clearance ) / ( Nribs - 1 )$$

$$= ( 2400.0000 - 25.4 ) / ( 4 - 1 ) = 791.5333 \text{ mm}$$

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Baseplate Pressure Area [Ap]:

$$= e * Bpwid / 2$$

$$= 791.5333 * 400.0000 / 2 = 0.2E+06 \text{ mm}^2$$

Axial Load [P]:

$$= Ap * Bp$$

$$= 158306.7 * 0.49 = 78287.7 \text{ N}$$

Area of the Rib and Web [Ar]:

$$= ( Bpwid - Clearance - Webtk ) * Ribtk + e/2 * Webtk$$

$$= ( 400.000 - 25.4 - 20.000 ) * 20.000 + 791.5333 / 2 * 20.000$$

$$= 15007.332 \text{ mm}^2$$

Compressive Stress [Sc]:

$$= P/Ar$$

$$= 78287.7 / 15007.3320 = 5.2171 \text{ MPa}$$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	Y	A	AY	Ay <sup>2</sup>	Io
Rib	200.0	7346.0	1469200.0	0.0	96825736.0
Web	200.0	7915.3	1583066.5	0.0	527688.2
Values	200.0	15261.3	3052266.5	0.0	97353424.0

Bending Moment [Rm]:

$$= Fl / ( 2 * Bplen ) * e * rl / 2$$

$$= 1371.7 / ( 2 * 2400.00 ) * 791.533 * 1191.00 / 2$$

$$= 134750.391 \text{ N-mm}$$

KL/R < Cc ( 14.7873 < 128.2550 ) per AISC E2-1

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = ( 1 - ( 14.79 )^2 / ( 2 * 128.25^2 ) ) * 239 /$$

$$( 5/3 + 3 * ( 14.79 ) / ( 8 * 128.25 ) - ( 14.79^3 ) / ( 8 * 128.25^3 ) )$$

$$Sca = 139.41 \text{ MPa}$$

**AISC Unity Check on Outside Ribs ( must be <= 1.0 )**

$$\text{Check} = Sc/Sca + (Rm/Z)/Sba$$

$$\text{Check} = 5.22 / 139.41 + (134750.39 / 486767.094) / 159.96$$

$$\text{Check} = 0.04$$

Check of Inside Ribs

Inertia of Saddle, Inner Ribs - Axial Direction

	Y	A	AY	Ay <sup>2</sup>	Io
Rib	187.3	7092.0	1328331.4	0.0	87609584.0
Web	187.3	15830.7	2965083.8	0.0	527688.2
Values	187.3	22922.7	4293415.0	0.0	88137272.0

KL/R < Cc ( 7.6142 < 128.2550 ) per AISC E2-1

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = ( 1 - ( 7.61 )^2 / ( 2 * 128.25^2 ) ) * 239 / ( 5/3 + 3 * ( 7.61 ) / ( 8 * 128.25 ) - ( 7.61^3 ) / ( 8 * 128.25^3 ) )$$

$$Sca = 141.82 \text{ MPa}$$

**AISC Unity Check on Inside Ribs ( must be <= 1.0 )**

$$\text{Check} = Sc/Sca + (Rm/Z)/Sba$$

$$\text{Check} = 6.83 / 141.82 + ( 106837.11 / 470567.375 ) / 159.96$$

$$\text{Check} = 0.05$$

**Input Data for Base Plate Bolting Calculations:**

Total Number of Bolts per BasePlate	Nbolts	4
Total Number of Bolts in Tension/Baseplate	Nbt	4
Bolt Material Specification	C45E	N
Bolt Allowable Stress	Stba	113.34 MPa
Bolt Corrosion Allowance	Bca	0.0000 mm
Distance from Bolts to Edge	Edgedis	99.9998 mm
Nominal Bolt Diameter	Bnd	30.0000 mm
Thread Series	Series	TEMA Metric
BasePlate Allowable Stress	S	204.17 MPa
Area Available in a Single Bolt	BltArea	502.9650 mm <sup>2</sup>
Saddle Load QO (Weight)	QO	477964.3 N
Saddle Load QL (Wind/Seismic contribution)	QL	313.5 N
Maximum Transverse Force	Ft	2233.8 N
Maximum Longitudinal Force	F1	1371.7 N
Saddle Bolted to Steel Foundation		No

**Bolt Area Calculation per Dennis R. Moss**

Bolt Area Requirement Due to Longitudinal Load [Bltarearl]:

= 0.0 (QO > QL --> No Uplift in Longitudinal direction)

Bolt Area due to Shear Load [Bltarears]:

= F1 / (Stba \* Nbolts)  
 = 1371.65 / (113.34 \* 4.00 )  
 = 3.0259 mm<sup>2</sup>

Bolt Area due to Transverse Load

Moment on Baseplate Due to Transverse Load [Rmom]:

= B \* Ft + Sum of X Moments  
 = 2000.00 \* 2233.84 + 0.00  
 = 4469496.00 N-mm

Eccentricity (e):

= Rmom / QO  
 = 4469496 / 477964.31  
 = 9.35 mm < Bplen/6 --> No Uplift in Transverse direction

Bolt Area due to Transverse Load [Bltareart]:

= 0 (No Uplift)

Required of a Single Bolt [Bltarear]

= max[Bltarearl, Bltarears, Bltareart]  
 = max[0.0000 , 3.0259 , 0.0000 ]  
 = 3.0259 mm<sup>2</sup>

**ASME Horizontal Vessel Analysis: Stresses for the Right Saddle**

(per ASME Sec. VIII Div. 2 based on the Zick method.)

**Input and Calculated Values:**

Vessel Mean Radius	Rm	1507.00	mm
Stiffened Vessel Length per 4.15.6	L	10450.00	mm
Distance from Saddle to Vessel tangent	a	685.00	mm
Saddle Width	b	400.00	mm
Saddle Bearing Angle	theta	120.00	
Wear Plate Width	b1	500.00	mm
Wear Plate Bearing Angle	theta1	127.00	
Wear Plate Thickness	tr	14.0	mm

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Horizontal Vessel Analysis (Test) : Step: 11 7:08a Nov 7,2012

Wear Plate Allowable Stress	Sr	204.17	MPa
Inside Depth of Head	h2	750.00	mm
Shell Allowable Stress used in Calculation		204.17	MPa
Head Allowable Stress used in Calculation		204.17	MPa
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Test Case, no Ext. Forces		465130.22	N

**Horizontal Vessel Analysis Results: Actual Allowable**

Long. Stress at Top of Midspan	73.50	204.17	MPa
Long. Stress at Bottom of Midspan	90.02	204.17	MPa
Long. Stress at Top of Saddles	81.88	204.17	MPa
Long. Stress at Bottom of Saddles	81.64	204.17	MPa
Tangential Shear in Shell	19.40	163.34	MPa
Tangential Shear in Head	19.40	163.34	MPa
Circ. Stress at Horn of Saddle	20.20	255.22	MPa
Addl. Stress in Head as Stiffener	171.61	255.22	MPa
Circ. Compressive Stress in Shell	4.03	204.17	MPa

**Intermediate Results: Saddle Reaction Q due to Wind or Seismic**

Saddle Reaction Force due to Wind Ft [Fwt]:

$$\begin{aligned}
 &= F_{tr} * ( Ft/Num \text{ of Saddles} + Z \text{ Force Load} ) * B / E \\
 &= 3.00 * ( 4467.7 / 2 + 0 ) * 2000.0000 / 2610.2007 \\
 &= 5134.9 \text{ N}
 \end{aligned}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$\begin{aligned}
 &= \text{Max}( F_l, \text{Friction Load, Sum of X Forces} ) * B / L_s \\
 &= \text{Max}( 4156.52 , 0.00 , 0 ) * 2000.0000 / 8750.0010 \\
 &= 313.5 \text{ N}
 \end{aligned}$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$\begin{aligned}
 &= \text{Saddle Load} + \text{Max}( F_{wl}, F_{wt}, F_{sl}, F_{st} ) \\
 &= 459995 + \text{Max}( 313 , 5134 , 0 , 0 ) \\
 &= 465130.2 \text{ N}
 \end{aligned}$$

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Horizontal Vessel Analysis (Test) : Step: 11 7:08a Nov 7,2012

**Summary of Loads at the base of this Saddle:**

Vertical Load (including saddle weight)	473478.94	N
Transverse Shear Load Saddle	2233.84	N
Longitudinal Shear Load Saddle	1371.65	N

Hydrostatic Test Pressure at center of Vessel: 1.519 MPa

**Formulas and Substitutions for Horizontal Vessel Analysis:**

Note: Wear Plate is Welded to the Shell, k = 0.1

**The Computed K values from Table 4.15.1:**

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0132	K8 = 0.3405
K9 = 0.2711	K10 = 0.0581	K1* = 0.1923	K6p = 0.0472
K7P = 0.0118			

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is less than Rm/2.

**Moment per Equation 4.15.3 [M1]:**

$$\begin{aligned}
 &= -Q*a [1 - (1 - a/L + (R^2 - h^2)/(2a*L))/(1 + (4h^2)/3L)] \\
 &= -465130*685.00 [1 - (1 - 685.00/10450.00 + (1507.000^2 - 750.000^2)/(2*685.00*10450.00))/(1 + (4*750.00)/(3*10450.00))] \\
 &= -12189741.0 \text{ N-mm}
 \end{aligned}$$

**Moment per Equation 4.15.4 [M2]:**

$$\begin{aligned}
 &= Q*L/4(1 + 2(R^2 - h^2)/(L^2))/(1 + (4h^2)/(3L)) - 4a/L \\
 &= 465130*10450/4(1 + 2(1507^2 - 750^2)/(10450^2))/(1 + (4*750)/(3*10450)) - 4*684/10450 \\
 &= 825449280.0 \text{ N-mm}
 \end{aligned}$$

**Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:**

$$\begin{aligned}
 &= P * Rm/(2t) - M2/(pi*Rm^2*t) \\
 &= 1.52 * 1507.000 / (2*14.00) - .82545E+09 / (pi*1507.0^2*14.00) \\
 &= 73.50 \text{ MPa}
 \end{aligned}$$

**Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:**

$$\begin{aligned}
 &= P * Rm/(2t) + M2/(pi * Rm^2 * t) \\
 &= 1.52 * 1507.000 / (2 * 14.00) + .82545E+09 / (pi * 1507.0^2 * 14.00) \\
 &= 90.02 \text{ MPa}
 \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.8) [Sigma3]:

$$\begin{aligned}
 &= P * Rm / (2t) - M1 / (\pi * Rm^2 * t) \\
 &= 1.52 * 1507.000 / (2 * 14.00) - -0.1E+08 / (\pi * 1507.0^2 * 14.00) \\
 &= 81.88 \text{ MPa}
 \end{aligned}$$

Longitudinal Stress at bottom of Shell at Support (4.15.9) [Sigma4]:

$$\begin{aligned}
 &= P * Rm / (2t) + M1 / (\pi * Rm^2 * t) \\
 &= 1.52 * 1507.000 / (2 * 14.00) + -0.1E+08 / (\pi * 1507.0^2 * 14.00) \\
 &= 81.64 \text{ MPa}
 \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
 &= Q(L-2a) / (L + (4 * h^2 / 3)) \\
 &= 465130 ( 10450.00 - 2 * 685.00 ) / (10450.00 + ( 4 * 750.00 / 3)) \\
 &= 368854.3 \text{ N}
 \end{aligned}$$

Shear Stress in the shell no rings, stiffened (4.15.15) [tau3]:

$$\begin{aligned}
 &= K3 * Q / ( Rm * t ) \\
 &= 0.8799 * 465130 / ( 1507.0000 * 14.0000 ) \\
 &= 19.40 \text{ MPa}
 \end{aligned}$$

Shear Stress in the head, shell stiffened (4.15.16) [tau3\*]:

$$\begin{aligned}
 &= K3 * Q / ( Rm * th ) \\
 &= 0.8799 * 465130 / ( 1507.0000 * 14.0000 ) \\
 &= 19.40 \text{ MPa}
 \end{aligned}$$

Membrane stress in the head as a stiffener (4.15.18) [sigma5]:

$$\begin{aligned}
 &= K4 * Q / (Rm * th) + ((P * Ri) / (2 * th)) * (Ri / h^2) \\
 &= 0.4011 * 465130 / (1507.00 * 14.000) + \\
 &\quad ((1.52 * 1500.00) / (2 * 14.000)) * (1500.00 / 750.00) \\
 &= 171.61 \text{ MPa}
 \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned}
 &= 0.78 * \text{sqrt}( Rm * t ) \\
 &= 0.78 * \text{sqrt}( 1507.000 * 14.000 ) \\
 &= 113.296 \text{ mm}
 \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned}
 &= -K5 * Q * k / ( t * ( b + X1 + X2 ) ) \\
 &= -0.7603 * 465130 * 0.1 / ( 14.000 * ( 400.00 + 113.30 + 113.30 ) ) \\
 &= -4.03 \text{ MPa}
 \end{aligned}$$



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Effective reinforcing plate width (4.15.1) [B1]:

$$= \min( b + 1.56 * \text{sqrt}( Rm * t ), 2a )$$

$$= \min( 400.00 + 1.56 * \text{sqrt}( 1507.000 * 14.000 ), 2 * 685.000 )$$

$$= 626.59 \text{ mm}$$

Wear Plate/Shell Stress ratio (4.15.29) [eta]:

$$= \min( Sr/S, 1 )$$

$$= \min( 204.174 / 204.174 , 1 )$$

$$= 1.0000$$

Circumferential Stress at wear plate (4.15.26) [sigma6,r]:

$$= -K5 * Q * k / ( B1( t + eta * tr ) )$$

$$= -0.7603 * 465130 * 0.1 / ( 626.592 ( 14.000 + 1.000 * 14.000 ) )$$

$$= -2.02 \text{ MPa}$$

Circ. Comp. Stress at Horn of Saddle, L<8Rm (4.15.28) [sigma7,r\*]:

$$= -Q/(4(t+eta*tr)b1) - 12*K7*Q*Rm/(L(t+eta*tr)^2)$$

$$= -465130 / (4(14.000 + 1.000 * 14.000 )626.592 ) -$$

$$12*0.013*465130*1507.000/(10450.00(14.000+1.000*14.000)^2)$$

$$= -20.20 \text{ MPa}$$

**Results for Vessel Ribs, Web and Base**

Baseplate Length	Bplen	2400.0000	mm
Baseplate Thickness	Bpthk	20.0000	mm
Baseplate Width	Bpwid	400.0000	mm
Number of Ribs ( inc. outside ribs )	Nribs	4	
Rib Thickness	Ribtk	20.0000	mm
Web Thickness	Webtk	20.0000	mm
Web Location	Webloc	Center	

Moment of Inertia of Saddle - Lateral Direction

	Y	A	AY	Io
Shell	7.	10165.	71154.	664107.
Wearplate	21.	7000.	147000.	3201331.
Web	228.	8000.	1824000.	522538176.
BasePlate	438.	8000.	3504000.	1535017088.
Totals	694.	33165.	5546154.	2061420672.

FileName : P3902-BC -----

Horizontal Vessel Analysis (Test) : Step: 11 7:08a Nov 7,2012

Value C1 = Sumof(Ay)/Sumof(A) = 167. mm  
 Value I = Sumof(Io) - C1\*Sumof(Ay) = 1133940736. mm\*\*4  
 Value As = Sumof(A) - Ashell = 23000. mm<sup>2</sup>  
 $K1 = (1 + \cos(\beta) - .5 * \sin(\beta)^2) / (\pi - \beta + \sin(\beta) * \cos(\beta)) = 0.2035$   
 $Fh = K1 * Q = 0.2035 * 465130.219 = 94664.1016 \text{ N}$   
 Tension Stress, St = ( Fh/As ) = 4.1162 MPa  
 Allowed Stress, Sa = 0.6 \* Yield Str = 143.9676 MPa  
 $d = B - R * \sin(\theta) / \theta = 707.5100 \text{ mm}$   
 Bending Moment, M = Fh \* d = 67002952.0000 N-mm  
 Bending Stress, Sb = ( M \* C1 / I ) = 9.8782 MPa  
 Allowed Stress, Sa = 2/3 \* Yield Str = 159.9640 MPa

**Minimum Thickness of Baseplate per Moss :**

$= ( 3 * ( Q + Saddle\_Wt ) * BasePlateWidth / ( 4 * BasePlateLength * AllStress ) )^{1/2}$   
 $= ( 3 * ( 465130 + 8348 ) * 400.00 / ( 4 * 2400.000 * 159.964 ) )^{1/2}$   
 = 19.236 mm

Calculation of Axial Load, Intermediate Values and Compressive Stress

**Effective Baseplate Length [e]:**

$= ( Bplen - Clearance ) / ( Nr ribs - 1 )$   
 $= ( 2400.0000 - 25.4 ) / ( 4 - 1 ) = 791.5333 \text{ mm}$

**Baseplate Pressure Area [Ap]:**

$= e * Bpwid / 2$   
 $= 791.5333 * 400.0000 / 2 = 0.2E+06 \text{ mm}^2$

**Axial Load [P]:**

$= Ap * Bp$   
 $= 158306.7 * 0.48 = 76701.3 \text{ N}$

**Area of the Rib and Web [Ar]:**

$= ( Bpwid - Clearance - Webtk ) * Ribtk + e/2 * Webtk$   
 $= ( 400.000 - 25.4 - 20.000 ) * 20.000 + 791.5333 / 2 * 20.000$   
 = 15007.332 mm<sup>2</sup>

**Compressive Stress [Sc]:**

$= P/Ar = 76701.3 / 15007.3320 = 5.1114 \text{ MPa}$

Check of Outside Ribs:

Inertia of Saddle, Outer Ribs - Longitudinal Direction

	Y	A	AY	Ay <sup>2</sup>	Io
Rib	200.0	7346.0	1469200.0	0.0	96825736.0
Web	200.0	7915.3	1583066.5	0.0	527688.2
Values	200.0	15261.3	3052266.5	0.0	97353424.0

Bending Moment [Rm]:

$$= F1 / ( 2 * Bplen ) * e * r1 / 2$$

$$= 1371.7 / ( 2 * 2400.00 ) * 791.533 * 1191.00 / 2$$

$$= 134750.391 \text{ N-mm}$$

KL/R < Cc ( 14.7873 < 128.2550 ) per AISC E2-1

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = ( 1 - ( 14.79 )^2 / ( 2 * 128.25^2 ) ) * 239 / ( 5/3 + 3 * ( 14.79 ) / ( 8 * 128.25 ) - ( 14.79^3 ) / ( 8 * 128.25^3 ) )$$

$$Sca = 139.41 \text{ MPa}$$

**AISC Unity Check on Outside Ribs ( must be <= 1.0 )**

$$\text{Check} = Sc / Sca + (Rm / Z) / Sba$$

$$\text{Check} = 5.11 / 139.41 + (134750.39 / 486767.094) / 159.96$$

$$\text{Check} = 0.04$$

Check of Inside Ribs

Inertia of Saddle, Inner Ribs - Axial Direction

	Y	A	AY	Ay <sup>2</sup>	Io
Rib	187.3	7092.0	1328331.4	0.0	87609584.0
Web	187.3	15830.7	2965083.8	0.0	527688.2
Values	187.3	22922.7	4293415.0	0.0	88137272.0

KL/R < Cc ( 7.6142 < 128.2550 ) per AISC E2-1

$$Sca = (1 - (Klr)^2 / (2 * Cc^2)) * Fy / (5/3 + 3 * (Klr) / (8 * Cc) - (Klr^3) / (8 * Cc^3))$$

$$Sca = ( 1 - ( 7.61 )^2 / ( 2 * 128.25^2 ) ) * 239 / ( 5/3 + 3 * ( 7.61 ) / ( 8 * 128.25 ) - ( 7.61^3 ) / ( 8 * 128.25^3 ) )$$

$$Sca = 141.82 \text{ MPa}$$

FileName : P3902-BC -----

Horizontal Vessel Analysis (Test) : Step: 11 7:08a Nov 7,2012

**AISC Unity Check on Inside Ribs ( must be <= 1.0 )**

$$\text{Check} = Sc/Sca + (Rm/Z)/Sba$$

$$\text{Check} = 6.69 / 141.82 + ( 106837.11 /470567.375) / 159.96$$

$$\text{Check} = 0.05$$

**Input Data for Base Plate Bolting Calculations:**

Total Number of Bolts per BasePlate	Nbolts	4	
Total Number of Bolts in Tension/Baseplate	Nbt	4	
Bolt Material Specification		C45E	N
Bolt Allowable Stress	Stba	113.34	MPa
Bolt Corrosion Allowance	Bca	0.0000	mm
Distance from Bolts to Edge	Edgedis	99.9998	mm
Nominal Bolt Diameter	Bnd	30.0000	mm
Thread Series	Series	TEMA	Metric
BasePlate Allowable Stress	S	204.17	MPa
Area Available in a Single Bolt	BltArea	502.9650	mm <sup>2</sup>
Saddle Load QO (Weight)	QO	468344.1	N
Saddle Load QL (Wind/Seismic contribution)	QL	313.5	N
Maximum Transverse Force	Ft	2233.8	N
Maximum Longitudinal Force	F1	1371.7	N
Saddle Bolted to Steel Foundation		No	

**Bolt Area Calculation per Dennis R. Moss**

Bolt Area Requirement Due to Longitudinal Load [Bltarear1]:

$$= 0.0 \text{ (} QO > QL \text{ --> No Uplift in Longitudinal direction)}$$

Bolt Area due to Shear Load [Bltarears]:

$$= F1 / (Stba * Nbolts)$$

$$= 1371.65 / (113.34 * 4.00 )$$

$$= 3.0259 \text{ mm}^2$$

Bolt Area due to Transverse Load

Moment on Baseplate Due to Transverse Load [Rmom]:

$$= B * Ft + \text{Sum of X Moments}$$

$$= 2000.00 * 2233.84 + 0.00$$

$$= 4469496.00 \text{ N-mm}$$

FileName : P3902-BC -----

Horizontal Vessel Analysis (Test) : Step: 11 7:08a Nov 7,2012

Eccentricity (e):

$$= R_{mom} / QO$$

$$= 4469496 / 468344.06$$

$$= 9.54 \text{ mm} < B_{plen}/6 \text{ --> No Uplift in Transverse direction}$$

Bolt Area due to Transverse Load [Bltareart]:

$$= 0 \text{ (No Uplift)}$$

Required of a Single Bolt [Bltarear]

$$= \max[Bltarearl, Bltarears, Bltareart]$$

$$= \max[0.0000, 3.0259, 0.0000]$$

$$= 3.0259 \text{ mm}^2$$

FileName : P3902-BC

Nozzle Calcs. : GV1a

Nozl: 1 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: GV1a From : 10**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	0.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		0.00	deg
Diameter		508.0000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P355NH	
Flange Type		Slip on	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	250.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	800.0000	mm

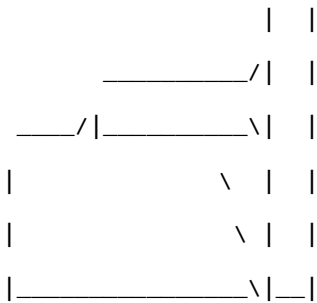
FileName : P3902-BC

Nozzle Calcs. : GV1a Nozl: 1 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		146.0000	mm
Class of attached Flange		10	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: GV1a**

Actual Outside Diameter Used in Calculation	508.000	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 494.000 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000$$

$$= 3.727 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Get the head depth from the head dimensions [h]:

Ar = Aspect ratio of the ellipse head

$$= D/2 * Ar + ci = 3000.000 / 2*2.000 + 3.000 = 753.0000 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : GV1a

Nozl: 1 7:08a Nov 7,2012

Compute the spherical radius per 9.5.1 [ris]:

$$= Di*(0.44 * Ar + 0.02) = 3050.800 * (0.44*2.000 +0.02) = 2705.3999 \text{ mm}$$

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2*ris+eas)*eas) = \text{Sqrt}(2*2705.400 + 11.000 ) * 11.000 )$$

$$= 244.2126 \text{ mm}$$

Credit Distance Along the Nozzle outside per 9.5.7.1 [Ibo]:

$$= \text{min}( \text{sqrt}( ( de - eab ) * eab ), ho )$$

$$= \text{min}( \text{sqrt}((508.0000 - 7.0000 ) * 7.0000 , 250.0000 )$$

$$= 59.2199 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [Ibi]:

$$= \text{Min}(hi, 0.50 * Ibo) = \text{Min}(0.000 , 0.50 * 59.220 ) = 0.0000 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= ris + eas / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}( \text{Offset} / Rm ) - \text{asin}( ( \text{Offset} - 0.5 * dib ) / Rm )$$

$$= \text{asin}(0.000 / 2710.900 ) - \text{asin}((0.000 - 0.5*508.000 ) / 2710.900 )$$

$$= 0.0938 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * rms = 0.094 * 2710.900 = 254.3731 \text{ mm}$$

Angle between nozzle centreline and head centreline [Beta]:

$$= \text{Asin}(\text{Offset} / rms) = \text{Asin}(0.000 ) / 2710.900 ) = 0.0000 \text{ radians}$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * ris^2 * ( Iso + a ) / ( 0.5 * eas + ris ) + a * ( eas + eap )$$

$$= 0.5*2705.400 **2*(244.213 +254.373 ) / (0.5*11.000 +2705.400 ) +$$

$$254.373 *(11.000 +14.000 ) = 679.4279 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \text{min}(ho, Ibo) * dib * 0.5 = \text{min}(250.000 , 59.220 ) * 494.000 * 0.5$$

$$= 14627.3213 \text{ mm}^2$$

$$\text{Afs} = Iso * eas = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$\text{Afb} = Ibo * eab + Ibi * \text{Max}(eab - cn + cnext, 0 )$$

$$= 59.220 * 7.000 + 0.000 * 0.000$$

$$= 414.5395 \text{ mm}^2$$

$$\text{Afp} = Ip * eap = 146.000 * 14.000 = 2044.0001 \text{ mm}^2$$



$$A_{fw} = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

$$f_{ob} = \text{Min}(f_s, f_b) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$f_{op} = \text{Min}(f_s, f_p) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$= (A_{fs} + A_{fw}) * (f_s - 0.5P) + A_{fp} * (\text{Min}(f_s, f_p) - 0.5P) + A_{fb} (\text{Min}(f_s, f_b) - 0.5P)$$

$$= (2686.339 + 196.000) * (204.174 - 0.5 * 0.600) +$$

$$(2044.000 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) +$$

$$(414.539 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600))$$

$$= 1088.7728 \times 10^3 \text{ N}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$= P * (A_{ps} + A_{pb})$$

$$= 0.600 * (679427 + 14627.321) = 416.3977 \times 10^3 \text{ N}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 11.9243 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 275.9242 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R1

Nozl: 2 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R1 From : 10**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	1100.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		0.00	deg
Diameter		219.1000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	280.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	490.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	310.0000	mm

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FileName : P3902-BC

Nozzle Calcs. : R1

Nozl: 2 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		45.4500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)

**Insert Nozzle With Pad, with Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R1**

Actual Outside Diameter Used in Calculation	219.100	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$\begin{aligned}
 &= P * D_i / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:} \\
 &= 0.600 * 205.100 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000 \\
 &= 3.302 \text{ mm}
 \end{aligned}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Note: This Is a Hillside Nozzle (Offset)

Get the head depth from the head dimensions [h]:

$$\begin{aligned}
 &Ar = \text{Aspect ratio of the ellipse head} \\
 &= D/2 * Ar + c_i = 3000.000 / 2 * 2.000 + 3.000 = 753.0000 \text{ mm}
 \end{aligned}$$

Compute the spherical radius per 9.5.1 [ris]:

FileName : P3902-BC

Nozzle Calcs. : R1

Nozl: 2 7:08a Nov 7,2012

$$= Di*(0.44 * Ar + 0.02) = 3050.800 * (0.44*2.000 +0.02) = 2705.3999 \text{ mm}$$

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2*ris+eas)*eas) = \text{Sqrt}(2*2705.400 + 11.000 ) * 11.000 )$$

$$= 244.2126 \text{ mm}$$

Credit Distance Along the Nozzle outside per 9.5.7.1 [Ibo]:

$$= \text{min}( \text{sqrt}( ( de - eab ) * eab ), ho )$$

$$= \text{min}( \text{sqrt}((219.1000 - 7.0000 ) * 7.0000 , 280.0000 )$$

$$= 38.5318 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [Ibi]:

$$= \text{Min}(hi, 0.50 * Ibo) = \text{Min}(490.000 , 0.50 * 38.532 ) = 19.2659 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= ris + eas / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}( \text{Offset} / Rm ) - \text{asin}( ( \text{Offset} - 0.5 * dib) / Rm )$$

$$= \text{asin}(1100.000 / 2710.900 ) - \text{asin}((1100.000 - 0.5 * 219.100 ) / 2710.900 )$$

$$= 0.0438 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * rms = 0.044 * 2710.900 = 118.7445 \text{ mm}$$

Angle between nozzle centerline and head centerline [Beta]:

$$= \text{Asin}(\text{Offset} / rms) = \text{Asin}(1100.000 ) / 2710.900 ) = 0.4178 \text{ radians}$$

Small triangular area fig 9.5-3 [Ap\_psi]:

$$= 0.5 * dib^2 * \text{tab}(\text{Beta}) = 0.5 * 205.100^2 * \text{Tan}(0.418 ) = 9337.8291 \text{ mm}^2$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * ris^2 * ( Iso + a ) / ( 0.5 * eas + ris ) + a * ( eas + eap )$$

$$= 0.5 * 2705.400 ** 2 * ( 244.213 + 118.745 ) / ( 0.5 * 11.000 + 2705.400 ) +$$

$$118.745 * ( 11.000 + 14.000 ) = 492.9446 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \text{min}(ho, Ibo) * dib * 0.5 = \text{min}(280.000 , 38.532 ) * 205.100 * 0.5$$

$$= 3951.4365 \text{ mm}^2$$

$$\text{Afs} = \text{Iso} * eas = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$\text{Afb} = \text{Ibo} * eab + \text{Ibi} * \text{Max}(eab - cn + cnext, 0 )$$

$$= 38.532 * 7.000 + 19.266 * 0.000$$

$$= 346.7862 \text{ mm}^2$$

FileName : P3902-BC

Nozzle Calcs. : R1

Noz1: 2 7:08a Nov 7,2012

$$A_{fp} = I_p * e_{ap} = 45.450 * 14.000 = 636.3000 \text{ mm}^2$$

$$A_{fw} = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

$$f_{ob} = \text{Min}(f_s, f_b) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$f_{op} = \text{Min}(f_s, f_p) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$= (A_{fs} + A_{fw}) * (f_s - 0.5P) + A_{fp} * (\text{Min}(f_s, f_p) - 0.5P) + A_{fb} (\text{Min}(f_s, f_b) - 0.5P)$$

$$= (2686.339 + 196.000) * (204.174 - 0.5 * 0.600) +$$

$$(636.300 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) +$$

$$(346.786 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600))$$

$$= 787.9920 \times 10^3 \text{ N}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$= P * (A_{ps} + A_{pb} + .5 * A_{psi})$$

$$= 0.600 * (492944.594 + 3951.437 + 0.5 * 0.000) = 300.9134 \times 10^3 \text{ N}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 51.5879 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 836.8975 mm

For possible nozzle interference - See Nozzle Summary

FileName : P3902-BC

Nozzle Calcs. : R3

Nozl: 3 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R3 From : 10**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	1200.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		180.00	deg
Diameter		114.3000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	250.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	100.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	310.0000	mm

FileName : P3902-BC

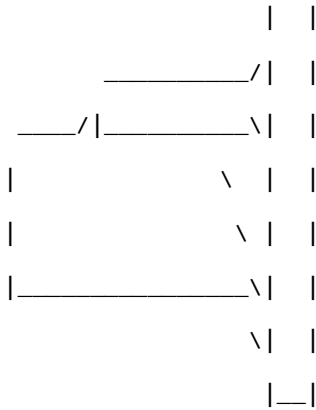
Nozzle Calcs. : R3

Nozl: 3 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		97.8500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



**Insert Nozzle With Pad, with Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R3**

Actual Outside Diameter Used in Calculation	114.300	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$\begin{aligned}
 &= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:} \\
 &= 0.600 * 100.300 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000 \\
 &= 3.148 \text{ mm}
 \end{aligned}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Note: This Is a Hillside Nozzle (Offset)

Get the head depth from the head dimensions [h]:

$$\begin{aligned}
 &Ar = \text{Aspect ratio of the ellipse head} \\
 &= D/2 * Ar + ci = 3000.000 / 2*2.000 + 3.000 = 753.0000 \text{ mm}
 \end{aligned}$$

Compute the spherical radius per 9.5.1 [ris]:

$$= Di*(0.44 * Ar + 0.02) = 3050.800 * (0.44*2.000 + 0.02) = 2705.3999 \text{ mm}$$

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2*\text{ris}+\text{eas})*\text{eas}) = \text{Sqrt}(2*2705.400 + 11.000 ) * 11.000 )$$

$$= 244.2126 \text{ mm}$$

Credit Distance Along the Nozzle outside per 9.5.7.1 [Ibo]:

$$= \text{min}( \text{sqrt}( ( \text{de} - \text{eab} ) * \text{eab} ), \text{ho} )$$

$$= \text{min}( \text{sqrt}((114.3000 - 7.0000 ) * 7.0000 , 250.0000 )$$

$$= 27.4062 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [Ibi]:

$$= \text{Min}(\text{hi}, 0.50 * \text{Ibo}) = \text{Min}(100.000 , 0.50 * 27.406 ) = 13.7031 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= \text{ris} + \text{eas} / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}( \text{Offset} / \text{Rm} ) - \text{asin}( ( \text{Offset} - 0.5 * \text{dib} ) / \text{Rm} )$$

$$= \text{asin}(1200.000 / 2710.900 ) - \text{asin}((1200.000 - 0.5 * 114.300 ) / 2710.900 )$$

$$= 0.0234 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * \text{rms} = 0.023 * 2710.900 = 63.3744 \text{ mm}$$

Angle between nozzle centerline and head centerline [Beta]:

$$= \text{Asin}(\text{Offset} / \text{rms}) = \text{Asin}(1200.000 / 2710.900 ) = 0.4586 \text{ radians}$$

Small triangular area fig 9.5-3 [Ap\_psi]:

$$= 0.5 * \text{dib}^2 * \text{tan}(\text{Beta}) = 0.5 * 100.300^2 * \text{Tan}(0.459 ) = 2483.1152 \text{ mm}^2$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * \text{ris}^2 * ( \text{Iso} + \text{a} ) / ( 0.5 * \text{eas} + \text{ris} ) + \text{a} * ( \text{eas} + \text{eap} )$$

$$= 0.5 * 2705.400 ** 2 * (244.213 + 63.374 ) / (0.5 * 11.000 + 2705.400 ) +$$

$$63.374 * (11.000 + 14.000 ) = 416.8131 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \text{min}(\text{ho}, \text{Ibo}) * \text{dib} * 0.5 = \text{min}(250.000 , 27.406 ) * 100.300 * 0.5$$

$$= 1374.4211 \text{ mm}^2$$

$$\text{Afs} = \text{Iso} * \text{eas} = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$\text{Afb} = \text{Ibo} * \text{eab} + \text{Ibi} * \text{Max}(\text{eab} - \text{cn} + \text{cnext}, 0 )$$

$$= 27.406 * 7.000 + 13.703 * 0.000$$

$$= 246.6558 \text{ mm}^2$$



$$A_{fp} = I_p * e_{ap} = 97.850 * 14.000 = 1369.9000 \text{ mm}^2$$

$$A_{fw} = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

$$f_{ob} = \text{Min}(f_s, f_b) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$f_{op} = \text{Min}(f_s, f_p) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= (A_{fs} + A_{fw}) * (f_s - 0.5P) + A_{fp} * (\text{Min}(f_s, f_p) - 0.5P) + A_{fb} (\text{Min}(f_s, f_b) - 0.5P) \\ &= (2686.339 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (1369.900 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (246.656 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 917.1290 \times 10^3 \text{ N} \end{aligned}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * (A_{ps} + A_{pb} + .5 * A_{psi}) \\ &= 0.600 * (416813.156 + 1374.421 + 0.5 * 0.000) = 251.6361 \times 10^3 \text{ N} \end{aligned}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 29.0538 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 394.6566 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : GV2a

Nozl: 4 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: GV2a From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		550.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		0.00	deg
Diameter		508.0000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P355NH	
Flange Type		Slip on	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	250.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	800.0000	mm
Thickness of Pad	te	14.0000	mm

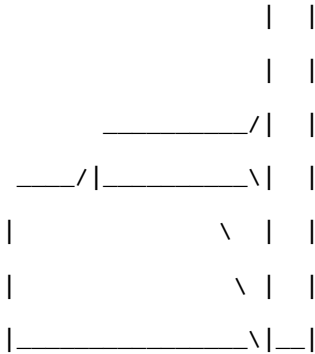
FileName : P3902-BC

Nozzle Calcs. : GV2a Nozl: 4 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	6.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	10.0000	mm
Reinforcing Pad Width		146.0000	mm
Class of attached Flange		10	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: GV2a**

Actual Outside Diameter Used in Calculation	508.000	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 494.000 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000$$

$$= 3.727 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( ( 508.000 - 2 * 0.000 - 7.000 ) * 7.000 ), 250.000 )$$

$$= 59.2199 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

FileName : P3902-BC -----

Nozzle Calcs. : GV2a Nozl: 4 7:08a Nov 7,2012

$$= \text{Min}(hi, 0.5 \cdot Ibo) = \text{Min}(0.000, 0.05 \cdot 59.220)$$

$$= 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 \cdot (Iso+a) \cdot (2 \cdot ris + (Iso+a) \cdot \tan(\text{Alpha}))$$

$$= 0.5 \cdot (182.173 + 247.000) \cdot (2 \cdot 1503.000 + (182.173 + 247.000) \cdot \tan(0.00))$$

$$= 645047 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= Ip \cdot ep = 146.000 \cdot 14.000$$

$$= 2044.0001 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 \cdot \tan(\text{Psi}) / 2 = 494.000^2 \cdot \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris \cdot (Iso + do / 2) \cdot ris + do/2 \cdot (eas + eap)$$

$$= ris \cdot (182.173 + 508.000 / 2) \cdot 1503.000 + 20.000 / 2 \cdot (11.000 + 14.000)$$

$$= 661.9180 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo \cdot dib / 2 = 59.220 \cdot 494.000 / 2$$

$$= 14.6273 \times 10^3 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(Dp - 2 \cdot cext - do/\text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(800.000 - 2.0 \cdot 0.000 - 508.000 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 146.0000 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) \cdot eas = (182.173 + 7.000) \cdot 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo \cdot eb + Ibi \cdot (eb - ci + co) = 59.220 \cdot 7.000 + 0.000 \cdot (7.000 - 3.000 + 0.000)$$

$$= 414.5395 \text{ mm}^2$$

$$Afp = Ip \cdot eap = 4.174 \cdot 14.000 = 2044.0001 \text{ mm}^2$$

$$Afw = (wshell^2 + wpad^2 + winside^2) / 2$$

$$= (14.0000^2 + 6.0000^2 + 0.0000^2) / 2 = 116.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb * (\text{Min}(fs, fb) - 0.5P)$$

$$= (2003.903 + 116.000) * (204.174 - 0.5 * 0.600) +$$

$$(2044.000 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) +$$

$$(414.539 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600))$$

$$= 933.3454 \times 10^3 \text{ N}$$

Force Term per 9.5.2.1.1 [Pa]:

$$= P * (Aps + Apb + 0.5 * APpsi)$$

$$= 0.600 * (661918 + 14627.321 + 0.5 * 0.000)$$

$$= 405.8927 \times 10^3 \text{ N}$$

Since  $F_a \geq P_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 21.6618 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 285.6618 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : GV2b

Nozl: 5 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: GV2b From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		7999.9995	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		0.00	deg
Diameter		508.0000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P355NH	
Flange Type		Slip on	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	250.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	800.0000	mm
Thickness of Pad	te	14.0000	mm

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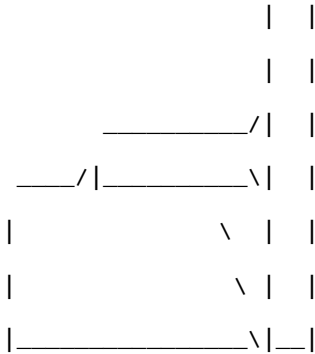
FileName : P3902-BC

Nozzle Calcs. : GV2b Nozl: 5 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	6.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	10.0000	mm
Reinforcing Pad Width		146.0000	mm
Class of attached Flange		10	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: GV2b**

Actual Outside Diameter Used in Calculation	508.000	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 494.000 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000$$

$$= 3.727 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( ( 508.000 - 2 * 0.000 - 7.000 ) * 7.000 ), 250.000 )$$

$$= 59.2199 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}( hi , 0.5 * lbo ) = \text{Min}( 0.000 , 0.5 * 59.220 )$$

$$= 0.0000 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : GV2b

Nozl: 5 7:08a Nov 7,2012

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$\begin{aligned}
 &= 0.5 * (Iso+a) * (2*ris+(Iso+a)*\tan(\text{Alpha})) \\
 &= 0.5 * (182.173 + 247.000) * (2*1503.000 + (182.173 + 247.000) * \tan(0.00)) \\
 &= 645047 \text{ mm}^2
 \end{aligned}$$

Compute area in the re-pad [Afp]:

$$\begin{aligned}
 &= I_p * e_p = 146.000 * 14.000 \\
 &= 2044.0001 \text{ mm}^2
 \end{aligned}$$

Compute triangular area [Ap.psi]:

$$\begin{aligned}
 &= dib^2 * \tan(\text{Psi}) / 2 = 494.000^2 * \tan(0.000) \\
 &= 0.0000 \text{ mm}^2
 \end{aligned}$$

Calculate the Pressure Area per 9.5-29 [Aps]:

**Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell**

$$\begin{aligned}
 &= ris * (Iso + do / 2) * ris + do/2 * (eas + eap) \\
 &= ris * (182.173 + 508.000 / 2) * 1503.000 + 20.000 / 2 * (11.000 + 14.000) \\
 &= 661.9180 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

$$\begin{aligned}
 &\text{(This is an inserted nozzle)} \\
 &= Ibo * dib / 2 = 59.220 * 494.000 / 2 \\
 &= 14.6273 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Compute the effective re-pad width [Ip]:

$$\begin{aligned}
 &= \text{Min}(\text{Max}(D_p - 2 * c_{ext} - do/\text{Cos}(\text{Phi}), 0) / 2, Iso) \\
 &= \text{Min}(\text{Max}(800.000 - 2.0 * 0.000 - 508.000 / \text{Cos}(0.000), 0.0) / 2.0, 182.173) \\
 &= 146.0000 \text{ mm}
 \end{aligned}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$\begin{aligned}
 Afs &= (Iso + eab) * eas = (182.173 + 7.000) * 11.000 = 2003.9032 \text{ mm}^2 \\
 Afb &= Ibo * eb + Ibi * (eb - ci + co) = 59.220 * 7.000 + 0.000 * (7.000 - 3.000 + 0.000) \\
 &= 414.5395 \text{ mm}^2 \\
 Afp &= I_p * eap = 4.174 * 14.000 = 2044.0001 \text{ mm}^2 \\
 Afw &= (wshell^2 + wpad^2 + winside^2) / 2 \\
 &= (14.0000^2 + 6.0000^2 + 0.0000^2) / 2 = 116.0000 \text{ mm}^2
 \end{aligned}$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$



$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb(\text{Min}(fs, fb) - 0.5P) \\ &= (2003.903 + 116.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (2044.000 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (414.539 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 933.3454 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * (Aps + Apb + 0.5 * APpsi) \\ &= 0.600 * (661918 + 14627.321 + 0.5 * 0.000) \\ &= 405.8927 \times 10^3 \text{ N} \end{aligned}$$

Since  $Fa \geq Pa$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 21.6618 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 285.6618 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : M1

Nozl: 6 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: M1 From : 20**

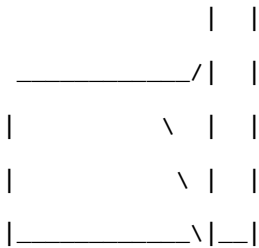
Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		3910.0002	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P285QH	
Allowable Stress at Temperature	Sn	162.51	MPa
Allowable Stress At Ambient	Sna	162.51	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		0.00	deg
Diameter		51.0000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	12.5000	mm
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	66.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



**Insert Nozzle No Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: M1**

Actual Outside Diameter Used in Calculation 51.000 mm.

Actual Thickness Used in Calculation 12.500 mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$\begin{aligned}
 &= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:} \\
 &= 0.600 * 32.000 / ( 2 * 162.506 * 0.000 - 0.600 ) + 3.000 \\
 &= -29.000 \text{ mm}
 \end{aligned}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$\begin{aligned}
 &= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2 \\
 &= 1503.0000 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$\begin{aligned}
 &= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho ) \\
 &= \min(\text{sqrt}((51.000 - 2*0.000 - 9.500 ) * 9.500 ), 66.000 ) \\
 &= 19.8557 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$\begin{aligned}
 &= \text{Min}(hi, 0.5*Ibo) = \text{Min}(0.000 , 05*19.856 ) \\
 &= 0.0000 \text{ mm}
 \end{aligned}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$\begin{aligned}
 &= 0.5*(Iso+a)*(2*ris+(Iso+a)*\tan(\text{Alpha})) \\
 &= 0.5*(182.173 +16.000 )*(2*1503.000 +(182.173 +16.000 )*\tan(0.00 )) \\
 &= 0.3E+06 \text{ mm}^2
 \end{aligned}$$

Compute area in the re-pad [Afp]:

$$= Ip * ep = 0.000 * 0.000 = 0.0000 \text{ mm}^2$$

FileName : P3902-BC

Nozzle Calcs. : M1

Nozl: 6 7:08a Nov 7,2012

Compute triangular area [Ap,psi]:

$$= \text{dib}^2 * \tan(\text{Psi}) / 2 = 32.000^2 * \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

**Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell**

$$= \text{ris} * (\text{Iso} + \text{do} / 2) * \text{ris} + \text{do}/2 * (\text{eas} + \text{eap})$$

$$= \text{ris} * (182.173 + 51.000 / 2) * 1503.000 + 2.008 / 2 * (11.000 + 0.000)$$

$$= 312.4130 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= \text{Ibo} * \text{dib} / 2 = 19.856 * 32.000 / 2$$

$$= 317.6917 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(\text{Dp} - 2 * \text{cext} - \text{do}/\text{Cos}(\text{Phi}), 0) / 2, \text{Iso})$$

$$= \text{Min}(\text{Max}(0.000 - 2.0 * 0.000 - 51.000 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 0.0000 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$\text{Afs} = (\text{Iso} + \text{eab}) * \text{eas} = (182.173 + 9.500) * 11.000 = 2003.9032 \text{ mm}^2$$

$$\text{Afb} = \text{Ibo} * \text{eb} + \text{Ibi} * (\text{eb} - \text{ci} + \text{co}) = 19.856 * 9.500 + 0.000 * (9.500 - 3.000 + 0.000)$$

$$= 188.6294 \text{ mm}^2$$

$$\text{Afp} = \text{Ip} * \text{eap} = 0.270 * 0.000 = 0.0000 \text{ mm}^2$$

$$\text{Afw} = (\text{wshell}^2 + \text{wpad}^2 + \text{winside}^2) / 2$$

$$= (14.0000^2 + 0.0000^2 + 0.0000^2) / 2 = 98.0000 \text{ mm}^2$$

**Note: There is No Reinforcement Pad**

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$\text{fob} = \text{Min}(\text{fs}, \text{fb}) = \text{Min}(204.174, 162.506) = 162.506 \text{ MPa}$$

$$\text{fop} = \text{Min}(\text{fs}, \text{fp}) = \text{Min}(204.174, 0.000) = 0.000 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$= (\text{Afs} + \text{Afw}) * (\text{fs} - 0.5P) + \text{Afp} * (\text{Min}(\text{fs}, \text{fp}) - 0.5P) + \text{Afb} * (\text{Min}(\text{fs}, \text{fb}) - 0.5P)$$

$$= (2003.903 + 98.000) * (204.174 - 0.5 * 0.600) +$$

$$(0.000 * (\text{Min}(204.174, 0.000) - 0.50 * 0.600) +$$

$$(188.629 * (\text{Min}(204.174, 162.506) - 0.50 * 0.600))$$

FileName : P3902-BC -----

Nozzle Calcs. : M1

Noz1: 6 7:08a Nov 7,2012

$$= 459.0808 \times 10^3 \text{ N}$$

Force Term per 9.5.2.1.1 [Pa]:

$$= P * ( A_{ps} + A_{pb} + 0.5 * A_{psi} )$$

$$= 0.600 * ( 312413.000 + 317.692 + 0.5*0.000 )$$

$$= 187.6225 \times 10^3 \text{ N}$$

Since  $F_a \geq P_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 0.2168 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 80.2168 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R8

Nozl: 7 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R8 From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		6990.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		0.00	deg
Diameter		168.3000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	145.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	310.0000	mm
Thickness of Pad	te	14.0000	mm

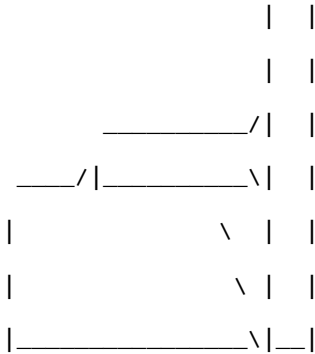
FileName : P3902-BC -----

Nozzle Calcs. : R8 Nozl: 7 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		70.8500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R8**

Actual Outside Diameter Used in Calculation	168.300	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 154.300 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -151.300 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( ( 168.300 - 2 * 0.000 - 7.000 ) * 7.000 ), 145.000 )$$

$$= 33.6021 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}( hi , 0.5 * lbo ) = \text{Min}( 0.000 , 05 * 33.602 ) = 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * (Iso+a) * (2*ris+(Iso+a)*tan(Alpha))$$

$$= 0.5 * (182.173 + 77.150) * (2*1503.000 + (182.173 + 77.150) * tan(0.00))$$

$$= 0.4E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= Ip * ep = 70.850 * 14.000$$

$$= 991.9000 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * tan(Psi) / 2 = 154.300^2 * tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do/2 * (eas + eap)$$

$$= ris * (182.173 + 168.300 / 2) * 1503.000 + 6.626 / 2 * (11.000 + 14.000)$$

$$= 402.3872 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo * dib / 2 = 33.602 * 154.300 / 2$$

$$= 2.5924 \times 10^3 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(Dp - 2 * cext - do/\text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(310.000 - 2.0 * 0.000 - 168.300 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 70.8500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) * eas = (182.173 + 7.000) * 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo * eb + Ibi * (eb - ci + co) = 33.602 * 7.000 + 0.000 * (7.000 - 3.000 + 0.000)$$

$$= 235.2146 \text{ mm}^2$$

$$Afp = Ip * eap = 1.304 * 14.000 = 991.9000 \text{ mm}^2$$

$$Afw = (wshell^2 + wpad^2 + winside^2) / 2$$

$$= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$



fop = Min(fs, fp) = Min(204.174 , 204.174 ) = 204.174 MPa

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned}
&= (Afs+Af_w)*(fs-0.5P)+Afp*(Min(fs,fp) - 0.5P) + Afb(Min(fs,fb) - 0.5P) \\
&= (2003.903 + 196.000 ) * (204.174 - 0.5 * 0.600 ) + \\
&\quad (991.900 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) + \\
&\quad (235.215 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) \\
&= 698.6199 \times 10^3 \text{ N}
\end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned}
&= P * ( Aps + Apb + 0.5 * APpsi ) \\
&= 0.600 * ( 402387.219 + 2592.401 + 0.5*0.000 ) \\
&= 242.9671 \times 10^3 \text{ N}
\end{aligned}$$

Since Fa >= Pa, Code Requirements are satisfied.

The Drop for this Nozzle is : 2.3623 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 161.3623 mm

For possible nozzle interference - See Nozzle Summary

FileName : P3902-BC

Nozzle Calcs. : R2

Nozl: 8 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R2 From: 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		9270.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		0.00	deg
Diameter		168.3000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	145.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	310.0000	mm
Thickness of Pad	te	14.0000	mm

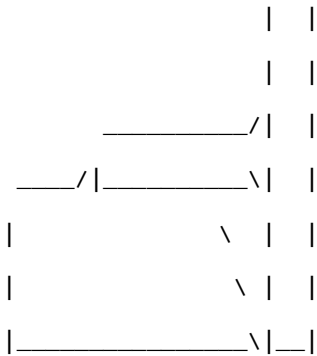
FileName : P3902-BC

Nozzle Calcs. : R2 Nozl: 8 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		70.8500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R2**

Actual Outside Diameter Used in Calculation	168.300	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 154.300 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -151.300 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( (168.300 - 2 * 0.000 - 7.000 ) * 7.000 ), 145.000 )$$

$$= 33.6021 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : R2

Nozl: 8 7:08a Nov 7,2012

Credit Distance Along the Nozzle per 9.5.77 [Ibi]:

$$= \text{Min}(hi, 0.5 \cdot Ibo) = \text{Min}(0.000, 0.5 \cdot 33.602)$$

$$= 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 \cdot (Iso+a) \cdot (2 \cdot ris + (Iso+a) \cdot \tan(\text{Alpha}))$$

$$= 0.5 \cdot (182.173 + 77.150) \cdot (2 \cdot 1503.000 + (182.173 + 77.150) \cdot \tan(0.00))$$

$$= 0.4E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= I_p \cdot e_p = 70.850 \cdot 14.000$$

$$= 991.9000 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 \cdot \tan(\text{Psi}) / 2 = 154.300^2 \cdot \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

**Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell**

$$= ris \cdot (Iso + do / 2) \cdot ris + do/2 \cdot (eas + eap)$$

$$= ris \cdot (182.173 + 168.300 / 2) \cdot 1503.000 + 6.626 / 2 \cdot (11.000 + 14.000)$$

$$= 402.3872 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo \cdot dib / 2 = 33.602 \cdot 154.300 / 2$$

$$= 2.5924 \times 10^3 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(D_p - 2 \cdot c_{ext} - do/\text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(310.000 - 2.0 \cdot 0.000 - 168.300 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 70.8500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) \cdot eas = (182.173 + 7.000) \cdot 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo \cdot eb + Ibi \cdot (eb - ci + co) = 33.602 \cdot 7.000 + 0.000 \cdot (7.000 - 3.000 + 0.000)$$

$$= 235.2146 \text{ mm}^2$$

$$Afp = I_p \cdot eap = 1.304 \cdot 14.000 = 991.9000 \text{ mm}^2$$

$$Afw = (wshell^2 + wpad^2 + winside^2) / 2 = (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb * (\text{Min}(fs, fb) - 0.5P)$$

$$= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) +$$

$$(991.900 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) +$$

$$(235.215 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600))$$

$$= 698.6199 \times 10^3 \text{ N}$$

Force Term per 9.5.2.1.1 [Pa]:

$$= P * (Aps + Apb + 0.5 * APpsi)$$

$$= 0.600 * (402387.219 + 2592.401 + 0.5 * 0.000)$$

$$= 242.9671 \times 10^3 \text{ N}$$

Since  $Fa \geq Pa$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 2.3623 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 161.3623 mm

For possible nozzle interference - See Nozzle Summary

FileName : P3902-BC

Nozzle Calcs. : R6

Nozl: 9 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R6 From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		10000.0010	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		0.00	deg
Diameter		219.1000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	145.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	400.0000	mm
Thickness of Pad	te	14.0000	mm

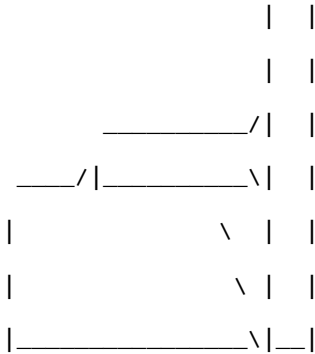
FileName : P3902-BC

Nozzle Calcs. : R6 Nozl: 9 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		90.4500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R6**

Actual Outside Diameter Used in Calculation	219.100	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 205.100 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -202.100 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( ( 219.100 - 2 * 0.000 - 7.000 ) * 7.000 ), 145.000 )$$

$$= 38.5318 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}( hi , 0.5 * lbo ) = \text{Min}( 0.000 , 0.5 * 38.532 )$$

$$= 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * (Iso+a) * (2*ris+(Iso+a)*tan(Alpha))$$

$$= 0.5 * (182.173 + 102.550) * (2*1503.000 + (182.173 + 102.550) * tan(0.00))$$

$$= 0.4E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= Ip * ep = 90.450 * 14.000$$

$$= 1266.3000 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * tan(Psi) / 2 = 205.100^2 * tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do/2 * (eas + eap)$$

$$= ris * (182.173 + 219.100 / 2) * 1503.000 + 8.626 / 2 * (11.000 + 14.000)$$

$$= 441.1984 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo * dib / 2 = 38.532 * 205.100 / 2$$

$$= 3.9514 \times 10^3 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(Dp - 2 * cext - do/\text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(400.000 - 2.0 * 0.000 - 219.100 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 90.4500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) * eas = (182.173 + 7.000) * 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo * eb + Ibi * (eb - ci + co) = 38.532 * 7.000 + 0.000 * (7.000 - 3.000 + 0.000)$$

$$= 269.7226 \text{ mm}^2$$

$$Afp = Ip * eap = 1.733 * 14.000 = 1266.3000 \text{ mm}^2$$

$$Afw = (wshell^2 + wpad^2 + winside^2) / 2$$

$$= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$



FileName : P3902-BC

Nozzle Calcs. : R6

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$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb * (\text{Min}(fs, fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (1266.300 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (269.723 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 761.5928 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * (Aps + Apb + 0.5 * APpsi) \\ &= 0.600 * (441198.406 + 3951.437 + 0.5 * 0.000) \\ &= 267.0672 \times 10^3 \text{ N} \end{aligned}$$

Since  $Fa \geq Pa$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 4.0058 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 163.0058 mm

For possible nozzle interference - See Nozzle Summary

FileName : P3902-BC

Nozzle Calcs. : R7c

Nozl: 10 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R7c From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		300.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		180.00	deg
Diameter		168.3000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	100.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	310.0000	mm
Thickness of Pad	te	14.0000	mm

FileName : P3902-BC

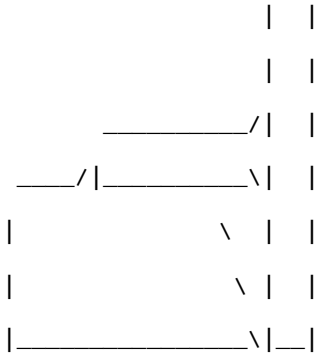
Nozzle Calcs. : R7c

Nozl: 10 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		70.8500	mm
Class of attached Flange		40	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R7c**

Actual Outside Diameter Used in Calculation	168.300	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 154.300 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -151.300 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min(\text{sqrt}((168.300 - 2*0.000 - 7.000 ) * 7.000 ), 100.000 )$$

$$= 33.6021 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}(hi, 0.5 * lbo) = \text{Min}(0.000, 0.5 * 33.602 )$$

$$= 0.0000 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : R7c

Nozl: 10 7:08a Nov 7,2012

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * (Iso+a) * (2*ris+(Iso+a)*tan(Alpha))$$

$$= 0.5 * (182.173 + 77.150) * (2*1503.000 + (182.173 + 77.150) * tan(0.00))$$

$$= 0.4E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= Ip * ep = 70.850 * 14.000$$

$$= 991.9000 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * tan(Psi) / 2 = 154.300^2 * tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do/2 * (eas + eap)$$

$$= ris * (182.173 + 168.300 / 2) * 1503.000 + 6.626 / 2 * (11.000 + 14.000)$$

$$= 402.3872 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo * dib / 2 = 33.602 * 154.300 / 2$$

$$= 2.5924 \times 10^3 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(Dp - 2 * cext - do/\text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(310.000 - 2.0 * 0.000 - 168.300 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 70.8500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) * eas = (182.173 + 7.000) * 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo * eb + Ibi * (eb - ci + co) = 33.602 * 7.000 + 0.000 * (7.000 - 3.000 + 0.000)$$

$$= 235.2146 \text{ mm}^2$$

$$Afp = Ip * eap = 1.304 * 14.000 = 991.9000 \text{ mm}^2$$

$$Afw = (wshell^2 + wpad^2 + winside^2) / 2$$

$$= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb(\text{Min}(fs, fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (991.900 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (235.215 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 698.6199 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * (Aps + Apb + 0.5 * APpsi) \\ &= 0.600 * (402387.219 + 2592.401 + 0.5 * 0.000) \\ &= 242.9671 \times 10^3 \text{ N} \end{aligned}$$

Since  $Fa \geq Pa$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 2.3623 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 116.3623 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R7b

Nozl: 11 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R7b From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		2750.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		180.00	deg
Diameter		168.3000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	100.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	310.0000	mm
Thickness of Pad	te	14.0000	mm

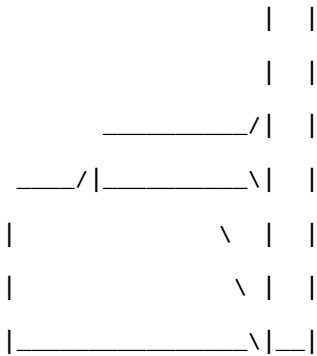
FileName : P3902-BC -----

Nozzle Calcs. : R7b Nozl: 11 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		70.8500	mm
Class of attached Flange		40	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R7b**

Actual Outside Diameter Used in Calculation	168.300	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 154.300 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -151.300 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min(\text{sqrt}((168.300 - 2*0.000 - 7.000 ) * 7.000 ), 100.000 ) = 33.6021 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : R7b

Nozl: 11 7:08a Nov 7,2012

Credit Distance Along the Nozzle per 9.5.77 [Ibi]:

$$= \text{Min}(hi, 0.5 \cdot Ibo) = \text{Min}(0.000, 0.5 \cdot 33.602)$$

$$= 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 \cdot (Iso+a) \cdot (2 \cdot ris + (Iso+a) \cdot \tan(\text{Alpha}))$$

$$= 0.5 \cdot (182.173 + 77.150) \cdot (2 \cdot 1503.000 + (182.173 + 77.150) \cdot \tan(0.00))$$

$$= 0.4E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= Ip \cdot ep = 70.850 \cdot 14.000$$

$$= 991.9000 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 \cdot \tan(\text{Psi}) / 2 = 154.300^2 \cdot \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris \cdot (Iso + do / 2) \cdot ris + do/2 \cdot (eas + eap)$$

$$= ris \cdot (182.173 + 168.300 / 2) \cdot 1503.000 + 6.626 / 2 \cdot (11.000 + 14.000)$$

$$= 402.3872 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo \cdot dib / 2 = 33.602 \cdot 154.300 / 2$$

$$= 2.5924 \times 10^3 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(Dp - 2 \cdot cext - do/\text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(310.000 - 2.0 \cdot 0.000 - 168.300 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 70.8500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) \cdot eas = (182.173 + 7.000) \cdot 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo \cdot eb + Ibi \cdot (eb - ci + co) = 33.602 \cdot 7.000 + 0.000 \cdot (7.000 - 3.000 + 0.000)$$

$$= 235.2146 \text{ mm}^2$$

$$Afp = Ip \cdot eap = 1.304 \cdot 14.000 = 991.9000 \text{ mm}^2$$

$$Afw = (wshell^2 + wpad^2 + winside^2) / 2 = (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$



Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb * (\text{Min}(fs, fb) - 0.5P)$$

$$= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) +$$

$$(991.900 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) +$$

$$(235.215 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600))$$

$$= 698.6199 \times 10^3 \text{ N}$$

Force Term per 9.5.2.1.1 [Pa]:

$$= P * (Aps + Apb + 0.5 * APpsi)$$

$$= 0.600 * (402387.219 + 2592.401 + 0.5 * 0.000)$$

$$= 242.9671 \times 10^3 \text{ N}$$

Since  $F_a \geq P_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 2.3623 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 116.3623 mm

For possible nozzle interference - See Nozzle Summary

FileName : P3902-BC

Nozzle Calcs. : R7a

Nozl: 12 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R7a From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		7700.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		180.00	deg
Diameter		168.3000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	100.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	310.0000	mm
Thickness of Pad	te	14.0000	mm

FileName : P3902-BC

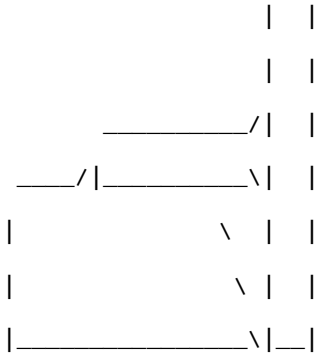
Nozzle Calcs. : R7a

Nozl: 12 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		70.8500	mm
Class of attached Flange		40	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R7a**

Actual Outside Diameter Used in Calculation	168.300	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 154.300 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -151.300 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min(\text{sqrt}((168.300 - 2*0.000 - 7.000 ) * 7.000 ), 100.000 )$$

$$= 33.6021 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}(hi, 0.5 * lbo) = \text{Min}(0.000, 0.5 * 33.602 )$$

$$= 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * (Iso + a) * (2 * ris + (Iso + a) * \tan(\text{Alpha}))$$

$$= 0.5 * (182.173 + 77.150) * (2 * 1503.000 + (182.173 + 77.150) * \tan(0.00))$$

$$= 0.4E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= I_p * e_p = 70.850 * 14.000$$

$$= 991.9000 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * \tan(\text{Psi}) / 2 = 154.300^2 * \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do / 2 * (eas + eap)$$

$$= ris * (182.173 + 168.300 / 2) * 1503.000 + 6.626 / 2 * (11.000 + 14.000)$$

$$= 402.3872 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo * dib / 2 = 33.602 * 154.300 / 2$$

$$= 2.5924 \times 10^3 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(D_p - 2 * c_{ext} - do / \cos(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(310.000 - 2.0 * 0.000 - 168.300 / \cos(0.000), 0.0) / 2.0, 182.173)$$

$$= 70.8500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) * eas = (182.173 + 7.000) * 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo * eb + Ibi * (eb - ci + co) = 33.602 * 7.000 + 0.000 * (7.000 - 3.000 + 0.000)$$

$$= 235.2146 \text{ mm}^2$$

$$Afp = I_p * eap = 1.304 * 14.000 = 991.9000 \text{ mm}^2$$

$$Afw = (w_{shell}^2 + w_{pad}^2 + w_{inside}^2) / 2$$

$$= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

fop = Min(fs, fp) = Min(204.174 , 204.174 ) = 204.174 MPa

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned}
&= (Afs+Afw)*(fs-0.5P)+Afp*(Min(fs,fp) - 0.5P) + Afb(Min(fs,fb) - 0.5P) \\
&= (2003.903 + 196.000 ) * (204.174 - 0.5 * 0.600 ) + \\
&\quad (991.900 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) + \\
&\quad (235.215 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) \\
&= 698.6199 \times 10^3 \text{ N}
\end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned}
&= P * ( Aps + Apb + 0.5 * APpsi ) \\
&= 0.600 * ( 402387.219 + 2592.401 + 0.5*0.000 ) \\
&= 242.9671 \times 10^3 \text{ N}
\end{aligned}$$

Since Fa >= Pa, Code Requirements are satisfied.

The Drop for this Nozzle is : 2.3623 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 116.3623 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R5

Nozl: 13 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R5 From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		8960.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		180.00	deg
Diameter		219.1000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	140.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	400.0000	mm
Thickness of Pad	te	14.0000	mm

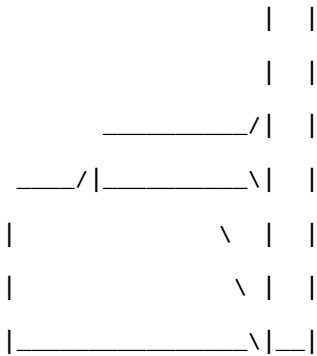
FileName : P3902-BC -----

Nozzle Calcs. : R5 Nozl: 13 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		90.4500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R5**

Actual Outside Diameter Used in Calculation	219.100	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$\begin{aligned}
 &= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:} \\
 &= 0.600 * 205.100 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000 \\
 &= -202.100 \text{ mm}
 \end{aligned}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$\begin{aligned}
 &= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2 \\
 &= 1503.0000 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$\begin{aligned}
 &= \min( \text{sqrt}( ( deb - 2 * cext - eab) * eab), ho ) \\
 &= \min(\text{sqrt}((219.100 - 2*0.000 - 7.000 ) * 7.000 ), 140.000 ) \\
 &= 38.5318 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}(hi, 0.5*Ibo) = \text{Min}(0.000, 05*38.532) = 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * (Iso+a) * (2*ris+(Iso+a)*tan(Alpha))$$

$$= 0.5 * (182.173 + 102.550) * (2*1503.000 + (182.173 + 102.550) * tan(0.00))$$

$$= 0.4E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= Ip * ep = 90.450 * 14.000$$

$$= 1266.3000 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * tan(Psi) / 2 = 205.100^2 * tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do/2 * (eas + eap)$$

$$= ris * (182.173 + 219.100 / 2) * 1503.000 + 8.626 / 2 * (11.000 + 14.000)$$

$$= 441.1984 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo * dib / 2 = 38.532 * 205.100 / 2$$

$$= 3.9514 \times 10^3 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(Dp - 2 * cext - do/\text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(400.000 - 2.0 * 0.000 - 219.100 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 90.4500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) * eas = (182.173 + 7.000) * 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo * eb + Ibi * (eb - ci + co) = 38.532 * 7.000 + 0.000 * (7.000 - 3.000 + 0.000)$$

$$= 269.7226 \text{ mm}^2$$

$$Afp = Ip * eap = 1.733 * 14.000 = 1266.3000 \text{ mm}^2$$

$$Afw = (wshell^2 + wpad^2 + winside^2) / 2$$

$$= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$



$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb(\text{Min}(fs, fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (1266.300 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (269.723 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 761.5928 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * (Aps + Apb + 0.5 * APpsi) \\ &= 0.600 * (441198.406 + 3951.437 + 0.5 * 0.000) \\ &= 267.0672 \times 10^3 \text{ N} \end{aligned}$$

Since  $Fa \geq Pa$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 4.0058 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 158.0058 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R4

Nozl: 14 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R4 From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		10150.0010	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		180.00	deg
Diameter		168.3000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	145.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	310.0000	mm
Thickness of Pad	te	14.0000	mm

FileName : P3902-BC

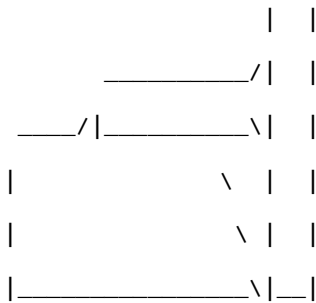
Nozzle Calcs. : R4

Nozl: 14 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		70.8500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R4**

Actual Outside Diameter Used in Calculation	168.300	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 154.300 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -151.300 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min(\text{sqrt}((168.300 - 2*0.000 - 7.000 ) * 7.000 ), 145.000 )$$

$$= 33.6021 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}(hi, 0.5 * lbo) = \text{Min}(0.000, 05 * 33.602 )$$

FileName : P3902-BC -----

Nozzle Calcs. : R4

Nozl: 14 7:08a Nov 7,2012

$$= 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * (Iso+a) * (2*ris+(Iso+a)*tan(Alpha))$$

$$= 0.5 * (182.173 + 77.150) * (2*1503.000 + (182.173 + 77.150) * tan(0.00))$$

$$= 0.4E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= Ip * ep = 70.850 * 14.000$$

$$= 991.9000 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * tan(Psi) / 2 = 154.300^2 * tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do/2 * (eas + eap)$$

$$= ris * (182.173 + 168.300 / 2) * 1503.000 + 6.626 / 2 * (11.000 + 14.000)$$

$$= 402.3872 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo * dib / 2 = 33.602 * 154.300 / 2$$

$$= 2.5924 \times 10^3 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(Dp - 2 * cext - do/\text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(310.000 - 2.0 * 0.000 - 168.300 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 70.8500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) * eas = (182.173 + 7.000) * 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo * eb + Ibi * (eb - ci + co) = 33.602 * 7.000 + 0.000 * (7.000 - 3.000 + 0.000)$$

$$= 235.2146 \text{ mm}^2$$

$$Afp = Ip * eap = 1.304 * 14.000 = 991.9000 \text{ mm}^2$$

$$Afw = (wshell^2 + wpad^2 + winside^2) / 2$$

$$= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb * (\text{Min}(fs, fb) - 0.5P)$$

$$= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) +$$

$$(991.900 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) +$$

$$(235.215 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600))$$

$$= 698.6199 \times 10^3 \text{ N}$$

Force Term per 9.5.2.1.1 [Pa]:

$$= P * (Aps + Apb + 0.5 * APpsi)$$

$$= 0.600 * (402387.219 + 2592.401 + 0.5 * 0.000)$$

$$= 242.9671 \times 10^3 \text{ N}$$

Since  $F_a \geq P_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 2.3623 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 161.3623 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R17b

Nozl: 15 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R17b From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		8700.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P285QH	
Allowable Stress at Temperature	Sn	162.51	MPa
Allowable Stress At Ambient	Sna	162.51	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		180.00	deg
Diameter		51.0000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	12.5000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	60.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Class of attached Flange		40	
Grade of attached Flange		GR 1.1	

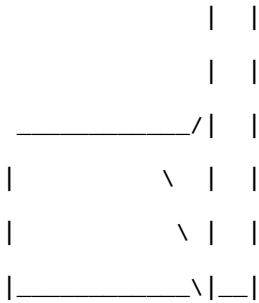
The Pressure Design option was Design Pressure + static head.

FileName : P3902-BC -----

Nozzle Calcs. : R17b

Nozl: 15 7:08a Nov 7,2012

Nozzle Sketch (may not represent actual weld type/configuration)



**Insert Nozzle No Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R17b**

Actual Outside Diameter Used in Calculation 51.000 mm.  
 Actual Thickness Used in Calculation 12.500 mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 32.000 / ( 2 * 162.506 * 0.000 - 0.600 ) + 3.000$$

$$= -29.000 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [Ibo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( ( 51.000 - 2 * 0.000 - 9.500 ) * 9.500 ), 60.000 )$$

$$= 19.8557 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [Ibi]:

$$= \text{Min}( hi , 0.5 * Ibo ) = \text{Min}( 0.000 , 0.5 * 19.856 )$$

$$= 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * ( Iso + a ) * ( 2 * ris + ( Iso + a ) * \tan( Alpha ) )$$

$$= 0.5 * ( 182.173 + 16.000 ) * ( 2 * 1503.000 + ( 182.173 + 16.000 ) * \tan( 0.00 ) )$$

$$= 0.3E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= I_p * e_p = 0.000 * 0.000$$

$$= 0.0000 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * \tan(\Psi) / 2 = 32.000^2 * \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do/2 * (eas + eap)$$

$$= ris * (182.173 + 51.000 / 2) * 1503.000 + 2.008 / 2 * (11.000 + 0.000)$$

$$= 312.4130 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo * dib / 2 = 19.856 * 32.000 / 2$$

$$= 317.6917 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(D_p - 2 * c_{ext} - do/\text{Cos}(\Phi), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(0.000 - 2.0 * 0.000 - 51.000 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 0.0000 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) * eas = (182.173 + 9.500) * 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo * eb + Ibi * (eb - ci + co) = 19.856 * 9.500 + 0.000 * (9.500 - 3.000 + 0.000)$$

$$= 188.6294 \text{ mm}^2$$

$$Afp = I_p * eap = 0.270 * 0.000 = 0.0000 \text{ mm}^2$$

$$Afw = (w_{shell}^2 + w_{pad}^2 + w_{inside}^2) / 2$$

$$= (14.0000^2 + 0.0000^2 + 0.0000^2) / 2 = 98.0000 \text{ mm}^2$$

Note: There is No Reinforcement Pad

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 162.506) = 162.506 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 0.000) = 0.000 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb * (\text{Min}(fs, fb) - 0.5P)$$



FileName : P3902-BC

Nozzle Calcs. : R17b

Nozl: 15 7:08a Nov 7,2012

$$\begin{aligned} &= (2003.903 + 98.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (0.000 * (\text{Min}(204.174, 0.000) - 0.50 * 0.600) + \\ &\quad (188.629 * (\text{Min}(204.174, 162.506) - 0.50 * 0.600) ) \\ &= 459.0808 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * ( A_{ps} + A_{pb} + 0.5 * A_{psi} ) \\ &= 0.600 * ( 312413.000 + 317.692 + 0.5*0.000 ) \\ &= 187.6225 \times 10^3 \text{ N} \end{aligned}$$

Since  $F_a \geq P_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 0.2168 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 74.2168 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R17d

Nozl: 16 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R17d From : 20**

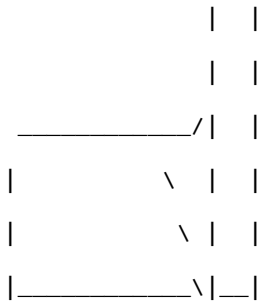
Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Cylinder/Cone Centerline	L1	500.0000	mm
Distance from Bottom/Left Tangent		10290.0010	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P285QH	
Allowable Stress at Temperature	Sn	162.51	MPa
Allowable Stress At Ambient	Sna	162.51	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		-161.00	deg
Diameter		51.0000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	12.5000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	38.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Class of attached Flange		40	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



**Insert Nozzle No Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R17d**

Actual Outside Diameter Used in Calculation 51.000 mm.

Actual Thickness Used in Calculation 12.500 mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$\begin{aligned}
 &= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:} \\
 &= 0.600 * 32.000 / ( 2 * 162.506 * 0.000 - 0.600 ) + 3.000 \\
 &= -29.000 \text{ mm}
 \end{aligned}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$\begin{aligned}
 &= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2 \\
 &= 1503.0000 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$\begin{aligned}
 &= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho ) \\
 &= \min( \text{sqrt}( ( 51.000 - 2 * 0.000 - 9.500 ) * 9.500 ), 38.000 ) \\
 &= 19.8557 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$\begin{aligned}
 &= \text{Min}( hi, 0.5 * lbo ) = \text{Min}( 0.000, 0.5 * 19.856 ) \\
 &= 0.0000 \text{ mm}
 \end{aligned}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$\begin{aligned}
 &= 0.5 * ( Iso + a ) * ( 2 * ris + ( Iso + a ) * \tan( Alpha ) ) \\
 &= 0.5 * ( 182.173 + 16.000 ) * ( 2 * 1503.000 + ( 182.173 + 16.000 ) * \tan( 0.00 ) ) \\
 &= 0.3E+06 \text{ mm}^2
 \end{aligned}$$

Compute area in the re-pad [Afp]:

$$= Ip * ep = 0.000 * 0.000 = 0.0000 \text{ mm}^2$$

FileName : P3902-BC -----

Nozzle Calcs. : R17d

Nozl: 16 7:08a Nov 7,2012

Compute triangular area [Ap.psi]:

$$= \text{dib}^2 * \tan(\text{Psi}) / 2 = 32.000^2 * \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= \text{ris} * (\text{Iso} + \text{do} / 2) * \text{ris} + \text{do}/2 * (\text{eas} + \text{eap})$$

$$= \text{ris} * (182.173 + 51.000 / 2) * 1503.000 + 2.008 / 2 * (11.000 + 0.000)$$

$$= 312.4130 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and lbo [Apb]:

(This is an inserted nozzle)

$$= \text{lbo} * \text{dib} / 2 = 19.856 * 32.000 / 2$$

$$= 317.6917 \text{ mm}^2$$

Compute the effective re-pad width [lp]:

$$= \text{Min}(\text{Max}(\text{Dp} - 2 * \text{cext} - \text{do}/\text{Cos}(\text{Phi}), 0) / 2, \text{Iso})$$

$$= \text{Min}(\text{Max}(0.000 - 2.0 * 0.000 - 51.000 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 0.0000 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$\text{Afs} = (\text{Iso} + \text{eab}) * \text{eas} = (182.173 + 9.500) * 11.000 = 2003.9032 \text{ mm}^2$$

$$\text{Afb} = \text{lbo} * \text{eb} + \text{lbi} * (\text{eb} - \text{ci} + \text{co}) = 19.856 * 9.500 + 0.000 * (9.500 - 3.000 + 0.000)$$

$$= 188.6294 \text{ mm}^2$$

$$\text{Afp} = \text{Ip} * \text{eap} = 0.270 * 0.000 = 0.0000 \text{ mm}^2$$

$$\text{Afw} = (\text{wshell}^2 + \text{wpad}^2 + \text{winside}^2) / 2$$

$$= (14.0000^2 + 0.0000^2 + 0.0000^2) / 2 = 98.0000 \text{ mm}^2$$

Note: There is No Reinforcement Pad

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$\text{fob} = \text{Min}(\text{fs}, \text{fb}) = \text{Min}(204.174, 162.506) = 162.506 \text{ MPa}$$

$$\text{fop} = \text{Min}(\text{fs}, \text{fp}) = \text{Min}(204.174, 0.000) = 0.000 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$= (\text{Afs} + \text{Afw}) * (\text{fs} - 0.5P) + \text{Afp} * (\text{Min}(\text{fs}, \text{fp}) - 0.5P) + \text{Afb} * (\text{Min}(\text{fs}, \text{fb}) - 0.5P)$$

$$= (2003.903 + 98.000) * (204.174 - 0.5 * 0.600) +$$

$$(0.000 * (\text{Min}(204.174, 0.000) - 0.50 * 0.600) +$$

FileName : P3902-BC -----

Nozzle Calcs. : R17d Nozl: 16 7:08a Nov 7,2012

$$(188.629 * (\text{Min}(204.174 , 162.506 ) - 0.50 * 0.600 )$$

$$= 459.0808 \times 10^3 \text{ N}$$

Force Term per 9.5.2.1.1 [Pa]:

$$= P * ( \text{Aps} + \text{Apb} + 0.5 * \text{APpsi} )$$

$$= 0.600 * ( 312413.000 + 317.692 + 0.5*0.000 )$$

$$= 187.6225 \times 10^3 \text{ N}$$

Since  $F_a \geq P_a$ , Code Requirements are satisfied.

[Checking the Nozzle Reinforcement at Right Angle to the Vessel Axis:](#)

Points where Nozzle Inside Diameter Cuts ris [x1,y1 : x2,y2]:

$$(484.0000 , 1422.9381 : 516.0000 , 1411.6490 )$$

Area [APsi]

$$= 178.4259 \text{ mm}^2$$

Angle Nozzle Outside Diameter Subtends Inside Radius of Head [Alpha]:

$$= \text{asin}((500.000 + 51.000 /2) / 1503.000 ) - \text{asin}((500.000 - 51.000 /2)$$

$$= 0.0360 \text{ Radians}$$

Angle Subtended By Iso [Beta]

$$= \text{Iso} / \text{ris} = 7.172 / 59.173 = 0.1212 \text{ Radians}$$

Area [Aps]:

$$= (\text{Alpha} + \text{Beta}) * \text{ris}^2 / 2 = (0.036 + 0.121 ) * 1503.000^2 / 2$$

$$= 0.2\text{E}+06 \text{ mm}^2$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$= (\text{Afs} + \text{Afw}) * (\text{fs} - 0.5P) + \text{Afp} * (\text{Min}(\text{fs}, \text{fp}) - 0.5P) + \text{Afb} * (\text{Min}(\text{fs}, \text{fb}) - 0.5P)$$

$$= (2003.903 + 98.000 ) * (204.174 - 0.5 * 0.600 ) +$$

$$(0.000 * (\text{Min}(204.174 , 0.000 ) - 0.50 * 0.600 ) +$$

$$(188.629 * (\text{Min}(204.174 , 162.506 ) - 0.50 * 0.600 )$$

$$= 459.0808 \times 10^3 \text{ N}$$

Force Term per 9.5.2.1.1 [Fa]:

$$= P * ( \text{Aps} + \text{Apb} + 0.5 * \text{APpsi} )$$

$$= 0.600 * ( 177547.281 + 317.692 + 0.5 * 178.426 )$$

$$= 106.7635 \times 10^3 \text{ N}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

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FileName : P3902-BC -----

Nozzle Calcs. : R17d

Noz1: 16 7:08a Nov 7,2012

The Drop for this Nozzle is : 9.2759 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 62.1166 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R11

Nozl: 17 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R11 From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		9150.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		90.00	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	150.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm
Thickness of Pad	te	14.0000	mm

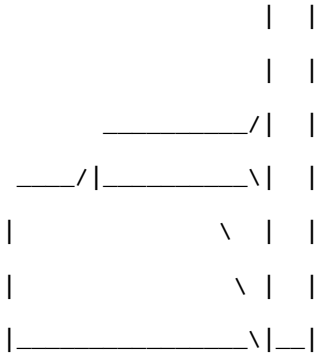
FileName : P3902-BC -----

Nozzle Calcs. : R11 Nozl: 17 7:08a Nov 7,2012

Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R11**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -75.900 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [Ibo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab) * eab), ho )$$

$$= \min(\text{sqrt}((88.900 - 2*0.000 - 5.000 ) * 5.000 ), 150.000 )$$

$$= 20.4817 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [Ibi]:

$$= \text{Min}(hi, 0.5*Ibo) = \text{Min}(0.000, 05*20.482 ) = 0.0000 \text{ mm}$$



Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * (Iso + a) * (2 * ris + (Iso + a) * \tan(\text{Alpha}))$$

$$= 0.5 * (182.173 + 39.450) * (2 * 1503.000 + (182.173 + 39.450) * \tan(0.00))$$

$$= 0.3E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= I_p * e_p = 55.550 * 14.000$$

$$= 777.7001 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * \tan(\text{Psi}) / 2 = 78.900^2 * \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do / 2 * (eas + eap)$$

$$= ris * (182.173 + 88.900 / 2) * 1503.000 + 3.500 / 2 * (11.000 + 14.000)$$

$$= 341.7256 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo * dib / 2 = 20.482 * 78.900 / 2$$

$$= 808.0030 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(D_p - 2 * c_{ext} - do / \text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(200.000 - 2.0 * 0.000 - 88.900 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 55.5500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) * eas = (182.173 + 5.000) * 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo * eb + Ibi * (eb - ci + co) = 20.482 * 5.000 + 0.000 * (5.000 - 3.000 + 0.000)$$

$$= 102.4085 \text{ mm}^2$$

$$Afp = I_p * eap = 0.667 * 14.000 = 777.7001 \text{ mm}^2$$

$$Afw = (w_{shell}^2 + w_{pad}^2 + w_{inside}^2) / 2$$

$$= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

fop = Min(fs, fp) = Min(204.174 , 204.174 ) = 204.174 MPa

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned}
&= (Afs+Af_w)*(fs-0.5P)+Afp*(Min(fs,fp) - 0.5P) + Afb(Min(fs,fb) - 0.5P) \\
&= (2003.903 + 196.000 ) * (204.174 - 0.5 * 0.600 ) + \\
&\quad (777.700 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) + \\
&\quad (102.408 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) \\
&= 627.8804 \times 10^3 \text{ N}
\end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned}
&= P * ( Aps + Apb + 0.5 * APpsi ) \\
&= 0.600 * ( 341725.625 + 808.003 + 0.5*0.000 ) \\
&= 205.5027 \times 10^3 \text{ N}
\end{aligned}$$

Since Fa >= Pa, Code Requirements are satisfied.

The Drop for this Nozzle is : 0.6588 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 164.6588 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R9

Nozl: 18 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R9 From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Cylinder/Cone Centerline	L1	150.0000	mm
Distance from Bottom/Left Tangent		8850.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		95.69	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	138.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm

FileName : P3902-BC

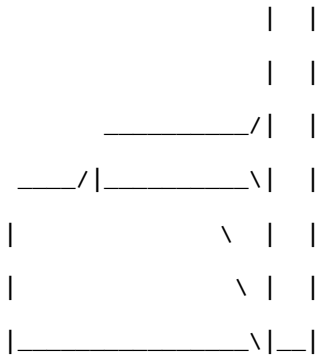
Nozzle Calcs. : R9

Nozl: 18 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R9**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -75.900 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( ( 88.900 - 2 * 0.000 - 5.000 ) * 5.000 ), 138.000 )$$

$$= 20.4817 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}( hi , 0.5 * lbo ) = \text{Min}( 0.000 , 0.5 * 20.482 ) = 0.0000 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : R9

Nozl: 18 7:08a Nov 7,2012

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$\begin{aligned}
 &= 0.5 * (Iso+a) * (2*ris+(Iso+a)*\tan(\text{Alpha})) \\
 &= 0.5 * (182.173 + 39.450) * (2*1503.000 + (182.173 + 39.450) * \tan(0.00)) \\
 &= 0.3E+06 \text{ mm}^2
 \end{aligned}$$

Compute area in the re-pad [Afp]:

$$\begin{aligned}
 &= I_p * e_p = 55.550 * 14.000 \\
 &= 777.7001 \text{ mm}^2
 \end{aligned}$$

Compute triangular area [Ap.psi]:

$$\begin{aligned}
 &= dib^2 * \tan(\text{Psi}) / 2 = 78.900^2 * \tan(0.000) \\
 &= 0.0000 \text{ mm}^2
 \end{aligned}$$

Calculate the Pressure Area per 9.5-29 [Aps]:

**Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell**

$$\begin{aligned}
 &= ris * (Iso + do / 2) * ris + do/2 * (eas + eap) \\
 &= ris * (182.173 + 88.900 / 2) * 1503.000 + 3.500 / 2 * (11.000 + 14.000) \\
 &= 341.7256 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

$$\begin{aligned}
 &\text{(This is an inserted nozzle)} \\
 &= I_{bo} * dib / 2 = 20.482 * 78.900 / 2 \\
 &= 808.0030 \text{ mm}^2
 \end{aligned}$$

Compute the effective re-pad width [Ip]:

$$\begin{aligned}
 &= \text{Min}(\text{Max}(D_p - 2 * c_{ext} - do/\text{Cos}(\text{Phi}), 0) / 2, Iso) \\
 &= \text{Min}(\text{Max}(200.000 - 2.0 * 0.000 - 88.900 / \text{Cos}(0.000), 0.0) / 2.0, 182.173) \\
 &= 55.5500 \text{ mm}
 \end{aligned}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$\begin{aligned}
 Afs &= (Iso + eab) * eas = (182.173 + 5.000) * 11.000 = 2003.9032 \text{ mm}^2 \\
 Afb &= I_{bo} * e_b + I_{bi} * (e_b - c_i + c_o) = 20.482 * 5.000 + 0.000 * (5.000 - 3.000 + 0.000) \\
 &= 102.4085 \text{ mm}^2 \\
 Afp &= I_p * e_{ap} = 0.667 * 14.000 = 777.7001 \text{ mm}^2 \\
 Afw &= (w_{shell}^2 + w_{pad}^2 + w_{inside}^2) / 2 \\
 &= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2
 \end{aligned}$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

FileName : P3902-BC -----

Nozzle Calcs. : R9

Nozl: 18 7:08a Nov 7,2012

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= (Afs+Afw)*(fs-0.5P)+Afp*(\text{Min}(fs,fp) - 0.5P) + Afb(\text{Min}(fs,fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 627.8804 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * ( Aps + Apb + 0.5 * APpsi ) \\ &= 0.600 * ( 341725.625 + 808.003 + 0.5*0.000 ) \\ &= 205.5027 \times 10^3 \text{ N} \end{aligned}$$

Since  $Fa \geq Pa$ , Code Requirements are satisfied.

Checking the Nozzle Reinforcement at Right Angle to the Vessel Axis:

Points where Nozzle Inside Diameter Cuts ris [x1,y1 : x2,y2]:

$$(110.5500, 1498.9288 : 189.4500, 1491.0122)$$

Area [APsi]

$$= 284.6544 \text{ mm}^2$$

Angle Nozzle Outside Diameter Subtends Inside Radius of Head [Alpha]:

$$\begin{aligned} &= \text{asin}((150.000 + 88.900 / 2) / 1503.000) - \text{asin}((150.000 - 88.900 / 2) \\ &= 0.0595 \text{ Radians} \end{aligned}$$

Angle Subtended By Iso [Beta]

$$= \text{Iso} / \text{ris} = 7.172 / 59.173 = 0.1212 \text{ Radians}$$

Area [Aps]:

$$\begin{aligned} &= (\text{Alpha} + \text{Beta}) * \text{ris}^2 / 2 = (0.059 + 0.121) * 1503.000^2 / 2 \\ &= 0.2E+06 \text{ mm}^2 \end{aligned}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= (Afs+Afw)*(fs-0.5P)+Afp*(\text{Min}(fs,fp) - 0.5P) + Afb(\text{Min}(fs,fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 627.8804 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Fa]:

$$= P * ( Aps + Apb + 0.5 * APpsi )$$

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FileName : P3902-BC -----

Nozzle Calcs. : R9

Noz1: 18 7:08a Nov 7,2012

$$= 0.600 * ( 204056.750 + 808.003 + 0.5 * 284.654 )$$

$$= 122.9938 \times 10^3 \text{ N}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 5.1382 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 157.2080 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R18a

Nozl: 19 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R18a From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Cylinder/Cone Centerline	L1	600.0000	mm
Distance from Bottom/Left Tangent		9000.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		113.35	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	135.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa



FileName : P3902-BC

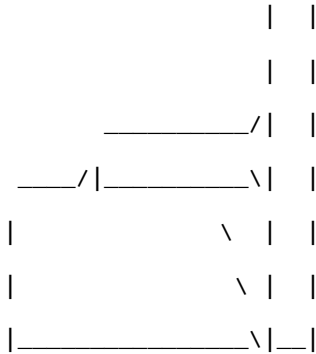
Nozzle Calcs. : R18a

Nozl: 19 7:08a Nov 7,2012

Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm
Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R18a**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -75.900 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( ( 88.900 - 2 * 0.000 - 5.000 ) * 5.000 ), 135.000 ) = 20.4817 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : R18a

Nozl: 19 7:08a Nov 7,2012

Credit Distance Along the Nozzle per 9.5.77 [Ibi]:

$$= \text{Min}(hi, 0.5 \cdot Ibo) = \text{Min}(0.000, 0.5 \cdot 20.482)$$

$$= 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 \cdot (Iso+a) \cdot (2 \cdot ris + (Iso+a) \cdot \tan(\text{Alpha}))$$

$$= 0.5 \cdot (182.173 + 39.450) \cdot (2 \cdot 1503.000 + (182.173 + 39.450) \cdot \tan(0.000))$$

$$= 0.3E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= I_p \cdot e_p = 55.550 \cdot 14.000$$

$$= 777.7001 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 \cdot \tan(\text{Psi}) / 2 = 78.900^2 \cdot \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

**Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell**

$$= ris \cdot (Iso + do / 2) \cdot ris + do/2 \cdot (eas + eap)$$

$$= ris \cdot (182.173 + 88.900 / 2) \cdot 1503.000 + 3.500 / 2 \cdot (11.000 + 14.000)$$

$$= 341.7256 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo \cdot dib / 2 = 20.482 \cdot 78.900 / 2$$

$$= 808.0030 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(D_p - 2 \cdot c_{ext} - do / \cos(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(200.000 - 2.0 \cdot 0.000 - 88.900 / \cos(0.000), 0.0) / 2.0, 182.173)$$

$$= 55.5500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) \cdot eas = (182.173 + 5.000) \cdot 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo \cdot eb + Ibi \cdot (eb - ci + co) = 20.482 \cdot 5.000 + 0.000 \cdot (5.000 - 3.000 + 0.000)$$

$$= 102.4085 \text{ mm}^2$$

$$Afp = I_p \cdot eap = 0.667 \cdot 14.000 = 777.7001 \text{ mm}^2$$

FileName : P3902-BC

Nozzle Calcs. : R18a

Nozl: 19 7:08a Nov 7,2012

$$\begin{aligned} A_{fw} &= (w_{shell}^2 + w_{pad}^2 + w_{inside}^2) / 2 \\ &= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2 \end{aligned}$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(f_s, f_b) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$fop = \text{Min}(f_s, f_p) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= (A_{fs} + A_{fw}) * (f_s - 0.5P) + A_{fp} * (\text{Min}(f_s, f_p) - 0.5P) + A_{fb} (\text{Min}(f_s, f_b) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600)) + \\ &\quad (102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600)) \\ &= 627.8804 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * (A_{ps} + A_{pb} + 0.5 * A_{psi}) \\ &= 0.600 * (341725.625 + 808.003 + 0.5 * 0.000) \\ &= 205.5027 \times 10^3 \text{ N} \end{aligned}$$

Since  $F_a \geq P_a$ , Code Requirements are satisfied.Checking the Nozzle Reinforcement at Right Angle to the Vessel Axis:

Points where Nozzle Inside Diameter Cuts ris [x1,y1 : x2,y2]:

$$(560.5500, 1394.5582 : 639.4500, 1360.1885)$$

Area [APsi] = 1320.5806 mm<sup>2</sup>

Angle Nozzle Outside Diameter Subtends Inside Radius of Head [Alpha]:

$$\begin{aligned} &= \text{asin}((600.000 + 88.900 / 2) / 1503.000) - \text{asin}((600.000 - 88.900 / 2) / 1503.000) \\ &= 0.0645 \text{ Radians} \end{aligned}$$

Angle Subtended By Iso [Beta]

$$= \text{Iso} / \text{ris} = 7.172 / 59.173 = 0.1212 \text{ Radians}$$

Area [Aps]:

$$\begin{aligned} &= (\text{Alpha} + \text{Beta}) * \text{ris}^2 / 2 = (0.065 + 0.121) * 1503.000^2 / 2 \\ &= 0.2E+06 \text{ mm}^2 \end{aligned}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= (A_{fs} + A_{fw}) * (f_s - 0.5P) + A_{fp} * (\text{Min}(f_s, f_p) - 0.5P) + A_{fb} (\text{Min}(f_s, f_b) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600)) + \\ &\quad (102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600)) = 627.8804 \times 10^3 \text{ N} \end{aligned}$$

FileName : P3902-BC

Nozzle Calcs. : R18a

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Force Term per 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= P * ( Aps + Apb + 0.5 * APpsi ) \\ &= 0.600 * ( 209789.047 + 808.003 + 0.5 * 1320.581 ) \\ &= 126.7436 \times 10^3 \text{ N} \end{aligned}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 20.2675 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 170.5293 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R10

Nozl: 20 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R10 From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Cylinder/Cone Centerline	L1	1200.0000	mm
Distance from Bottom/Left Tangent		9150.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		142.43	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	157.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm

FileName : P3902-BC

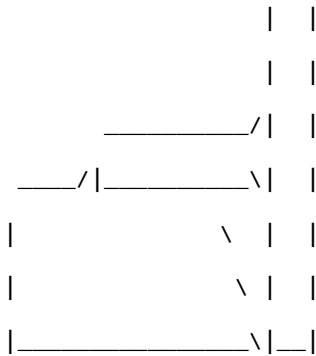
Nozzle Calcs. : R10

Nozl: 20 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R10**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -75.900 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( ( 88.900 - 2 * 0.000 - 5.000 ) * 5.000 ), 157.000 )$$

$$= 20.4817 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}( hi , 0.5 * lbo ) = \text{Min}( 0.000 , 0.5 * 20.482 ) = 0.0000 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : R10

Nozl: 20 7:08a Nov 7,2012

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$\begin{aligned}
 &= 0.5 * (Iso+a) * (2*ris+(Iso+a)*\tan(\text{Alpha})) \\
 &= 0.5 * (182.173 + 39.450) * (2*1503.000 + (182.173 + 39.450) * \tan(0.00)) \\
 &= 0.3E+06 \text{ mm}^2
 \end{aligned}$$

Compute area in the re-pad [Afp]:

$$\begin{aligned}
 &= I_p * e_p = 55.550 * 14.000 \\
 &= 777.7001 \text{ mm}^2
 \end{aligned}$$

Compute triangular area [Ap.psi]:

$$\begin{aligned}
 &= dib^2 * \tan(\text{Psi}) / 2 = 78.900^2 * \tan(0.000) \\
 &= 0.0000 \text{ mm}^2
 \end{aligned}$$

Calculate the Pressure Area per 9.5-29 [Aps]:

**Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell**

$$\begin{aligned}
 &= ris * (Iso + do / 2) * ris + do/2 * (eas + eap) \\
 &= ris * (182.173 + 88.900 / 2) * 1503.000 + 3.500 / 2 * (11.000 + 14.000) \\
 &= 341.7256 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

$$\begin{aligned}
 &\text{(This is an inserted nozzle)} \\
 &= I_{bo} * dib / 2 = 20.482 * 78.900 / 2 \\
 &= 808.0030 \text{ mm}^2
 \end{aligned}$$

Compute the effective re-pad width [Ip]:

$$\begin{aligned}
 &= \text{Min}(\text{Max}(D_p - 2 * c_{ext} - do/\text{Cos}(\text{Phi}), 0) / 2, Iso) \\
 &= \text{Min}(\text{Max}(200.000 - 2.0 * 0.000 - 88.900 / \text{Cos}(0.000), 0.0) / 2.0, 182.173) \\
 &= 55.5500 \text{ mm}
 \end{aligned}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$\begin{aligned}
 Afs &= (Iso + eab) * eas = (182.173 + 5.000) * 11.000 = 2003.9032 \text{ mm}^2 \\
 Afb &= I_{bo} * e_b + I_{bi} * (e_b - c_i + c_o) = 20.482 * 5.000 + 0.000 * (5.000 - 3.000 + 0.000) \\
 &= 102.4085 \text{ mm}^2 \\
 Afp &= I_p * e_{ap} = 0.667 * 14.000 = 777.7001 \text{ mm}^2 \\
 Afw &= (w_{shell}^2 + w_{pad}^2 + w_{inside}^2) / 2 \\
 &= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2
 \end{aligned}$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

FileName : P3902-BC -----

Nozzle Calcs. : R10

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$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= (Afs+Afw)*(fs-0.5P)+Afp*(\text{Min}(fs,fp) - 0.5P) + Afb(\text{Min}(fs,fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 627.8804 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * ( Aps + Apb + 0.5 * APpsi ) \\ &= 0.600 * ( 341725.625 + 808.003 + 0.5*0.000 ) \\ &= 205.5027 \times 10^3 \text{ N} \end{aligned}$$

Since  $Fa \geq Pa$ , Code Requirements are satisfied.

Checking the Nozzle Reinforcement at Right Angle to the Vessel Axis:

Points where Nozzle Inside Diameter Cuts ris [x1,y1 : x2,y2]:

$$(1160.5499, 955.0564 : 1239.4500, 850.1604)$$

$$\text{Area [APsi]} = 4012.7544 \text{ mm}^2$$

Angle Nozzle Outside Diameter Subtends Inside Radius of Head [Alpha]:

$$\begin{aligned} &= \text{asin}((1200.000 + 88.900 / 2) / 1503.000) - \text{asin}((1200.000 - 88.900 / 2) / 1503.000) \\ &= 0.0985 \text{ Radians} \end{aligned}$$

Angle Subtended By Iso [Beta]

$$= \text{Iso} / \text{ris} = 7.172 / 59.173 = 0.1212 \text{ Radians}$$

Area [Aps]:

$$\begin{aligned} &= (\text{Alpha} + \text{Beta}) * \text{ris}^2 / 2 = (0.098 + 0.121) * 1503.000^2 / 2 \\ &= 0.2E+06 \text{ mm}^2 \end{aligned}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= (Afs+Afw)*(fs-0.5P)+Afp*(\text{Min}(fs,fp) - 0.5P) + Afb(\text{Min}(fs,fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 627.8804 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= P * ( Aps + Apb + 0.5 * APpsi ) \\ &= 0.600 * ( 248140.219 + 808.003 + 0.5 * 4012.754 ) = 150.5600 \times 10^3 \text{ N} \end{aligned}$$



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FileName : P3902-BC

Nozzle Calcs. : R10

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Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 62.5371 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 242.6817 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R12

Nozl: 21 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R12 From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Cylinder/Cone Centerline	L1	1000.0000	mm
Distance from Bottom/Left Tangent		8850.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		131.34	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	109.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm

FileName : P3902-BC

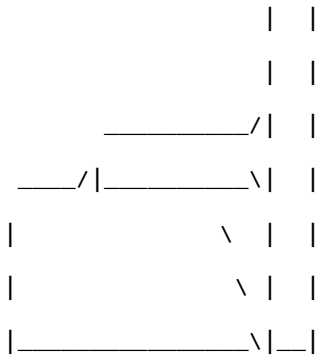
Nozzle Calcs. : R12

Nozl: 21 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R12**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 204.174 * 0.000 - 0.600 ) + 3.000$$

$$= -75.900 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2$$

$$= 1503.0000 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho )$$

$$= \min( \text{sqrt}( ( 88.900 - 2 * 0.000 - 5.000 ) * 5.000 ), 109.000 )$$

$$= 20.4817 \text{ mm}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$= \text{Min}( hi , 0.5 * lbo ) = \text{Min}( 0.000 , 0.5 * 20.482 ) = 0.0000 \text{ mm}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * (Iso + a) * (2 * ris + (Iso + a) * \tan(\text{Alpha}))$$

$$= 0.5 * (182.173 + 39.450) * (2 * 1503.000 + (182.173 + 39.450) * \tan(0.00))$$

$$= 0.3E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= I_p * e_p = 55.550 * 14.000$$

$$= 777.7001 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * \tan(\text{Psi}) / 2 = 78.900^2 * \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do / 2 * (eas + eap)$$

$$= ris * (182.173 + 88.900 / 2) * 1503.000 + 3.500 / 2 * (11.000 + 14.000)$$

$$= 341.7256 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= Ibo * dib / 2 = 20.482 * 78.900 / 2$$

$$= 808.0030 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(D_p - 2 * c_{ext} - do / \text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(200.000 - 2.0 * 0.000 - 88.900 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 55.5500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$Afs = (Iso + eab) * eas = (182.173 + 5.000) * 11.000 = 2003.9032 \text{ mm}^2$$

$$Afb = Ibo * eb + Ibi * (eb - ci + co) = 20.482 * 5.000 + 0.000 * (5.000 - 3.000 + 0.000)$$

$$= 102.4085 \text{ mm}^2$$

$$Afp = I_p * eap = 0.667 * 14.000 = 777.7001 \text{ mm}^2$$

$$Afw = (wshell^2 + wpad^2 + winside^2) / 2$$

$$= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

FileName : P3902-BC

Nozzle Calcs. : R12

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$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= (Afs+Afw)*(fs-0.5P)+Afp*(\text{Min}(fs,fp) - 0.5P) + Afb(\text{Min}(fs,fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 627.8804 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * ( Aps + Apb + 0.5 * APpsi ) \\ &= 0.600 * ( 341725.625 + 808.003 + 0.5*0.000 ) \\ &= 205.5027 \times 10^3 \text{ N} \end{aligned}$$

Since  $Fa \geq Pa$ , Code Requirements are satisfied.

Checking the Nozzle Reinforcement at Right Angle to the Vessel Axis:

Points where Nozzle Inside Diameter Cuts ris [x1,y1 : x2,y2]:

$$(960.5500, 1156.0072 : 1039.4500, 1085.6116)$$

Area [APsi]

$$= 2711.5723 \text{ mm}^2$$

Angle Nozzle Outside Diameter Subtends Inside Radius of Head [Alpha]:

$$\begin{aligned} &= \text{asin}((1000.000 + 88.900 / 2) / 1503.000) - \text{asin}((1000.000 - 88.900 / 2) / 1503.000) \\ &= 0.0793 \text{ Radians} \end{aligned}$$

Angle Subtended By Iso [Beta]

$$= \text{Iso} / \text{ris} = 7.172 / 59.173 = 0.1212 \text{ Radians}$$

Area [Aps]:

$$\begin{aligned} &= (\text{Alpha} + \text{Beta}) * \text{ris}^2 / 2 = (0.079 + 0.121) * 1503.000^2 / 2 \\ &= 0.2E+06 \text{ mm}^2 \end{aligned}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= (Afs+Afw)*(fs-0.5P)+Afp*(\text{Min}(fs,fp) - 0.5P) + Afb(\text{Min}(fs,fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 627.8804 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Fa]:

$$= P * ( Aps + Apb + 0.5 * APpsi )$$

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FileName : P3902-BC -----

Nozzle Calcs. : R12

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$$= 0.600 * ( 226472.625 + 808.003 + 0.5 * 2711.572 )$$

$$= 137.1702 \times 10^3 \text{ N}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 41.4076 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 169.1217 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R17a

Nozl: 22 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R17a From : 20**

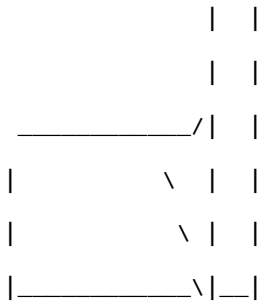
Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Cylinder/Cone Centerline	L1	1000.0000	mm
Distance from Bottom/Left Tangent		8700.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P285QH	
Allowable Stress at Temperature	Sn	162.51	MPa
Allowable Stress At Ambient	Sna	162.51	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		283.00	deg
Diameter		51.0000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	12.5000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	132.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Class of attached Flange		40	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



**Insert Nozzle No Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R17a**

Actual Outside Diameter Used in Calculation 51.000 mm.

Actual Thickness Used in Calculation 12.500 mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$\begin{aligned}
 &= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:} \\
 &= 0.600 * 32.000 / ( 2 * 162.506 * 0.000 - 0.600 ) + 3.000 \\
 &= -29.000 \text{ mm}
 \end{aligned}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$\begin{aligned}
 &= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2 \\
 &= 1503.0000 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$\begin{aligned}
 &= \min( \text{sqrt}( ( deb - 2 * cext - eab ) * eab ), ho ) \\
 &= \min( \text{sqrt}( ( 51.000 - 2 * 0.000 - 9.500 ) * 9.500 ), 132.000 ) \\
 &= 19.8557 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$\begin{aligned}
 &= \text{Min}( hi, 0.5 * lbo ) = \text{Min}( 0.000, 0.5 * 19.856 ) \\
 &= 0.0000 \text{ mm}
 \end{aligned}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$\begin{aligned}
 &= 0.5 * ( Iso + a ) * ( 2 * ris + ( Iso + a ) * \tan( Alpha ) ) \\
 &= 0.5 * ( 182.173 + 16.000 ) * ( 2 * 1503.000 + ( 182.173 + 16.000 ) * \tan( 0.00 ) ) \\
 &= 0.3E+06 \text{ mm}^2
 \end{aligned}$$

Compute area in the re-pad [Afp]:

$$= Ip * ep = 0.000 * 0.000 = 0.0000 \text{ mm}^2$$



FileName : P3902-BC

Nozzle Calcs. : R17a

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Compute triangular area [Ap,psi]:

$$= \text{dib}^2 * \tan(\text{Psi}) / 2 = 32.000^2 * \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= \text{ris} * (\text{Iso} + \text{do} / 2) * \text{ris} + \text{do}/2 * (\text{eas} + \text{eap})$$

$$= \text{ris} * (182.173 + 51.000 / 2) * 1503.000 + 2.008 / 2 * (11.000 + 0.000)$$

$$= 312.4130 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= \text{Ibo} * \text{dib} / 2 = 19.856 * 32.000 / 2$$

$$= 317.6917 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(\text{Dp} - 2 * \text{cext} - \text{do}/\text{Cos}(\text{Phi}), 0) / 2, \text{Iso})$$

$$= \text{Min}(\text{Max}(0.000 - 2.0 * 0.000 - 51.000 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 0.0000 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$\text{Afs} = (\text{Iso} + \text{eab}) * \text{eas} = (182.173 + 9.500) * 11.000 = 2003.9032 \text{ mm}^2$$

$$\text{Afb} = \text{Ibo} * \text{eb} + \text{Ibi} * (\text{eb} - \text{ci} + \text{co}) = 19.856 * 9.500 + 0.000 * (9.500 - 3.000 + 0.000)$$

$$= 188.6294 \text{ mm}^2$$

$$\text{Afp} = \text{Ip} * \text{eap} = 0.270 * 0.000 = 0.0000 \text{ mm}^2$$

$$\text{Afw} = (\text{wshell}^2 + \text{wpad}^2 + \text{winside}^2) / 2$$

$$= (14.0000^2 + 0.0000^2 + 0.0000^2) / 2 = 98.0000 \text{ mm}^2$$

Note: There is No Reinforcement Pad

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$\text{fob} = \text{Min}(\text{fs}, \text{fb}) = \text{Min}(204.174, 162.506) = 162.506 \text{ MPa}$$

$$\text{fop} = \text{Min}(\text{fs}, \text{fp}) = \text{Min}(204.174, 0.000) = 0.000 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$= (\text{Afs} + \text{Afw}) * (\text{fs} - 0.5P) + \text{Afp} * (\text{Min}(\text{fs}, \text{fp}) - 0.5P) + \text{Afb} * (\text{Min}(\text{fs}, \text{fb}) - 0.5P)$$

$$= (2003.903 + 98.000) * (204.174 - 0.5 * 0.600) +$$

$$(0.000 * (\text{Min}(204.174, 0.000) - 0.50 * 0.600) +$$

$$(188.629 * (\text{Min}(204.174, 162.506) - 0.50 * 0.600)) = 459.0808 \times 10^3 \text{ N}$$

FileName : P3902-BC

Nozzle Calcs. : R17a

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Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned}
 &= P * ( Aps + Apb + 0.5 * APpsi ) \\
 &= 0.600 * ( 312413.000 + 317.692 + 0.5*0.000 ) \\
 &= 187.6225 \times 10^3 \text{ N}
 \end{aligned}$$

Since  $F_a \geq P_a$ , Code Requirements are satisfied.

Checking the Nozzle Reinforcement at Right Angle to the Vessel Axis:

Points where Nozzle Inside Diameter Cuts ris [x1,y1 : x2,y2]:

$$(984.0000 , 1136.1130 : 1016.0000 , 1107.5889 )$$

Area [APsi]

$$= 452.0270 \text{ mm}^2$$

Angle Nozzle Outside Diameter Subtends Inside Radius of Head [Alpha]:

$$\begin{aligned}
 &= \text{asin}((1000.000 + 51.000 / 2) / 1503.000) - \text{asin}((1000.000 - 51.000 / 2) / 1503.000) \\
 &= 0.0455 \text{ Radians}
 \end{aligned}$$

Angle Subtended By Iso [Beta]

$$= \text{Iso} / \text{ris} = 7.172 / 59.173 = 0.1212 \text{ Radians}$$

Area [Aps]:

$$= (\text{Alpha} + \text{Beta}) * \text{ris}^2 / 2 = (0.045 + 0.121) * 1503.000^2 / 2 = 0.2E+06 \text{ mm}^2$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned}
 &= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb * (\text{Min}(fs, fb) - 0.5P) \\
 &= (2003.903 + 98.000) * (204.174 - 0.5 * 0.600) + \\
 &\quad (0.000 * (\text{Min}(204.174, 0.000) - 0.50 * 0.600)) + \\
 &\quad (188.629 * (\text{Min}(204.174, 162.506) - 0.50 * 0.600)) \\
 &= 459.0808 \times 10^3 \text{ N}
 \end{aligned}$$

Force Term per 9.5.2.1.1 [Fa]:

$$\begin{aligned}
 &= P * ( Aps + Apb + 0.5 * APpsi ) \\
 &= 0.600 * ( 188256.531 + 317.692 + 0.5 * 452.027 ) \\
 &= 113.2705 \times 10^3 \text{ N}
 \end{aligned}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 23.3424 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 174.0564 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R20

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**INPUT VALUES, Nozzle Description: R20 From : 20**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Cylindrical Shell	D	3000.00	mm
Shell Finished (Minimum) Thickness	t	14.0000	mm
Shell Internal Corrosion Allowance	c	3.0000	mm
Shell External Corrosion Allowance	co	0.0000	mm
Distance from Bottom/Left Tangent		8970.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P285QH	
Allowable Stress at Temperature	Sn	162.51	MPa
Allowable Stress At Ambient	Sna	162.51	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		0.00	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	150.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm
Thickness of Pad	te	14.0000	mm

FileName : P3902-BC

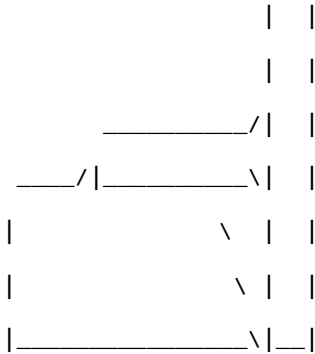
Nozzle Calcs. : R20

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Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R20**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$\begin{aligned}
 &= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:} \\
 &= 0.600 * 78.900 / ( 2 * 162.506 * 0.000 - 0.600 ) + 3.000 \\
 &= -75.900 \text{ mm}
 \end{aligned}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Inside Radius per paragraph 9.5.3 [ris]:

$$\begin{aligned}
 &= ( Di + 2 * cs ) / 2 = ( 3000.000 + 2 * 3.000 ) / 2 \\
 &= 1503.0000 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.76 [lbo]:

$$\begin{aligned}
 &= \min( \text{sqrt}( ( deb - 2 * cext - eab) * eab), ho ) \\
 &= \min(\text{sqrt}((88.900 - 2*0.000 - 5.000 ) * 5.000 ), 150.000 ) \\
 &= 20.4817 \text{ mm}
 \end{aligned}$$

Credit Distance Along the Nozzle per 9.5.77 [lbi]:

$$\begin{aligned}
 &= \text{Min}(hi, 0.5*Ibo) = \text{Min}(0.000, 05*20.482 ) \\
 &= 0.0000 \text{ mm}
 \end{aligned}$$

Compute cone pressure area per Equation 9.5.36 [Aps]:

$$= 0.5 * (Iso + a) * (2 * ris + (Iso + a) * \tan(\text{Alpha}))$$

$$= 0.5 * (182.173 + 39.450) * (2 * 1503.000 + (182.173 + 39.450) * \tan(0.00))$$

$$= 0.3E+06 \text{ mm}^2$$

Compute area in the re-pad [Afp]:

$$= I_p * e_p = 55.550 * 14.000$$

$$= 777.7001 \text{ mm}^2$$

Compute triangular area [Ap.psi]:

$$= dib^2 * \tan(\text{Psi}) / 2 = 78.900^2 * \tan(0.000)$$

$$= 0.0000 \text{ mm}^2$$

Calculate the Pressure Area per 9.5-29 [Aps]:

Note: do (nozzle O/Dia.) taken as the diameter of the hole in the shell

$$= ris * (Iso + do / 2) * ris + do / 2 * (eas + eap)$$

$$= ris * (182.173 + 88.900 / 2) * 1503.000 + 3.500 / 2 * (11.000 + 14.000)$$

$$= 341.7256 \times 10^3 \text{ mm}^2$$

Pressure Area of Nozzle per using dib and Ibo [Apb]:

(This is an inserted nozzle)

$$= I_{bo} * dib / 2 = 20.482 * 78.900 / 2$$

$$= 808.0030 \text{ mm}^2$$

Compute the effective re-pad width [Ip]:

$$= \text{Min}(\text{Max}(D_p - 2 * c_{ext} - do / \text{Cos}(\text{Phi}), 0) / 2, Iso)$$

$$= \text{Min}(\text{Max}(200.000 - 2.0 * 0.000 - 88.900 / \text{Cos}(0.000), 0.0) / 2.0, 182.173)$$

$$= 55.5500 \text{ mm}$$

Available Metal Areas per 9.5-78 to 9.5-81 [Afs, Afb, Afp, Afw]:

Note: The welds are considered in the corroded condition

$$A_{fs} = (Iso + e_{ab}) * e_{as} = (182.173 + 5.000) * 11.000 = 2003.9032 \text{ mm}^2$$

$$A_{fb} = I_{bo} * e_b + I_{bi} * (e_b - c_i + c_o) = 20.482 * 5.000 + 0.000 * (5.000 - 3.000 + 0.000)$$

$$= 102.4085 \text{ mm}^2$$

$$A_{fp} = I_p * e_{ap} = 0.667 * 14.000 = 777.7001 \text{ mm}^2$$

$$A_{fw} = (w_{shell}^2 + w_{pad}^2 + w_{inside}^2) / 2$$

$$= (14.0000^2 + 14.0000^2 + 0.0000^2) / 2 = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]

$$f_{ob} = \text{Min}(f_s, f_b) = \text{Min}(204.174, 162.506) = 162.506 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Force requirement per paragraph 9.5.2.1.1 [Fa]:

$$\begin{aligned} &= (Afs+Af_w)*(fs-0.5P)+Afp*(\text{Min}(fs,fp) - 0.5P) + Afb(\text{Min}(fs,fb) - 0.5P) \\ &= (2003.903 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (102.408 * (\text{Min}(204.174, 162.506) - 0.50 * 0.600) \\ &= 623.6136 \times 10^3 \text{ N} \end{aligned}$$

Force Term per 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * ( Aps + Apb + 0.5 * APpsi ) \\ &= 0.600 * ( 341725.625 + 808.003 + 0.5*0.000 ) \\ &= 205.5027 \times 10^3 \text{ N} \end{aligned}$$

Since  $F_a \geq P_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 0.6588 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 164.6588 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : GV1b

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**INPUT VALUES, Nozzle Description: GV1b From: 30**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	0.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		ID	
Layout Angle		0.00	deg
Diameter		508.0000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P355NH	
Flange Type		Slip on	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	250.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	800.0000	mm

FileName : P3902-BC

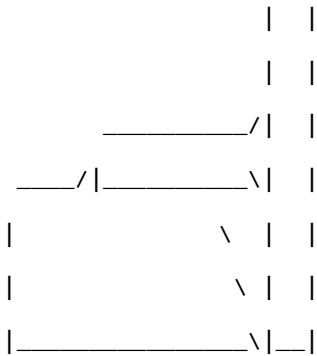
Nozzle Calcs. : GV1b

Nozl: 24 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		136.0000	mm
Class of attached Flange		10	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: GV1b**

Actual Inside Diameter Used in Calculation	508.000	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 514.000 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000$$

$$= 3.756 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Get the head depth from the head dimensions [h]:

$$\text{Ar} = \text{Aspect ratio of the ellipse head}$$

$$= D/2 * \text{Ar} + ci = 3000.000 / 2*2.000 + 3.000 = 753.0000 \text{ mm}$$

Compute the spherical radius per 9.5.1 [ris]:

$$= Di*(0.44 * \text{Ar} + 0.02) = 3050.800 * (0.44*2.000 + 0.02) = 2705.3999 \text{ mm}$$

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2*ris+eas)*eas) = \text{Sqrt}(2*2705.400 + 11.000 ) * 11.000 )$$

$$= 244.2126 \text{ mm}$$



Credit Distance Along the Nozzle outside per 9.5.7.1 [lbo]:

$$= \min(\text{sqrt}((de - eab) * eab), ho)$$

$$= \min(\text{sqrt}((528.0000 - 7.0000) * 7.0000), 250.0000)$$

$$= 60.3904 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [lbi]:

$$= \text{Min}(hi, 0.50 * Ibo) = \text{Min}(0.000, 0.50 * 60.390) = 0.0000 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= ris + eas / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}(\text{Offset} / Rm) - \text{asin}((\text{Offset} - 0.5 * dib) / Rm)$$

$$= \text{asin}(0.000 / 2710.900) - \text{asin}((0.000 - 0.5 * 528.000) / 2710.900)$$

$$= 0.0975 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * rms = 0.098 * 2710.900 = 264.4191 \text{ mm}$$

Angle between nozzle centreline and head centrlne [Beta]:

$$= \text{Asin}(\text{Offset} / rms) = \text{Asin}(0.000 / 2710.900) = 0.0000 \text{ radians}$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * ris^2 * (Iso + a) / (0.5 * eas + ris) + a * (eas + eap)$$

$$= 0.5 * 2705.400 ** 2 * (244.213 + 264.419) / (0.5 * 11.000 + 2705.400) +$$

$$264.419 * (11.000 + 14.000) = 693.2406 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \min(ho, Ibo) * dib * 0.5 = \min(250.000, 60.390) * 514.000 * 0.5$$

$$= 15520.3320 \text{ mm}^2$$

$$Afs = Iso * eas = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$Afb = Ibo * eab + Ibi * \text{Max}(eab - cn + cnext, 0)$$

$$= 60.390 * 7.000 + 0.000 * 0.000$$

$$= 422.7327 \text{ mm}^2$$

$$Afp = Ip * eap = 136.000 * 14.000 = 1904.0001 \text{ mm}^2$$

$$Afw = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= (A_{fs}+A_{fw}) * (f_s-0.5P)+A_{fp} * (\text{Min}(f_s, f_p) - 0.5P) + A_{fb}(\text{Min}(f_s, f_b) - 0.5P) \\ &= (2686.339 + 196.000 ) * (204.174 - 0.5 * 0.600 ) + \\ &\quad (1904.000 * (\text{Min}(204.174 , 204.174 ) - 0.50 * 0.600 ) + \\ &\quad (422.733 * (\text{Min}(204.174 , 204.174 ) - 0.50 * 0.600 ) \\ &= 1061.9032 \times 10^3 \text{ N} \end{aligned}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * ( A_{ps} + A_{pb} ) \\ &= 0.600 * ( 693240 + 15520.332 ) = 425.2205 \times 10^3 \text{ N} \end{aligned}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 12.8839 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 276.8838 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R19

Nozl: 25 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R19 From : 30**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	850.0001	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		ID	
Layout Angle		180.00	deg
Diameter		219.1000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	10.0000	mm
Flange Material		P285QH	
Flange Type		Slip on	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	250.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	310.0000	mm

FileName : P3902-BC

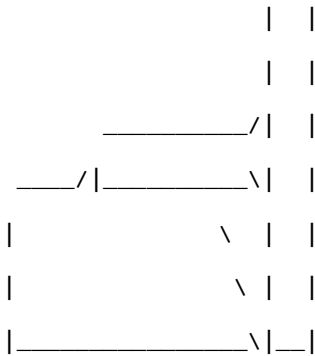
Nozzle Calcs. : R19

Nozl: 25 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		35.4500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R19**

Actual Inside Diameter Used in Calculation	219.100	mm.
Actual Thickness Used in Calculation	10.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 225.100 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000$$

$$= 3.331 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Note: This Is a Hillside Nozzle (Offset)

Get the head depth from the head dimensions [h]:

$$\text{Ar} = \text{Aspect ratio of the ellipse head}$$

$$= D/2 * \text{Ar} + ci = 3000.000 / 2 * 2.000 + 3.000 = 753.0000 \text{ mm}$$

Compute the spherical radius per 9.5.1 [ris]:

$$= Di * (0.44 * \text{Ar} + 0.02) = 3050.800 * (0.44 * 2.000 + 0.02) = 2705.3999 \text{ mm}$$

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2 * ris + eas) * eas) = \text{Sqrt}(2 * 2705.400 + 11.000) * 11.000)$$

$$= 244.2126 \text{ mm}$$

Credit Distance Along the Nozzle outside per 9.5.7.1 [lbo]:

$$= \min(\text{sqrt}((de - eab) * eab), ho)$$

$$= \min(\text{sqrt}((239.1000 - 7.0000) * 7.0000), 250.0000)$$

$$= 40.3076 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [lbi]:

$$= \text{Min}(hi, 0.50 * Ibo) = \text{Min}(0.000, 0.50 * 40.308) = 0.0000 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= ris + eas / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}(\text{Offset} / Rm) - \text{asin}((\text{Offset} - 0.5 * dib) / Rm)$$

$$= \text{asin}(850.000 / 2710.900) - \text{asin}((850.000 - 0.5 * 239.100) / 2710.900)$$

$$= 0.0461 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * rms = 0.046 * 2710.900 = 124.9917 \text{ mm}$$

Angle between nozzle centreline and head centreline [Beta]:

$$= \text{Asin}(\text{Offset} / rms) = \text{Asin}(850.000 / 2710.900) = 0.3189 \text{ radians}$$

Small triangular area fig 9.5-3 [Ap\_psi]:

$$= 0.5 * dib^2 * \text{tan}(\text{Beta}) = 0.5 * 225.100^2 * \text{Tan}(0.319) = 8365.6270 \text{ mm}^2$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * ris^2 * (Iso + a) / (0.5 * eas + ris) + a * (eas + eap)$$

$$= 0.5 * 2705.400^2 * (244.213 + 124.992) / (0.5 * 11.000 + 2705.400) +$$

$$124.992 * (11.000 + 14.000) = 501.5342 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \min(ho, Ibo) * dib * 0.5 = \min(250.000, 40.308) * 225.100 * 0.5$$

$$= 4536.6162 \text{ mm}^2$$

$$Afs = Iso * eas = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$Afb = Ibo * eab + Ibi * \text{Max}(eab - cn + cnext, 0)$$

$$= 40.308 * 7.000 + 0.000 * 0.000$$

$$= 282.1529 \text{ mm}^2$$

$$Afp = Ip * eap = 35.450 * 14.000 = 496.3000 \text{ mm}^2$$

$$Afw = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb(\text{Min}(fs, fb) - 0.5P) \\ &= (2686.339 + 196.000) * (204.174 - 0.5 * 0.600) + \\ &\quad (496.300 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) + \\ &\quad (282.153 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) \\ &= 746.2762 \times 10^3 \text{ N} \end{aligned}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned} &= P * (Aps + Apb + .5 * APpsi) \\ &= 0.600 * (501534 + 4536.616 + 0.5 * 0.000) = 306.1262 \times 10^3 \text{ N} \end{aligned}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 42.5988 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 307.3380 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R15

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**INPUT VALUES, Nozzle Description: R15 From : 30**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	700.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		90.00	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	163.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm

FileName : P3902-BC

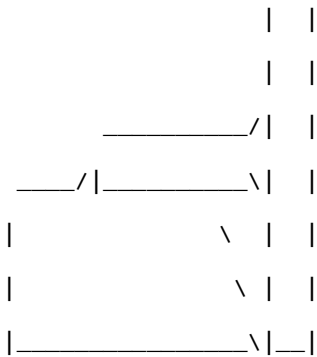
Nozzle Calcs. : R15

Nozl: 26 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R15**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000$$

$$= 3.116 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Note: This Is a Hillside Nozzle (Offset)

Get the head depth from the head dimensions [h]:

$$\text{Ar} = \text{Aspect ratio of the ellipse head}$$

$$= D/2 * \text{Ar} + ci = 3000.000 / 2*2.000 +3.000 = 753.0000 \text{ mm}$$

Compute the spherical radius per 9.5.1 [ris]:

$$= Di*(0.44 * \text{Ar} + 0.02) = 3050.800 * (0.44*2.000 +0.02) = 2705.3999 \text{ mm}$$



FileName : P3902-BC

Nozzle Calcs. : R15

Nozl: 26 7:08a Nov 7,2012

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2*\text{ris}+\text{eas})*\text{eas}) = \text{Sqrt}(2*2705.400 + 11.000 ) * 11.000 )$$

$$= 244.2126 \text{ mm}$$

Credit Distance Along the Nozzle outside per 9.5.7.1 [Ibo]:

$$= \text{min}( \text{sqrt}( ( \text{de} - \text{eab} ) * \text{eab} ), \text{ho} )$$

$$= \text{min}( \text{sqrt}((88.9000 - 5.0000 ) * 5.0000 , 163.0000 )$$

$$= 20.4817 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [Ibi]:

$$= \text{Min}(\text{hi}, 0.50 * \text{Ibo}) = \text{Min}(0.000 , 0.50 * 20.482 ) = 0.0000 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= \text{ris} + \text{eas} / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}( \text{Offset} / \text{Rm} ) - \text{asin}( ( \text{Offset} - 0.5 * \text{dib} ) / \text{Rm} )$$

$$= \text{asin}(700.000 / 2710.900 ) - \text{asin}((700.000 - 0.5 * 88.900 ) / 2710.900 )$$

$$= 0.0169 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * \text{rms} = 0.017 * 2710.900 = 45.9087 \text{ mm}$$

Angle between nozzle centreline and head centrelines [Beta]:

$$= \text{Asin}(\text{Offset} / \text{rms}) = \text{Asin}(700.000 ) / 2710.900 ) = 0.2612 \text{ radians}$$

Small triangular area fig 9.5-3 [Ap\_psi]:

$$= 0.5 * \text{dib}^2 * \text{tan}(\text{Beta}) = 0.5 * 78.900^2 * \text{Tan}(0.261 ) = 831.9406 \text{ mm}^2$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * \text{ris}^2 * ( \text{Iso} + \text{a} ) / ( 0.5 * \text{eas} + \text{ris} ) + \text{a} * ( \text{eas} + \text{eap} )$$

$$= 0.5 * 2705.400 ** 2 * (244.213 + 45.909 ) / (0.5 * 11.000 + 2705.400 ) +$$

$$45.909 * (11.000 + 14.000 ) = 392.7986 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \text{min}(\text{ho}, \text{Ibo}) * \text{dib} * 0.5 = \text{min}(163.000 , 20.482 ) * 78.900 * 0.5$$

$$= 808.0030 \text{ mm}^2$$

$$\text{Afs} = \text{Iso} * \text{eas} = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$\text{Afb} = \text{Ibo} * \text{eab} + \text{Ibi} * \text{Max}(\text{eab} - \text{cn} + \text{cnext}, 0 )$$

$$= 20.482 * 5.000 + 0.000 * 0.000$$

$$= 102.4085 \text{ mm}^2$$

$$\text{Afp} = \text{Ip} * \text{eap} = 55.550 * 14.000 = 777.7001 \text{ mm}^2$$

FileName : P3902-BC

Nozzle Calcs. : R15

Noz1: 26 7:08a Nov 7,2012

$$A_{fw} = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

$$f_{ob} = \text{Min}(f_s, f_b) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$f_{op} = \text{Min}(f_s, f_p) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$= (A_{fs} + A_{fw}) * (f_s - 0.5P) + A_{fp} * (\text{Min}(f_s, f_p) - 0.5P) + A_{fb} (\text{Min}(f_s, f_b) - 0.5P)$$

$$= (2686.339 + 196.000) * (204.174 - 0.5 * 0.600) +$$

$$(777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) +$$

$$(102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600))$$

$$= 766.9993 \times 10^3 \text{ N}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$= P * (A_{ps} + A_{pb} + .5 * A_{Ppsi})$$

$$= 0.600 * (392798.594 + 808.003 + 0.5 * 0.000) = 236.3935 \times 10^3 \text{ N}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 12.2853 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 189.7740 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R18b

Nozl: 27 7:08a Nov 7,2012

**NPUT VALUES, Nozzle Description: R18b From : 30**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	860.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		126.00	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	167.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm

FileName : P3902-BC

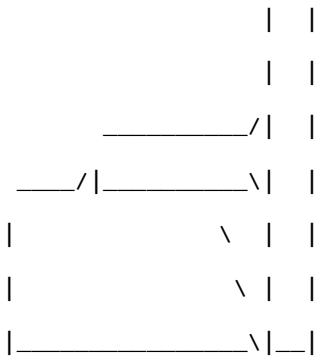
Nozzle Calcs. : R18b

Nozl: 27 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R18b**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000$$

$$= 3.116 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Note: This Is a Hillside Nozzle (Offset)

Get the head depth from the head dimensions [h]:

$$Ar = \text{Aspect ratio of the ellipse head}$$

$$= D/2 * Ar + ci = 3000.000 / 2 * 2.000 + 3.000 = 753.0000 \text{ mm}$$

Compute the spherical radius per 9.5.1 [ris]:

$$= Di * (0.44 * Ar + 0.02) = 3050.800 * (0.44 * 2.000 + 0.02) = 2705.3999 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : R18b

Nozl: 27 7:08a Nov 7,2012

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2*\text{ris}+\text{eas})*\text{eas}) = \text{Sqrt}(2*2705.400 + 11.000 ) * 11.000 )$$

$$= 244.2126 \text{ mm}$$

Credit Distance Along the Nozzle outside per 9.5.7.1 [Ibo]:

$$= \text{min}( \text{sqrt}( ( \text{de} - \text{eab} ) * \text{eab} ), \text{ho} )$$

$$= \text{min}( \text{sqrt}((88.9000 - 5.0000 ) * 5.0000 , 167.0000 )$$

$$= 20.4817 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [Ibi]:

$$= \text{Min}(\text{hi}, 0.50 * \text{Ibo}) = \text{Min}(0.000 , 0.50 * 20.482 ) = 0.0000 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= \text{ris} + \text{eas} / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}( \text{Offset} / \text{Rm} ) - \text{asin}( ( \text{Offset} - 0.5 * \text{dib} ) / \text{Rm} )$$

$$= \text{asin}(860.000 / 2710.900 ) - \text{asin}((860.000 - 0.5 * 88.900 ) / 2710.900 )$$

$$= 0.0172 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * \text{rms} = 0.017 * 2710.900 = 46.7386 \text{ mm}$$

Angle between nozzle centreline and head centrline [Beta]:

$$= \text{Asin}(\text{Offset} / \text{rms}) = \text{Asin}(860.000 ) / 2710.900 ) = 0.3228 \text{ radians}$$

Small triangular area fig 9.5-3 [Ap\_psi]:

$$= 0.5 * \text{dib}^2 * \text{tab}(\text{Beta}) = 0.5 * 78.900^2 * \text{Tan}(0.323 ) = 1041.2191 \text{ mm}^2$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * \text{ris}^2 * ( \text{Iso} + \text{a} ) / ( 0.5 * \text{eas} + \text{ris} ) + \text{a} * ( \text{eas} + \text{eap} )$$

$$= 0.5 * 2705.400 ** 2 * (244.213 + 46.739 ) / (0.5 * 11.000 + 2705.400 ) +$$

$$46.739 * (11.000 + 14.000 ) = 393.9397 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \text{min}(\text{ho}, \text{Ibo}) * \text{dib} * 0.5 = \text{min}(167.000 , 20.482 ) * 78.900 * 0.5$$

$$= 808.0030 \text{ mm}^2$$

$$\text{Afs} = \text{Iso} * \text{eas} = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$\text{Afb} = \text{Ibo} * \text{eab} + \text{Ibi} * \text{Max}(\text{eab} - \text{cn} + \text{cnext}, 0 )$$

$$= 20.482 * 5.000 + 0.000 * 0.000$$

$$= 102.4085 \text{ mm}^2$$

$$\text{Afp} = \text{Ip} * \text{eap} = 55.550 * 14.000 = 777.7001 \text{ mm}^2$$

FileName : P3902-BC

Nozzle Calcs. : R18b

Noz1: 27 7:08a Nov 7,2012

$$A_{fw} = 196.0000 \text{ mm}^2$$

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

$$f_{ob} = \text{Min}(f_s, f_b) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

$$f_{op} = \text{Min}(f_s, f_p) = \text{Min}(204.174, 204.174) = 204.174 \text{ MPa}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$= (A_{fs} + A_{fw}) * (f_s - 0.5P) + A_{fp} * (\text{Min}(f_s, f_p) - 0.5P) + A_{fb} (\text{Min}(f_s, f_b) - 0.5P)$$

$$= (2686.339 + 196.000) * (204.174 - 0.5 * 0.600) +$$

$$(777.700 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600) +$$

$$(102.408 * (\text{Min}(204.174, 204.174) - 0.50 * 0.600))$$

$$= 766.9993 \times 10^3 \text{ N}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$= P * (A_{ps} + A_{pb} + .5 * A_{Ppsi})$$

$$= 0.600 * (393939.719 + 808.003 + 0.5 * 0.000) = 237.1408 \times 10^3 \text{ N}$$

Since  $P_a \geq F_a$ , Code Requirements are satisfied.

The Drop for this Nozzle is : 15.2971 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 197.0551 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R16

Nozl: 28 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R16 From : 30**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	1100.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		141.00	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	167.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm

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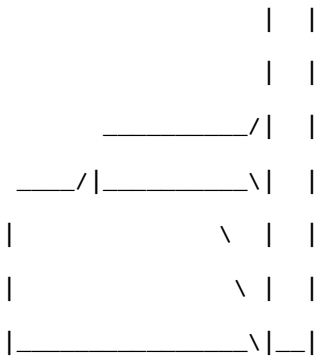
FileName : P3902-BC -----

Nozzle Calcs. : R16 Nozl: 28 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R16**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000$$

$$= 3.116 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Note: This Is a Hillside Nozzle (Offset)

Get the head depth from the head dimensions [h]:

$$\text{Ar} = \text{Aspect ratio of the ellipse head}$$

$$= D/2 * \text{Ar} + ci = 3000.000 / 2 * 2.000 + 3.000 = 753.0000 \text{ mm}$$

Compute the spherical radius per 9.5.1 [ris]:

$$= Di * (0.44 * \text{Ar} + 0.02) = 3050.800 * (0.44 * 2.000 + 0.02) = 2705.3999 \text{ mm}$$



FileName : P3902-BC

Nozzle Calcs. : R16

Nozl: 28 7:08a Nov 7,2012

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2*\text{ris}+\text{eas})*\text{eas}) = \text{Sqrt}(2*2705.400 + 11.000 ) * 11.000 )$$

$$= 244.2126 \text{ mm}$$

Credit Distance Along the Nozzle outside per 9.5.7.1 [Ibo]:

$$= \text{min}( \text{sqrt}( ( \text{de} - \text{eab} ) * \text{eab} ), \text{ho} )$$

$$= \text{min}( \text{sqrt}((88.9000 - 5.0000 ) * 5.0000 , 167.0000 )$$

$$= 20.4817 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [Ibi]:

$$= \text{Min}(\text{hi}, 0.50 * \text{Ibo}) = \text{Min}(0.000 , 0.50 * 20.482 ) = 0.0000 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= \text{ris} + \text{eas} / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}( \text{Offset} / \text{Rm} ) - \text{asin}( ( \text{Offset} - 0.5 * \text{dib} ) / \text{Rm} )$$

$$= \text{asin}(1100.000 / 2710.900 ) - \text{asin}((1100.000 - 0.5 * 88.900 ) / 2710.900 )$$

$$= 0.0179 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * \text{rms} = 0.018 * 2710.900 = 48.4440 \text{ mm}$$

Angle between nozzle centreline and head centrelines [Beta]:

$$= \text{Asin}(\text{Offset} / \text{rms}) = \text{Asin}(1100.000 ) / 2710.900 ) = 0.4178 \text{ radians}$$

Small triangular area fig 9.5-3 [Ap\_psi]:

$$= 0.5 * \text{dib}^2 * \text{tan}(\text{Beta}) = 0.5 * 78.900^2 * \text{Tan}(0.418 ) = 1381.8744 \text{ mm}^2$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * \text{ris}^2 * ( \text{Iso} + \text{a} ) / ( 0.5 * \text{eas} + \text{ris} ) + \text{a} * ( \text{eas} + \text{eap} )$$

$$= 0.5 * 2705.400 ** 2 * (244.213 + 48.444 ) / (0.5 * 11.000 + 2705.400 ) +$$

$$48.444 * (11.000 + 14.000 ) = 396.2846 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \text{min}(\text{ho}, \text{Ibo}) * \text{dib} * 0.5 = \text{min}(167.000 , 20.482 ) * 78.900 * 0.5$$

$$= 808.0030 \text{ mm}^2$$

$$\text{Afs} = \text{Iso} * \text{eas} = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$\text{Afb} = \text{Ibo} * \text{eab} + \text{Ibi} * \text{Max}(\text{eab} - \text{cn} + \text{cnext}, 0 )$$

$$= 20.482 * 5.000 + 0.000 * 0.000$$

$$= 102.4085 \text{ mm}^2$$

$$\text{Afp} = \text{Ip} * \text{eap} = 55.550 * 14.000 = 777.7001 \text{ mm}^2$$

Afw = 196.0000 mm<sup>2</sup>

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

fob = Min(fs, fb) = Min(204.174 , 204.174 ) = 204.174 MPa

fop = Min(fs, fp) = Min(204.174 , 204.174 ) = 204.174 MPa

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned}
&= (Afs+Afw)*(fs-0.5P)+Afp*(Min(fs,fp) - 0.5P) + Afb(Min(fs,fb) - 0.5P) \\
&= (2686.339 + 196.000 ) * (204.174 - 0.5 * 0.600 ) + \\
&\quad (777.700 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) + \\
&\quad (102.408 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) \\
&= 766.9993 \times 10^3 \text{ N}
\end{aligned}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned}
&= P * ( Aps + Apb + .5 * APpsi ) \\
&= 0.600 * ( 396284.594 + 808.003 + 0.5*0.000 ) = 238.6498 \times 10^3 \text{ N}
\end{aligned}$$

Since Pa >= Fa, Code Requirements are satisfied.

The Drop for this Nozzle is : 20.2125 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 202.5221 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R14

Nozl: 29 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R14 From : 30**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	1086.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P355NH	
Allowable Stress at Temperature	Sn	204.17	MPa
Allowable Stress At Ambient	Sna	204.17	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		158.00	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	167.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P355NH	
Pad Allowable Stress at Temperature	Sp	204.17	MPa
Pad Allowable Stress At Ambient	Spa	204.17	MPa
Diameter of Pad along vessel surface	Dp	200.0000	mm

FileName : P3902-BC

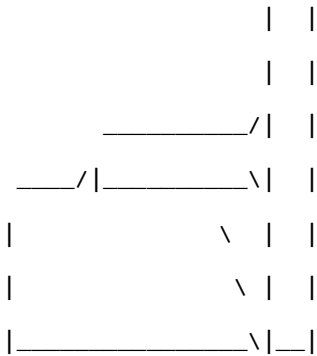
Nozzle Calcs. : R14

Nozl: 29 7:08a Nov 7,2012

Thickness of Pad	te	14.0000	mm
Weld leg size between Pad and Shell	Wp	14.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	14.0000	mm
Reinforcing Pad Width		55.5500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R14**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 204.174 * 1.000 - 0.600 ) + 3.000$$

$$= 3.116 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Note: This Is a Hillside Nozzle (Offset)

Get the head depth from the head dimensions [h]:

$$\text{Ar} = \text{Aspect ratio of the ellipse head}$$

$$= D/2 * \text{Ar} + ci = 3000.000 / 2*2.000 +3.000 = 753.0000 \text{ mm}$$

Compute the spherical radius per 9.5.1 [ris]:

$$= Di*(0.44 * \text{Ar} + 0.02) = 3050.800 * (0.44*2.000 +0.02) = 2705.3999 \text{ mm}$$

FileName : P3902-BC

Nozzle Calcs. : R14

Nozl: 29 7:08a Nov 7,2012

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2*\text{ris}+\text{eas})*\text{eas}) = \text{Sqrt}(2*2705.400 + 11.000 ) * 11.000 )$$

$$= 244.2126 \text{ mm}$$

Credit Distance Along the Nozzle outside per 9.5.7.1 [Ibo]:

$$= \text{min}( \text{sqrt}( ( \text{de} - \text{eab} ) * \text{eab} ), \text{ho} )$$

$$= \text{min}( \text{sqrt}((88.9000 - 5.0000 ) * 5.0000 , 167.0000 )$$

$$= 20.4817 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [Ibi]:

$$= \text{Min}(\text{hi}, 0.50 * \text{Ibo}) = \text{Min}(0.000 , 0.50 * 20.482 ) = 0.0000 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= \text{ris} + \text{eas} / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}( \text{Offset} / \text{Rm} ) - \text{asin}( ( \text{Offset} - 0.5 * \text{dib} ) / \text{Rm} )$$

$$= \text{asin}(1086.000 / 2710.900 ) - \text{asin}((1086.000 - 0.5 * 88.900 ) / 2710.900 )$$

$$= 0.0178 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * \text{rms} = 0.018 * 2710.900 = 48.3271 \text{ mm}$$

Angle between nozzle centreline and head centrelines [Beta]:

$$= \text{Asin}(\text{Offset} / \text{rms}) = \text{Asin}(1086.000 ) / 2710.900 ) = 0.4122 \text{ radians}$$

Small triangular area fig 9.5-3 [Ap\_psi]:

$$= 0.5 * \text{dib}^2 * \text{tan}(\text{Beta}) = 0.5 * 78.900^2 * \text{Tan}(0.412 ) = 1360.8990 \text{ mm}^2$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * \text{ris}^2 * ( \text{Iso} + \text{a} ) / ( 0.5 * \text{eas} + \text{ris} ) + \text{a} * ( \text{eas} + \text{eap} )$$

$$= 0.5 * 2705.400 ** 2 * (244.213 + 48.327 ) / (0.5 * 11.000 + 2705.400 ) +$$

$$48.327 * (11.000 + 14.000 ) = 396.1237 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \text{min}(\text{ho}, \text{Ibo}) * \text{dib} * 0.5 = \text{min}(167.000 , 20.482 ) * 78.900 * 0.5$$

$$= 808.0030 \text{ mm}^2$$

$$\text{Afs} = \text{Iso} * \text{eas} = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$\text{Afb} = \text{Ibo} * \text{eab} + \text{Ibi} * \text{Max}(\text{eab} - \text{cn} + \text{cnext}, 0 )$$

$$= 20.482 * 5.000 + 0.000 * 0.000$$

$$= 102.4085 \text{ mm}^2$$

$$\text{Afp} = \text{Ip} * \text{eap} = 55.550 * 14.000 = 777.7001 \text{ mm}^2$$

Afw = 196.0000 mm<sup>2</sup>

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

fob = Min(fs, fb) = Min(204.174 , 204.174 ) = 204.174 MPa

fop = Min(fs, fp) = Min(204.174 , 204.174 ) = 204.174 MPa

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned}
&= (Afs+Afw)*(fs-0.5P)+Afp*(Min(fs,fp) - 0.5P) + Afb(Min(fs,fb) - 0.5P) \\
&= (2686.339 + 196.000 ) * (204.174 - 0.5 * 0.600 ) + \\
&\quad (777.700 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) + \\
&\quad (102.408 * (Min(204.174 , 204.174 ) - 0.50 * 0.600 ) \\
&= 766.9993 \times 10^3 \text{ N}
\end{aligned}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned}
&= P * ( Aps + Apb + .5 * APpsi ) \\
&= 0.600 * ( 396123.750 + 808.003 + 0.5*0.000 ) = 238.5471 \times 10^3 \text{ N}
\end{aligned}$$

Since Pa >= Fa, Code Requirements are satisfied.

The Drop for this Nozzle is : 19.9094 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 202.1812 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Calcs. : R13

Nozl: 30 7:08a Nov 7,2012

**INPUT VALUES, Nozzle Description: R13 From : 30**

Pressure for Reinforcement Calculations	P	0.6000	MPa
Temperature for Internal Pressure	Temp	60	C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	S	204.17	MPa
Shell Allowable Stress At Ambient	Sa	204.17	MPa
Inside Diameter of Elliptical Head	D	3000.00	mm
Aspect Ratio of Elliptical Head	Ar	2.00	
Head Finished (Minimum) Thickness	t	14.0000	mm
Head Internal Corrosion Allowance	c	3.0000	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	1011.0000	mm

**Type of Element Connected to the Shell : Nozzle**

Nozzle Material		P285QH	
Allowable Stress at Temperature	Sn	162.51	MPa
Allowable Stress At Ambient	Sna	162.51	MPa
Diameter Basis (for tr calc only)		OD	
Layout Angle		81.00	deg
Diameter		88.9000	mm.
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.0000	mm
Flange Material		P285QH	
Flange Type		Weld Neck Flange	
Corrosion Allowance	can	3.0000	mm
Outside Projection	ho	167.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	14.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	14.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
Pad Material		P235GH	
Pad Allowable Stress at Temperature	Sp	135.21	MPa
Pad Allowable Stress At Ambient	Spa	150.01	MPa
Diameter of Pad along vessel surface	Dp	0.0000	mm

FileName : P3902-BC

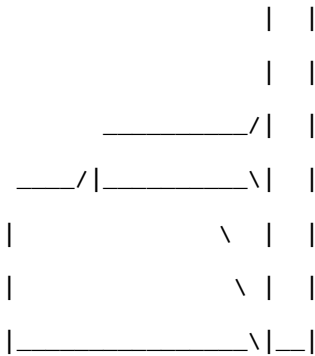
Nozzle Calcs. : R13

Nozl: 30 7:08a Nov 7,2012

Thickness of Pad	te	0.0000	mm
Weld leg size between Pad and Shell	Wp	0.0000	mm
Groove weld depth between Pad and Nozzle	Wgpn	0.0000	mm
Reinforcing Pad Width		-44.4500	mm
Class of attached Flange		16	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

**Nozzle Sketch (may not represent actual weld type/configuration)**



**Insert Nozzle With Pad, no Inside projection**

**Isolated Nozzle Calculation per EN 13445, Description: R13**

Actual Outside Diameter Used in Calculation	88.900	mm.
Actual Thickness Used in Calculation	8.000	mm

Required Thickness of Nozzle Neck due to Internal Pressure:

$$= P * Di / ( 2 * f * z - P ) + c \quad \text{EN-13445 Equation: 7.4.2:}$$

$$= 0.600 * 78.900 / ( 2 * 162.506 * 1.000 - 0.600 ) + 3.000$$

$$= 3.146 \text{ mm}$$

**EN13445 Section 9 - Pressure Area Design Method:**

Note: This Is a Hillside Nozzle (Offset)

Get the head depth from the head dimensions [h]:

$$\text{Ar} = \text{Aspect ratio of the ellipse head}$$

$$= D/2 * \text{Ar} + ci = 3000.000 / 2 * 2.000 + 3.000 = 753.0000 \text{ mm}$$

Compute the spherical radius per 9.5.1 [ris]:

$$= Di * (0.44 * \text{Ar} + 0.02) = 3050.800 * (0.44 * 2.000 + 0.02) = 2705.3999 \text{ mm}$$

Credit Distance Along the Shell per 9.5.1 [Iso]:

$$= \text{Sqrt}((2 * ris + eas) * eas) = \text{Sqrt}(2 * 2705.400 + 11.000) * 11.000)$$

$$= 244.2126 \text{ mm}$$



Credit Distance Along the Nozzle outside per 9.5.7.1 [lbo]:

$$= \min(\text{sqrt}((de - eab) * eab), ho)$$

$$= \min(\text{sqrt}((88.9000 - 5.0000) * 5.0000), 167.0000)$$

$$= 20.4817 \text{ mm}$$

Credit Distance Along the Nozzle per inside 9.5.7.1 [lbi]:

$$= \text{Min}(hi, 0.50 * Ibo) = \text{Min}(0.000, 0.50 * 20.482) = 0.0000 \text{ mm}$$

Mean radius of Head per Figure 9.4-10 [rm]:

$$= ris + eas / 2 = 2705.400 + 11.000 / 2 = 2710.8999 \text{ mm}$$

Angle subtending 'a' in figure 9.5-3 [Alpha]:

$$= \text{asin}(\text{Offset} / Rm) - \text{asin}((\text{Offset} - 0.5 * dib) / Rm)$$

$$= \text{asin}(1011.000 / 2710.900) - \text{asin}((1011.000 - 0.5 * 88.900) / 2710.900)$$

$$= 0.0176 \text{ radians}$$

The distance as depicted in figure 9.5-3 [a]:

$$= \text{Alpha} * rms = 0.018 * 2710.900 = 47.7396 \text{ mm}$$

Angle between nozzle centreline and head centrlne [Beta]:

$$= \text{Asin}(\text{Offset} / rms) = \text{Asin}(1011.000 / 2710.900) = 0.3822 \text{ radians}$$

Small triangular area fig 9.5-3 [Ap\_psi]:

$$= 0.5 * dib^2 * \text{tab}(\text{Beta}) = 0.5 * 78.900^2 * \text{Tan}(0.382) = 1251.0686 \text{ mm}^2$$

Pressure area inside shell/head per 9.5-56 [Aps]:

$$= 0.5 * ris^2 * (Iso + a) / (0.5 * eas + ris) + a * (eas + eap)$$

$$= 0.5 * 2705.400^2 * (244.213 + 47.740) / (0.5 * 11.000 + 2705.400) +$$

$$47.740 * (11.000 + 0.000) = 394.6477 \times 10^3 \text{ mm}^2$$

Area in the nozzle [Apb]:

$$= \min(ho, Ibo) * dib * 0.5 = \min(167.000, 20.482) * 78.900 * 0.5$$

$$= 808.0030 \text{ mm}^2$$

$$Afs = Iso * eas = 244.213 * 11.000 = 2686.3389 \text{ mm}^2$$

$$Afb = Ibo * eab + Ibi * \text{Max}(eab - cn + cnext, 0)$$

$$= 20.482 * 5.000 + 0.000 * 0.000$$

$$= 102.4085 \text{ mm}^2$$

$$Afp = Ip * eap = 0.000 * 0.000 = 0.0000 \text{ mm}^2$$

$$Afw = 98.0000 \text{ mm}^2$$

Note: There is No Reinforcement Pad

Calculation per paragraph 9.5.2.1.1 [fob, fop]:

FileName : P3902-BC

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Nozzle Calcs. : R13

Nozl: 30 7:08a Nov 7,2012

$$fob = \text{Min}(fs, fb) = \text{Min}(204.174, 162.506) = 162.506 \text{ MPa}$$

$$fop = \text{Min}(fs, fp) = \text{Min}(204.174, 135.206) = 135.206 \text{ MPa}$$

Pressure Area Check Term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned}
&= (Afs + Afw) * (fs - 0.5P) + Afp * (\text{Min}(fs, fp) - 0.5P) + Afb(\text{Min}(fs, fb) - 0.5P) \\
&= (2686.339 + 98.000) * (204.174 - 0.5 * 0.600) + \\
&\quad (0.000 * (\text{Min}(204.174, 135.206) - 0.50 * 0.600) + \\
&\quad (102.408 * (\text{Min}(204.174, 162.506) - 0.50 * 0.600)) \\
&= 584.2154 \times 10^3 \text{ N}
\end{aligned}$$

Stress Area term per paragraph 9.5.2.1.1 [Pa]:

$$\begin{aligned}
&= P * (Aps + Apb + .5 * APpsi) \\
&= 0.600 * (394647.656 + 808.003 + 0.5 * 0.000) = 237.6285 \times 10^3 \text{ N}
\end{aligned}$$

Since Pa >= Fa, Code Requirements are satisfied.

The Drop for this Nozzle is : 18.3233 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 200.4051 mm

For possible nozzle interference - See Nozzle Summary

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FileName : P3902-BC

Nozzle Schedule :

Step: 42 7:08a Nov 7,2012

Nozzle Schedule:

Description	Nominal Flange			Noz.	Wall	Re-Pad		Cut
	Size	Sch/Type	O/Dia	Thk	ODia	Thick	Length	
	mm	Cls	mm	mm	mm	mm	mm	
M1	51	-	None	51.000	12.500	-	-	80
R17b	51	40	WNF	51.000	12.500	-	-	74
R17d	51	40	WNF	51.000	12.500	-	-	62
R17a	51	40	WNF	51.000	12.500	-	-	174
R11	88	16	WNF	88.900	8.000	200.00	14.00	164
R9	88	16	WNF	88.900	8.000	200.00	14.00	157
R18a	88	16	WNF	88.900	8.000	200.00	14.00	170
R10	88	16	WNF	88.900	8.000	200.00	14.00	242
R12	88	16	WNF	88.900	8.000	200.00	14.00	169
R20	88	16	WNF	88.900	8.000	200.00	14.00	164
R15	88	16	WNF	88.900	8.000	200.00	14.00	189
R18b	88	16	WNF	88.900	8.000	200.00	14.00	197
R16	88	16	WNF	88.900	8.000	200.00	14.00	202
R14	88	16	WNF	88.900	8.000	200.00	14.00	202
R13	88	16	WNF	88.900	8.000	-	-	200
R3	114	16	WNF	114.300	10.000	310.00	14.00	394
R8	168	16	WNF	168.300	10.000	310.00	14.00	161
R2	168	16	WNF	168.300	10.000	310.00	14.00	161
R7c	168	40	WNF	168.300	10.000	310.00	14.00	116
R7b	168	40	WNF	168.300	10.000	310.00	14.00	116
R7a	168	40	WNF	168.300	10.000	310.00	14.00	116
R4	168	16	WNF	168.300	10.000	310.00	14.00	161
R1	219	16	WNF	219.100	10.000	310.00	14.00	836
R6	219	16	WNF	219.100	10.000	400.00	14.00	163
R5	219	16	WNF	219.100	10.000	400.00	14.00	158
R19	219	16	SlipOn	219.100	10.000	310.00	14.00	307
GV1a	508	10	SlipOn	508.000	10.000	800.00	14.00	275
GV2a	508	10	SlipOn	508.000	10.000	800.00	14.00	285

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GV2b	508	10	SlipOn508.000	10.000	800.00	14.00	285
GV1b	508	10	SlipOn528.000	10.000	800.00	14.00	276

*Note on the Cut Length Calculation:*

The Cut Length is the Outside Projection + Inside Projection + Drop + In Plane Shell Thickness. This value does not include weld gaps, nor does it account for shrinkage.

Please Note: In the case of Oblique Nozzles, the Outside Diameter must be increased. The Re-Pad WIDTH around the nozzle is calculated as follows:

$$\text{Width of Pad} = (\text{Pad Outside Dia. (per above)} - \text{Nozzle Outside Dia.})/2$$

**Nozzle Material and Weld Fillet Leg Size Details:**

Nozzle	Material	Shl Grve Weld mm	Noz Shl/Pad Weld mm	Pad OD Weld mm	Pad Grve Weld mm	Inside Weld mm
M1	P285QH	14.000	14.000	-	-	-
R17b	P285QH	14.000	14.000	-	-	-
R17d	P285QH	14.000	14.000	-	-	-
R17a	P285QH	14.000	14.000	-	-	-
R11	P355NH	14.000	14.000	14.000	14.000	-
R9	P355NH	14.000	14.000	14.000	14.000	-
R18a	P355NH	14.000	14.000	14.000	14.000	-
R10	P355NH	14.000	14.000	14.000	14.000	-
R12	P355NH	14.000	14.000	14.000	14.000	-
R20	P285QH	14.000	14.000	14.000	14.000	-
R15	P355NH	14.000	14.000	14.000	14.000	-
R18b	P355NH	14.000	14.000	14.000	14.000	-
R16	P355NH	14.000	14.000	14.000	14.000	-
R14	P355NH	14.000	14.000	14.000	14.000	-
R13	P285QH	14.000	14.000	-	-	-
R3	P355NH	14.000	14.000	14.000	14.000	-

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Nozzle	Part	1	2	3	4	5
R8	P355NH	14.000	14.000	14.000	14.000	-
R2	P355NH	14.000	14.000	14.000	14.000	-
R7c	P355NH	14.000	14.000	14.000	14.000	-
R7b	P355NH	14.000	14.000	14.000	14.000	-
R7a	P355NH	14.000	14.000	14.000	14.000	-
R4	P355NH	14.000	14.000	14.000	14.000	-
R1	P355NH	14.000	14.000	14.000	14.000	-
R6	P355NH	14.000	14.000	14.000	14.000	-
R5	P355NH	14.000	14.000	14.000	14.000	-
R19	P355NH	14.000	14.000	14.000	14.000	-
GV1a	P355NH	14.000	14.000	14.000	14.000	-
GV2a	P355NH	14.000	14.000	6.000	10.000	-
GV2b	P355NH	14.000	14.000	6.000	10.000	-
GV1b	P355NH	14.000	14.000	14.000	14.000	-

Note: The Outside projections below do not include the flange thickness.

**Nozzle Miscellaneous Data:**

Nozzle	Elevation/Distance	Layout	Projection		Installed In
	From Datum	Angle	Outside	Inside	Component
	mm	deg.	mm	mm	
M1	3910.000	0.00	66.00	0.00	VIROLA
R17b	8700.001	180.00	60.00	0.00	VIROLA
R17d	10290.001	-161.00	38.00	0.00	VIROLA
R17a	8700.001	283.00	132.00	0.00	VIROLA
R11	9150.000	90.00	150.00	0.00	VIROLA
R9	8850.000	95.69	138.00	0.00	VIROLA
R18a	9000.000	113.35	135.00	0.00	VIROLA
R10	9150.000	142.43	157.00	0.00	VIROLA
R12	8850.000	131.34	109.00	0.00	VIROLA
R20	8970.000	0.00	150.00	0.00	VIROLA
R15		90.00	163.00	0.00	FUND ELIPSOI

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R18b		126.00	167.00	0.00	FUND ELIPSOI
R16		141.00	167.00	0.00	FUND ELIPSOI
R14		158.00	167.00	0.00	FUND ELIPSOI
R13		81.00	167.00	0.00	FUND ELIPSOI
R3		180.00	250.00	100.00	FUND ELIPSOI
R8	6990.000	0.00	145.00	0.00	VIROLA
R2	9270.001	0.00	145.00	0.00	VIROLA
R7c	300.000	180.00	100.00	0.00	VIROLA
R7b	2750.000	180.00	100.00	0.00	VIROLA
R7a	7700.000	180.00	100.00	0.00	VIROLA
R4	10150.001	180.00	145.00	0.00	VIROLA
R1		0.00	280.00	490.00	FUND ELIPSOI
R6	10000.001	0.00	145.00	0.00	VIROLA
R5	8960.000	180.00	140.00	0.00	VIROLA
R19		180.00	250.00	0.00	FUND ELIPSOI
GV1a		0.00	250.00	0.00	FUND ELIPSOI
GV2a	550.000	0.00	250.00	0.00	VIROLA
GV2b	8000.000	0.00	250.00	0.00	VIROLA
GV1b		0.00	250.00	0.00	FUND ELIPSOI

**Nozzle Calculation Summary:**

**Per EN 13445 Section 9:**

Description	Pressure Area Result
GV1a	Pressure Area Passed
R1	Pressure Area Passed
R3	Pressure Area Passed
GV2a	Pressure Area Passed
GV2b	Pressure Area Passed
M1	Pressure Area Passed
R8	Pressure Area Passed
R2	Pressure Area Passed
R6	Pressure Area Passed
R7c	Pressure Area Passed
R7b	Pressure Area Passed
R7a	Pressure Area Passed
R5	Pressure Area Passed
R4	Pressure Area Passed
R17b	Pressure Area Passed
R17d	Pressure Area Passed
R11	Pressure Area Passed
R9	Pressure Area Passed
R18a	Pressure Area Passed
R10	Pressure Area Passed
R12	Pressure Area Passed
R17a	Pressure Area Passed
R20	Pressure Area Passed
GV1b	Pressure Area Passed
R19	Pressure Area Passed
R15	Pressure Area Passed
R18b	Pressure Area Passed
R16	Pressure Area Passed
R14	Pressure Area Passed

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R13 Pressure Area Passed

-----  
Check the Spatial Relationship between the Nozzles

From Node	Nozzle Description	X Coordinate,	Layout Angle,	Mean Radius
10	GV1a	0.000	0.000	250.500
10	R1	0.000	0.000	106.050
10	R3	0.000	180.000	53.650
20	GV2a	550.000	0.000	250.500
20	GV2b	8000.000	0.000	250.500
20	M1	3910.000	0.000	20.750
20	R8	6990.000	0.000	80.650
20	R2	9270.000	0.000	80.650
20	R6	10000.001	0.000	106.050
20	R7c	300.000	180.000	80.650
20	R7b	2750.000	180.000	80.650
20	R7a	7700.000	180.000	80.650
20	R5	8960.000	180.000	106.050
20	R4	10150.001	180.000	80.650
20	R17b	8700.001	180.000	20.750
20	R17d	10290.001	-161.000	20.750
20	R11	9150.000	90.000	41.950
20	R9	8850.000	95.686	41.950
20	R18a	9000.000	113.347	41.950
20	R10	9150.000	142.429	41.950
20	R12	8850.000	131.338	41.950
20	R17a	8700.001	283.000	20.750
20	R20	8970.000	0.000	41.950
30	GV1b	0.000	0.000	260.500
30	R19	0.000	180.000	116.050
30	R15	0.000	90.000	41.950
30	R18b	0.000	126.000	41.950
30	R16	0.000	141.000	41.950
30	R14	0.000	158.000	41.950



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30 R13 0.000 81.000 41.950

If any interferences/violations are found, they will be noted below.

Note: Items not derived explicitly will be found in the individual nozzle

calculation. These items include such values as Afls etc.

Nozzles R2 and R20 are a Group per EN 13445 para 9.6.4

-----  
 The Nozzles are Installed In a Cylinder

$$\begin{aligned}
 ris &= (ris1 + ris2)/2 = (1503.000 + 1503.000) / 2 = 1503.0000 \text{ mm} \\
 Lb &= 300.0003 \text{ mm (= Inside center to center distance)} \\
 a1 &= deb1/2 = 84.1500 \text{ and } a2 = deb2/2 = 44.4500 \text{ mm} \\
 k &= 2 - (Lb + a1 + a2)/(Iso1 + Iso2) \\
 &= 2 - (300.000 + 84.150 + 44.450)/(182.173 + 182.173) = 0.8236 \\
 Lb1 &= Lb + a1 + a2 + k * (Iso1 + Iso2) \\
 &= 300.000 + 84.150 + 44.450 + 0.824 * (182.173 + 182.173) \\
 &= 728.6920 \text{ mm} \\
 psi &= atan(ris * abs(Angle2 - Angle1) / abs(y2 - y1)) \\
 &= atan(1503.000 * abs(0.000 - 0.000) / abs(9220.001 - 8920.000)) \\
 &= 0.0000 \text{ Deg} \\
 Apls &= 0.5*ris^2*Lb * (1 + cos(psi))/(ris+0.5 * eas * sin(psi)) \\
 &= 0.5*1503.00^2*300.00 *(1+cos(0.00))/(1503.00 +0.5*11.00 * sin(0. \\
 &= 450.9004 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Metal area

$$\begin{aligned}
 Afb1 &= 235.2146, Afb2 = 102.4085 \text{ mm}^2 \\
 Afls &= eas * (Lb - a1 - a2) \\
 &= 11.000 * (300.000 - 84.150 - 44.450) = 1885.4032 \text{ mm}^2 \\
 Afw &= Afw1 + Afw2 = 196.000 + 196.000 = 392.0000 \text{ mm}^2 \\
 APsil &= 0.0000 \text{ and } APpsi2 = 0.0000 \text{ mm}^2 \\
 f1 &= (Afls + Afw) * (fs - 0.5 * p) \\
 &+ Afb1 * (fob1 - 0.5 * p) \\
 &+ Afb2 * (fob2 - 0.5 * p) \\
 &+ Afp1 * (fop1 - 0.5 * p) \\
 &+ Afp2 * (fop2 - 0.5 * p) \\
 &= (1885.403 + 392.000) * (204.174 - 0.5 * 0.600)
 \end{aligned}$$

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$$\begin{aligned}
 &+ 235.215 * (204.174 - 0.5 * 0.600 ) \\
 &+ 102.408 * (162.506 - 0.5 * 0.600 ) \\
 &+ 991.900 * (204.174 - 0.5 * 0.600 ) \\
 &+ 777.700 * (204.174 - 0.5 * 0.600 ) \\
 &= 889.5678 \times 10^3 \text{ N} \\
 f2 &= P * ( Apls + Apb1 + 0.5 * APpsi1 + Apb2 + 0.5 * APpsi2) \\
 &= 0.600 *(450900.438+2592.401 + .5*0.000 +808.003 +0.50* 0.000 ) \\
 &= 272.5573 \times 10^3 \text{ N}
 \end{aligned}$$

Because  $f1 > f2$ : The Group Is Adequately Reinforced

Nozzles R5 and R17b are a Group per EN 13445 para 9.6.4

The Nozzles are Installed In a Cylinder

$$\begin{aligned}
 ris &= (ris1 + ris2)/2 = (1503.000 + 1503.000 ) / 2 = 1503.0000 \text{ mm} \\
 Lb &= 259.9996 \text{ mm (= Inside center to center distance)} \\
 a1 &= deb1/2 = 109.5500 \text{ and } a2 = deb2/2 = 25.5000 \text{ mm} \\
 k &= 2 - (Lb + a1 + a2)/(Iso1 + Iso2) \\
 &= 2 - (260.000 + 109.550 + 25.500 )/(182.173 + 182.173 ) = 0.9157 \\
 Lb1 &= Lb + a1 + a2 + k * (Iso1 + Iso2) \\
 &= 260.000 + 109.550 + 25.500 + 0.916 * (182.173 + 182.173 ) \\
 &= 728.6920 \text{ mm} \\
 psi &= atan(ris * abs(Angle2 - Angle1) / abs( y2 - y1 ) \\
 &= atan(1503.000 * abs(180.000 - 180.000 ) / abs( 8910.000 - 8650.0 \\
 &= 0.0000 \text{ Deg} \\
 Apls &= 0.5*ris^2*Lb * (1 + cos(psi))/(ris+0.5 * eas * sin(psi)) \\
 &= 0.5*1503.00^2*260.00 *(1+cos(0.00 ))/(1503.00 +0.5*11.00 * sin(0. \\
 &= 390.7794 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Metal area

$$\begin{aligned}
 Afb1 &= 269.7226 , Afb2 = 188.6294 \text{ mm}^2 \\
 Afls &= eas * (Lb - a1 - a2) \\
 &= 11.000 * (260.000 - 109.550 - 25.500 ) = 1374.4460 \text{ mm}^2 \\
 Afw &= Afw1 + Afw2 = 196.000 + 98.000 = 294.0000 \text{ mm}^2 \\
 APsil &= 0.0000 \text{ and } APpsi2 = 0.0000 \text{ mm}^2
 \end{aligned}$$

$$\begin{aligned}
 f1 &= (Afls + Afw) * (fs - 0.5 * p) \\
 &+ Afb1 * (fob1 - 0.5 * p) \\
 &+ Afb2 * (fob2 - 0.5 * p) \\
 &+ Afp1 * (fop1 - 0.5 * p) \\
 &+ Afp2 * (fop2 - 0.5 * p) \\
 &= (1374.446 + 294.000) * (204.174 - 0.5 * 0.600) \\
 &+ 269.723 * (204.174 - 0.5 * 0.600) \\
 &+ 188.629 * (162.506 - 0.5 * 0.600) \\
 &+ 1266.300 * (204.174 - 0.5 * 0.600) \\
 &+ 0.000 * (0.000 - 0.5 * 0.600) \\
 &= 683.8460 \times 10^3 \text{ N} \\
 f2 &= P * (Apls + Apb1 + 0.5 * APpsi1 + Apb2 + 0.5 * APpsi2) \\
 &= 0.600 * (390779.438 + 3951.437 + .5 * 0.000 + 317.692 + 0.50 * 0.000) \\
 &= 237.0090 \times 10^3 \text{ N}
 \end{aligned}$$

Because  $f1 > f2$ : The Group Is Adequately Reinforced

Nozzles R11 and R9 are a Group per EN 13445 para 9.6.4

The Nozzles are Installed In a Cylinder

$$\begin{aligned}
 ris &= (ris1 + ris2) / 2 = (1503.000 + 1503.000) / 2 = 1503.0000 \text{ mm} \\
 Lb &= 335.0330 \text{ mm (= Inside center to center distance)} \\
 a1 &= deb1 / 2 = 44.4500 \text{ and } a2 = deb2 / 2 = 44.4500 \text{ mm} \\
 k &= 2 - (Lb + a1 + a2) / (Iso1 + Iso2) \\
 &= 2 - (335.033 + 44.450 + 44.450) / (182.173 + 182.173) = 0.8365 \\
 Lb1 &= Lb + a1 + a2 + k * (Iso1 + Iso2) \\
 &= 335.033 + 44.450 + 44.450 + 0.836 * (182.173 + 182.173) \\
 &= 728.6920 \text{ mm} \\
 psi &= \text{atan}(ris * \text{abs}(\text{Angle2} - \text{Angle1}) / \text{abs}(y2 - y1)) \\
 &= \text{atan}(1503.000 * \text{abs}(90.000 - 95.686) / \text{abs}(9100.000 - 8800.000)) \\
 &= 26.4357 \text{ Deg} \\
 Apls &= 0.5 * ris^2 * Lb * (1 + \text{cos}(psi)) / (ris + 0.5 * eas * \text{sin}(psi)) \\
 &= 0.5 * 1503.00^2 * 335.03 * (1 + \text{cos}(26.44)) / (1503.00 + 0.5 * 11.00 * \text{sin}(26.44)) \\
 &= 476.4514 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Metal area

$$Afb1 = 102.4085 , Afb2 = 102.4085 \text{ mm}^2$$

$$Afls = eas * (Lb - a1 - a2)$$

$$= 11.000 * (335.033 - 44.450 - 44.450 ) = 2707.4636 \text{ mm}^2$$

$$Afw = Afw1 + Afw2 = 196.000 + 196.000 = 392.0000 \text{ mm}^2$$

$$APsil = 0.0000 \text{ and } APpsi2 = 0.0000 \text{ mm}^2$$

$$f1 = (Afls + Afw ) * (fs - 0.5 * p)$$

$$+ Afb1 * (fob1 - 0.5 * p)$$

$$+ Afb2 * (fob2 - 0.5 * p)$$

$$+ Afp1 * (fop1 - 0.5 * p)$$

$$+ Afp2 * (fop2 - 0.5 * p)$$

$$= (2707.464 + 392.000 ) * (204.174 - 0.5 * 0.600 )$$

$$+ 102.408 * (204.174 - 0.5 * 0.600 )$$

$$+ 102.408 * (204.174 - 0.5 * 0.600 )$$

$$+ 777.700 * (204.174 - 0.5 * 0.600 )$$

$$+ 777.700 * (204.174 - 0.5 * 0.600 )$$

$$= 990.6776 \times 10^3 \text{ N}$$

$$f2 = P * ( Apl1 + Apb1 + 0.5 * APpsi1 + Apb2 + 0.5 * APpsi2)$$

$$= 0.600 * (476451.344 + 808.003 + .5 * 0.000 + 808.003 + 0.5 * 0.000 )$$

$$= 286.8160 \times 10^3 \text{ N}$$

Because  $f1 > f2$ : The Group Is Adequately Reinforced

Nozzles R10 and R12 are a Group per EN 13445 para 9.6.4

The Nozzles are Installed In a Cylinder

$$ris = (ris1 + ris2)/2 = (1503.000 + 1503.000 ) / 2 = 1503.0000 \text{ mm}$$

$$Lb = 417.9088 \text{ mm } (= \text{ Inside center to center distance})$$

$$a1 = deb1/2 = 44.4500 \text{ and } a2 = deb2/2 = 44.4500 \text{ mm}$$

$$k = 2 - (Lb + a1 + a2)/(Iso1 + Iso2)$$

$$= 2 - (417.909 + 44.450 + 44.450 )/(182.173 + 182.173 ) = 0.6090$$

$$Lb1 = Lb + a1 + a2 + k * (Iso1 + Iso2)$$

$$= 417.909 + 44.450 + 44.450 + 0.609 * (182.173 + 182.173 )$$

$$= 728.6920 \text{ mm}$$

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$$\begin{aligned} \text{psi} &= \text{atan}(\text{ris} * \text{abs}(\text{Angle2} - \text{Angle1}) / \text{abs}(y2 - y1)) \\ &= \text{atan}(1503.000 * \text{abs}(142.429 - 131.338) / \text{abs}(9100.000 - 8800.0)) \\ &= 44.1219 \text{ Deg} \\ \text{Apls} &= 0.5 * \text{ris}^2 * \text{Lb} * (1 + \cos(\text{psi})) / (\text{ris} + 0.5 * \text{eas} * \sin(\text{psi})) \\ &= 0.5 * 1503.00^2 * 417.91 * (1 + \cos(44.12)) / (1503.00 + 0.5 * 11.00 * \sin(44.12)) \\ &= 538.1377 \times 10^3 \text{ mm}^2 \end{aligned}$$

**Metal area**

$$\begin{aligned} \text{Afb1} &= 102.4085, \text{Afb2} = 102.4085 \text{ mm}^2 \\ \text{Afls} &= \text{eas} * (\text{Lb} - a1 - a2) \\ &= 11.000 * (417.909 - 44.450 - 44.450) = 3619.0962 \text{ mm}^2 \\ \text{Afw} &= \text{Afw1} + \text{Afw2} = 196.000 + 196.000 = 392.0000 \text{ mm}^2 \\ \text{APsil} &= 0.0000 \text{ and } \text{APpsi2} = 0.0000 \text{ mm}^2 \\ \text{f1} &= (\text{Afls} + \text{Afw}) * (\text{fs} - 0.5 * p) \\ &\quad + \text{Afb1} * (\text{fob1} - 0.5 * p) \\ &\quad + \text{Afb2} * (\text{fob2} - 0.5 * p) \\ &\quad + \text{Afp1} * (\text{fop1} - 0.5 * p) \\ &\quad + \text{Afp2} * (\text{fop2} - 0.5 * p) \\ &= (3619.096 + 392.000) * (204.174 - 0.5 * 0.600) \\ &\quad + 102.408 * (204.174 - 0.5 * 0.600) \\ &\quad + 102.408 * (204.174 - 0.5 * 0.600) \\ &\quad + 777.700 * (204.174 - 0.5 * 0.600) \\ &\quad + 777.700 * (204.174 - 0.5 * 0.600) \\ &= 1176.5198 \times 10^3 \text{ N} \\ \text{f2} &= P * (\text{Apls} + \text{Apl1} + 0.5 * \text{APpsi1} + \text{Apl2} + 0.5 * \text{APpsi2}) \\ &= 0.600 * (538137 + 808.003 + .5 * 0.000 + 808.003 + 0.5 * 0.000) \\ &= 323.8247 \times 10^3 \text{ N} \end{aligned}$$

Because f1 > f2: The Group Is Adequately Reinforced

Nozzles GV1b and R19 are a Group per EN 13445 para 9.6.4

**The Nozzles are Installed In an Ellipsoidal Head**

$$\begin{aligned} \text{ris} &= (\text{ris1} + \text{ris2}) / 2 = (2705.400 + 2705.400) / 2 = 2705.3999 \text{ mm} \\ \text{a1} &= \text{deb1} / 2 = 264.0000 \text{ and } \text{a2} = \text{deb2} / 2 = 119.5500 \text{ mm} \\ \text{k} &= 2 - (\text{Lb} + \text{a1} + \text{a2}) / (\text{Iso1} + \text{Iso2}) \end{aligned}$$

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$$= 2 - (864.645 + 264.000 + 119.550) / (244.213 + 244.213) = -0.555$$

$$Lb1 = Lb + a1 + a2 + k * (Iso1 + Iso2)$$

$$= 864.645 + 264.000 + 119.550 + -0.556 * (244.213 + 244.213)$$

$$= 976.8505 \text{ mm}$$

$$Lb = 864.6447 \text{ mm}$$

$$Apls = 0.25 * (2 * ris)^2 * Lb / (2 * ris + eas)$$

$$= 0.25 * (2 * 2705.400)^2 * 864.645 / (2 * 2705.400 + 0.433)$$

$$= 1167.2319 \times 10^3 \text{ mm}^2$$

Metal area

$$Afb1 = 422.7327, Afb2 = 282.1529 \text{ mm}^2$$

$$Afls = eas * (Lb - a1 - a2)$$

$$= 11.000 * (864.645 - 264.000 - 119.550) = 5292.0420 \text{ mm}^2$$

$$Afw = Afw1 + Afw2 = 196.000 + 196.000 = 392.0000 \text{ mm}^2$$

$$APsil = 0.0000 \text{ and } APpsi2 = 8365.6270 \text{ mm}^2$$

$$f1 = (Afls + Afw) * (fs - 0.5 * p)$$

$$+ Afb1 * (fob1 - 0.5 * p)$$

$$+ Afb2 * (fob2 - 0.5 * p)$$

$$+ Afp1 * (fop1 - 0.5 * p)$$

$$+ Afp2 * (fop2 - 0.5 * p)$$

$$= (5292.042 + 392.000) * (204.174 - 0.5 * 0.600)$$

$$+ 422.733 * (204.174 - 0.5 * 0.600)$$

$$+ 282.153 * (204.174 - 0.5 * 0.600)$$

$$+ 1904.000 * (204.174 - 0.5 * 0.600)$$

$$+ 496.300 * (204.174 - 0.5 * 0.600)$$

$$= 1791.7412 \times 10^3 \text{ N}$$

$$f2 = P * (Apls + Apb1 + 0.5 * APpsi1 + Apb2 + 0.5 * APpsi2)$$

$$= 0.600 * (1167231 + 18347.332 + .5 * 0.000 + 5774.667 + 0.5 * 8365.627)$$

$$= 717.2611 \times 10^3 \text{ N}$$

Because  $f1 > f2$ : The Group Is Adequately Reinforced

Nozzles GV1b and R15 are a Group per EN 13445 para 9.6.4

-----

The Nozzles are Installed In an Ellipsoidal Head

$$\begin{aligned}
 ris &= (ris1 + ris2)/2 = (2705.400 + 2705.400) / 2 = 2705.3999 \text{ mm} \\
 a1 &= deb1/2 = 264.0000 \text{ and } a2 = deb2/2 = 44.4500 \text{ mm} \\
 k &= 2 - (Lb + a1 + a2)/(Iso1 + Iso2) \\
 &= 2 - (708.056 + 264.000 + 44.450)/(244.213 + 244.213) = -0.0812 \\
 Lb1 &= Lb + a1 + a2 + k * (Iso1 + Iso2) \\
 &= 708.056 + 264.000 + 44.450 + -0.081 * (244.213 + 244.213) \\
 &= 976.8505 \text{ mm} \\
 Lb &= 708.0556 \text{ mm} \\
 Apl_s &= 0.25 * (2 * ris)^2 * Lb / (2 * ris + eas) \\
 &= 0.25 * (2 * 2705.400)^2 * 708.056 / (2 * 2705.400 + 0.433) \\
 &= 955.8436 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Metal area

$$\begin{aligned}
 Afb1 &= 422.7327, Afb2 = 102.4085 \text{ mm}^2 \\
 Afl_s &= eas * (Lb - a1 - a2) \\
 &= 11.000 * (708.056 - 264.000 - 44.450) = 4395.6621 \text{ mm}^2 \\
 Afw &= Afw1 + Afw2 = 196.000 + 196.000 = 392.0000 \text{ mm}^2 \\
 APsil &= 0.0000 \text{ and } APpsi2 = 831.9406 \text{ mm}^2 \\
 f1 &= (Afl_s + Afw) * (fs - 0.5 * p) \\
 &+ Afb1 * (fob1 - 0.5 * p) \\
 &+ Afb2 * (fob2 - 0.5 * p) \\
 &+ Afp1 * (fop1 - 0.5 * p) \\
 &+ Afp2 * (fop2 - 0.5 * p) \\
 &= (4395.662 + 392.000) * (204.174 - 0.5 * 0.600) \\
 &+ 422.733 * (204.174 - 0.5 * 0.600) \\
 &+ 102.408 * (204.174 - 0.5 * 0.600) \\
 &+ 1904.000 * (204.174 - 0.5 * 0.600) \\
 &+ 777.700 * (204.174 - 0.5 * 0.600) \\
 &= 1629.7316 \times 10^3 \text{ N} \\
 f2 &= P * (Apl_s + Apb1 + 0.5 * APsil + Apb2 + 0.5 * APpsi2) \\
 &= 0.600 * (955843 + 18347.332 + .5*0.000 + 1241.953 + 0.50* 831.941) \\
 &= 585.4596 \times 10^3 \text{ N}
 \end{aligned}$$

Because  $f_1 > f_2$ : The Group Is Adequately Reinforced

Nozzles R19 and R14 are a Group per EN 13445 para 9.6.4

The Nozzles are Installed In an Ellipsoidal Head

$$\begin{aligned} ris &= (ris_1 + ris_2)/2 = (2705.400 + 2705.400) / 2 = 2705.3999 \text{ mm} \\ a_1 &= deb_1/2 = 119.5500 \text{ and } a_2 = deb_2/2 = 44.4500 \text{ mm} \\ k &= 2 - (Lb + a_1 + a_2)/(Iso_1 + Iso_2) \\ &= 2 - (436.511 + 119.550 + 44.450)/(244.213 + 244.213) = 0.7705 \\ Lb_1 &= Lb + a_1 + a_2 + k * (Iso_1 + Iso_2) \\ &= 436.511 + 119.550 + 44.450 + 0.771 * (244.213 + 244.213) \\ &= 976.8505 \text{ mm} \\ Lb &= 436.5111 \text{ mm} \\ Apls &= 0.25 * (2 * ris)^2 * Lb / (2 * ris + eas) \\ &= 0.25 * (2 * 2705.400)^2 * 436.511 / (2 * 2705.400 + 0.433) \\ &= 589.2706 \times 10^3 \text{ mm}^2 \end{aligned}$$

Metal area

$$\begin{aligned} Afb_1 &= 282.1529, Afb_2 = 102.4085 \text{ mm}^2 \\ Afls &= eas * (Lb - a_1 - a_2) \\ &= 11.000 * (436.511 - 119.550 - 44.450) = 2997.6223 \text{ mm}^2 \\ Afw &= Afw_1 + Afw_2 = 196.000 + 196.000 = 392.0000 \text{ mm}^2 \\ APsil &= 8365.6270 \text{ and } APpsi_2 = 1360.8990 \text{ mm}^2 \\ f_1 &= (Afls + Afw) * (fs - 0.5 * p) \\ &\quad + Afb_1 * (fob_1 - 0.5 * p) \\ &\quad + Afb_2 * (fob_2 - 0.5 * p) \\ &\quad + Afp_1 * (fop_1 - 0.5 * p) \\ &\quad + Afp_2 * (fop_2 - 0.5 * p) \\ &= (2997.622 + 392.000) * (204.174 - 0.5 * 0.600) \\ &\quad + 282.153 * (204.174 - 0.5 * 0.600) \\ &\quad + 102.408 * (204.174 - 0.5 * 0.600) \\ &\quad + 496.300 * (204.174 - 0.5 * 0.600) \\ &\quad + 777.700 * (204.174 - 0.5 * 0.600) \\ &= 1029.1052 \times 10^3 \text{ N} \end{aligned}$$



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Nozzle Summary :

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$$\begin{aligned}
 f2 &= P * ( Apls + Apb1 + 0.5 * APpsi1 + Apb2 + 0.5 * APpsi2) \\
 &= 0.600 * (589270 + 5774.667 + .5 * 8365.627 + 1241.953 + 0.50 * 1360.899 ) \\
 &= 360.6596 \times 10^3 \text{ N}
 \end{aligned}$$

Because  $f1 > f2$ : The Group Is Adequately Reinforced

Nozzles R15 and R18b are a Group per EN 13445 para 9.6.4

-----

The Nozzles are Installed In an Ellipsoidal Head

$$\begin{aligned}
 ris &= (ris1 + ris2)/2 = (2705.400 + 2705.400 ) / 2 = 2705.3999 \text{ mm} \\
 a1 &= deb1/2 = 44.4500 \text{ and } a2 = deb2/2 = 44.4500 \text{ mm} \\
 k &= 2 - (Lb + a1 + a2)/(Iso1 + Iso2) \\
 &= 2 - (506.251 + 44.450 + 44.450 ) / (244.213 + 244.213 ) = 0.7815 \\
 Lb1 &= Lb + a1 + a2 + k * (Iso1 + Iso2) \\
 &= 506.251 + 44.450 + 44.450 + 0.781 * (244.213 + 244.213 ) \\
 &= 976.8505 \text{ mm} \\
 Lb &= 506.2511 \text{ mm} \\
 Apls &= 0.25 * (2 * ris)^2 * Lb / ( 2 * ris + eas ) \\
 &= 0.25 * (2 * 2705.400 )^2 * 506.251 / ( 2 * 2705.400 + 0.433 ) \\
 &= 683.4164 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Metal area

$$\begin{aligned}
 Afb1 &= 102.4085 , Afb2 = 102.4085 \text{ mm}^2 \\
 Afls &= eas * (Lb - a1 - a2) \\
 &= 11.000 * (506.251 - 44.450 - 44.450 ) = 4590.8618 \text{ mm}^2 \\
 Afw &= Afw1 + Afw2 = 196.000 + 196.000 = 392.0000 \text{ mm}^2 \\
 APpsi1 &= 831.9406 \text{ and } APpsi2 = 1041.2191 \text{ mm}^2 \\
 f1 &= (Afls + Afw ) * (fs - 0.5 * p) \\
 &\quad + Afb1 * (fob1 - 0.5 * p) \\
 &\quad + Afb2 * (fob2 - 0.5 * p) \\
 &\quad + Afp1 * (fop1 - 0.5 * p) \\
 &\quad + Afp2 * (fop2 - 0.5 * p) \\
 &= (4590.862 + 392.000 ) * (204.174 - 0.5 * 0.600 ) \\
 &\quad + 102.408 * (204.174 - 0.5 * 0.600 ) \\
 &\quad + 102.408 * (204.174 - 0.5 * 0.600 ) \\
 &\quad + 777.700 * (204.174 - 0.5 * 0.600 )
 \end{aligned}$$

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Nozzle Summary :

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$$\begin{aligned}
 & + 777.700 * (204.174 - 0.5 * 0.600 ) \\
 & = 1374.6206 \times 10^3 \text{ N} \\
 f2 & = P * ( Apls + Apb1 + 0.5 * APpsi1 + Apb2 + 0.5 * APpsi2) \\
 & = 0.600 *(683416 +1241.953 + .5*831.941 +1241.953 +0.50* 1041.219 ) \\
 & = 412.0671 \times 10^3 \text{ N}
 \end{aligned}$$

Because  $f1 > f2$ : The Group Is Adequately Reinforced

Nozzles R15 and R13 are a Group per EN 13445 para 9.6.4

The Nozzles are Installed In an Ellipsoidal Head

$$\begin{aligned}
 ris & = (ris1 + ris2)/2 = (2705.400 + 2705.400 ) / 2 = 2705.3999 \text{ mm} \\
 a1 & = deb1/2 = 44.4500 \text{ and } a2 = deb2/2 = 44.4500 \text{ mm} \\
 k & = 2 - (Lb + a1 + a2)/(Iso1 + Iso2) \\
 & = 2 - (338.076 + 44.450 + 44.450 )/(244.213 + 244.213 ) = 1.1258 \\
 \text{as } k > 1, & \text{ set } k \text{ to } 1.00 \\
 Lb1 & = Lb + a1 + a2 + k * (Iso1 + Iso2) \\
 & = 338.076 + 44.450 + 44.450 + 1.000 * (244.213 + 244.213 ) \\
 & = 915.4014 \text{ mm} \\
 Lb & = 338.0761 \text{ mm} \\
 Apls & = 0.25 * (2 * ris)^2 * Lb / ( 2 * ris + eas ) \\
 & = 0.25 * (2 * 2705.400 )^2 * 338.076 / ( 2 * 2705.400 + 0.433 ) \\
 & = 456.3878 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Metal area

$$\begin{aligned}
 Afb1 & = 102.4085 , Afb2 = 102.4085 \text{ mm}^2 \\
 Afls & = eas * (Lb - a1 - a2) \\
 & = 11.000 * (338.076 - 44.450 - 44.450 ) = 2740.9377 \text{ mm}^2 \\
 Afw & = Afw1 + Afw2 = 196.000 + 98.000 = 294.0000 \text{ mm}^2 \\
 APsil & = 831.9406 \text{ and } APpsi2 = 1251.0686 \text{ mm}^2
 \end{aligned}$$

$$\begin{aligned}
 f1 & = (Afls + Afw ) * (fs - 0.5 * p) \\
 & + Afb1 * (fob1 - 0.5 * p) \\
 & + Afb2 * (fob2 - 0.5 * p) \\
 & + Afp1 * (fop1 - 0.5 * p) \\
 & + Afp2 * (fop2 - 0.5 * p)
 \end{aligned}$$

$$\begin{aligned}
 &= (2740.938 + 294.000) * (204.174 - 0.5 * 0.600) \\
 &+ 102.408 * (204.174 - 0.5 * 0.600) \\
 &+ 102.408 * (162.506 - 0.5 * 0.600) \\
 &+ 777.700 * (204.174 - 0.5 * 0.600) \\
 &+ 0.000 * (135.206 - 0.5 * 0.600) \\
 &= 814.7175 \times 10^3 \text{ N} \\
 f2 &= P * (Apl1 + Apl2 + 0.5 * APpsi1 + Apl3 + 0.5 * APpsi2) \\
 &= 0.600 * (456387.750 + 1241.953 + .5 * 831.941 + 1241.953 + 0.50 * 1251.069) \\
 &= 275.9245 \times 10^3 \text{ N}
 \end{aligned}$$

Because  $f1 > f2$ : The Group Is Adequately Reinforced

Nozzles R18b and R16 are a Group per EN 13445 para 9.6.4

The Nozzles are Installed In an Ellipsoidal Head

$$\begin{aligned}
 ris &= (ris1 + ris2)/2 = (2705.400 + 2705.400) / 2 = 2705.3999 \text{ mm} \\
 a1 &= deb1/2 = 44.4500 \text{ and } a2 = deb2/2 = 44.4500 \text{ mm} \\
 k &= 2 - (Lb + a1 + a2)/(Iso1 + Iso2) \\
 &= 2 - (349.626 + 44.450 + 44.450) / (244.213 + 244.213) = 1.1022 \\
 \text{as } k &> 1, \text{ set } k \text{ to } 1.00 \\
 Lb1 &= Lb + a1 + a2 + k * (Iso1 + Iso2) \\
 &= 349.626 + 44.450 + 44.450 + 1.000 * (244.213 + 244.213) \\
 &= 926.9514 \text{ mm} \\
 Lb &= 349.6261 \text{ mm} \\
 Apl1 &= 0.25 * (2 * ris)^2 * Lb / (2 * ris + eas) \\
 &= 0.25 * (2 * 2705.400)^2 * 349.626 / (2 * 2705.400 + 0.433) \\
 &= 471.9797 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Metal area

$$\begin{aligned}
 Afb1 &= 102.4085, Afb2 = 102.4085 \text{ mm}^2 \\
 Afl1 &= eas * (Lb - a1 - a2) \\
 &= 11.000 * (349.626 - 44.450 - 44.450) = 2867.9873 \text{ mm}^2 \\
 Afw &= Afw1 + Afw2 = 196.000 + 196.000 = 392.0000 \text{ mm}^2 \\
 APpsi1 &= 1041.2191 \text{ and } APpsi2 = 1381.8744 \text{ mm}^2 \\
 f1 &= (Afl1 + Afb1) * (fs - 0.5 * p) \\
 &+ Afb1 * (fob1 - 0.5 * p)
 \end{aligned}$$

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Nozzle Summary :

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$$\begin{aligned}
 &+ Afb2 * (fob2 - 0.5 * p) \\
 &+ Afp1 * (fop1 - 0.5 * p) \\
 &+ Afp2 * (fop2 - 0.5 * p) \\
 = &(2867.987 + 392.000 ) * (204.174 - 0.5 * 0.600 ) \\
 &+ 102.408 * (204.174 - 0.5 * 0.600 ) \\
 &+ 102.408 * (204.174 - 0.5 * 0.600 ) \\
 &+ 777.700 * (204.174 - 0.5 * 0.600 ) \\
 &+ 777.700 * (204.174 - 0.5 * 0.600 ) \\
 = &1023.4013 \times 10^3 \text{ N} \\
 f2 = &P * ( Apls + Apb1 + 0.5 * APpsi1 + Apb2 + 0.5 * APpsi2) \\
 = &0.600 *(471979.688+1241.953 + .5*1041.219 +1241.953 +0.50* 1381.87 \\
 = &285.3809 \times 10^3 \text{ N}
 \end{aligned}$$

Because  $f1 > f2$ : The Group Is Adequately Reinforced

Nozzles R16 and R14 are a Group per EN 13445 para 9.6.4

-----  
 The Nozzles are Installed In an Ellipsoidal Head

$$\begin{aligned}
 ris &= (ris1 + ris2)/2 = (2705.400 + 2705.400 ) / 2 = 2705.3999 \text{ mm} \\
 a1 &= deb1/2 = 44.4500 \text{ and } a2 = deb2/2 = 44.4500 \text{ mm} \\
 k &= 2 - (Lb + a1 + a2)/(Iso1 + Iso2) \\
 &= 2 - (323.601 + 44.450 + 44.450 )/(244.213 + 244.213 ) = 1.1554 \\
 \text{as } k &> 1, \text{ set } k \text{ to } 1.00 \\
 Lb1 &= Lb + a1 + a2 + k * (Iso1 + Iso2) \\
 &= 323.601 + 44.450 + 44.450 + 1.000 * (244.213 + 244.213 ) \\
 &= 900.9262 \text{ mm} \\
 Lb &= 323.6009 \text{ mm} \\
 Apls &= 0.25 * (2 * ris)^2 * Lb / ( 2 * ris + eas ) \\
 &= 0.25 * (2 * 2705.400 )^2 * 323.601 / ( 2 * 2705.400 + 0.433 ) \\
 &= 436.8469 \times 10^3 \text{ mm}^2
 \end{aligned}$$

Metal area

$$\begin{aligned}
 Afb1 &= 102.4085 , Afb2 = 102.4085 \text{ mm}^2 \\
 Afls &= eas * (Lb - a1 - a2) \\
 &= 11.000 * (323.601 - 44.450 - 44.450 ) = 2581.7104 \text{ mm}^2 \\
 Afw &= Afw1 + Afw2 = 196.000 + 196.000 = 392.0000 \text{ mm}^2
 \end{aligned}$$

$$APsil = 1381.8744 \text{ and } APpsi2 = 1360.8990 \text{ mm}^2$$

$$f1 = (Afls + Afw ) * (fs - 0.5 * p)$$

$$+ Afb1 * (fob1 - 0.5 * p)$$

$$+ Afb2 * (fob2 - 0.5 * p)$$

$$+ Afp1 * (fop1 - 0.5 * p)$$

$$+ Afp2 * (fop2 - 0.5 * p)$$

$$= (2581.710 + 392.000 ) * (204.174 - 0.5 * 0.600 )$$

$$+ 102.408 * (204.174 - 0.5 * 0.600 )$$

$$+ 102.408 * (204.174 - 0.5 * 0.600 )$$

$$+ 777.700 * (204.174 - 0.5 * 0.600 )$$

$$+ 777.700 * (204.174 - 0.5 * 0.600 )$$

$$= 965.0420 \times 10^3 \text{ N}$$

$$f2 = P * ( AplS + Apb1 + 0.5 * APpsi1 + Apb2 + 0.5 * APpsi2)$$

$$= 0.600 * (436846.844 + 1241.953 + .5 * 1381.874 + 1241.953 + 0.50 * 1360.89$$

$$= 264.3988 \times 10^3 \text{ N}$$

Because  $f1 > f2$ : The Group Is Adequately Reinforced

No interference violations have been detected !

Please refer to the nozzles treated as isolated openings for more information

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Vessel Design Summary : Step: 44 7:08a Nov 7,2012

**Design Code: European Std: EN13445-3 2009(E) Issue 1 (2009-07)**

Diameter Spec : 3000.000 mm ID

Vessel Design Length, Tangent to Tangent 10450.00 mm

Specified Datum Line Distance 0.00 mm

Shell Material Specification P355NH

Nozzle Material Specification P285QH

Nozzle Material Specification P355NH

Re-Pad Material Specification P355NH

Re-Pad Material Specification P235GH

Internal Design Temperature 60 C

Internal Design Pressure 0.600 MPa

External Design Temperature -20 C

External Design Pressure 2.180 MPa

Maximum Allowable Working Pressure 1.093 MPa

Hydrostatic Test Pressure 1.504 MPa

Wind Design Code ASCE-93

Earthquake Design Code UBC-97

**Element Pressures and MAWP: MPa**

Element Desc	Design Pres.	External	M.A.W.P	Corrosion
	+ Stat. head	Pressure		Allowance

FUND ELIPSOIDAL 1	0.600	0.000	1.093	3.0000
VIROLA	0.600	0.000	1.489	3.0000
FUND ELIPSOIDAL 2	0.600	0.000	1.294	3.0000

Element Type	"To" Elev mm	Length mm	Element Thk mm	Req d Int.	Thk Ext.	Joint Long	Eff Circ
Ellipse	50.0	50.0	14.0	9.5	No Calc	1.00	1.00
Cylinder	10400.0	10350.0	14.0	7.4	No Calc	1.00	1.00
Ellipse	10450.0	50.0	14.0	8.9	No Calc	1.00	1.00

Element thicknesses are shown as Nominal if specified, otherwise are Minimum

**Saddle Parameters:**

Saddle Width	400.000	mm
Saddle Bearing Angle	120.000	deg.
Centerline Dimension	2000.000	mm
Wear Pad Width	500.000	mm
Wear Pad Thickness	14.000	mm
Wear Pad Bearing Angle	127.000	deg.
Distance from Saddle to Tangent	685.000	mm
Baseplate Length	2400.000	mm
Baseplate Thickness	20.000	mm
Baseplate Width	400.000	mm
Number of Ribs (including outside ribs)	4	
Rib Thickness	20.000	mm
Web Thickness	20.000	mm
Height of Center Web	448.000	mm

**Summary of Maximum Saddle Loads, Operating Case :**

Maximum Vertical Saddle Load	95112.34	N
Maximum Transverse Saddle Shear Load	14091.53	N
Maximum Longitudinal Saddle Shear Load	28183.05	N

**Summary of Maximum Saddle Loads, Hydrotest Case :**

Maximum Vertical Saddle Load	483099.19	N
Maximum Transverse Saddle Shear Load	2233.84	N
Maximum Longitudinal Saddle Shear Load	1371.65	N

**Weights:**

Fabricated - Bare W/O Removable Internals	15602.6	kg
Shop Test - Fabricated + Water ( Full )	96503.0	kg
Shipping - Fab. + Rem. Intls.+ Shipping App.	15602.6	kg
Erected - Fab. + Rem. Intls.+ Insul. (etc)	15602.6	kg
Empty - Fab. + Intls. + Details + Wghts.	15602.6	kg
Operating - Empty + Operating Liquid (No CA)	15602.6	kg
Field Test - Empty Weight + Water (Full)	96503.0	kg